#### **PAPER**

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# Vibration energy harvesters with optimized geometry, design, and nonlinearity for robust direct current power delivery

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#### **Abstract**

With an ever-growing Internet-of-things, vibration energy harvesting has attracted broad attention to replace consumable batteries to power the many microelectronic devices. To this end, an energy harvester must deliver the required power to an electrical load over a long time horizon. Yet, design practices for energy harvesters often report strategies based on maximizing output voltage and wide frequency range of operation, which is not directly related to performance-robust functioning. Motivated to provide valuable insight to practical development of vibration energy harvesters, this research develops an analytical modeling framework and optimization technique to guide attention to piezoelectric laminated energy harvesting cantilevers with balanced and robust performance characteristics. The model is numerically and experimentally validated to confirm the efficacy of the optimization outcomes. The results indicate that laminated trapezoidal beam shapes with monostable configuration are the best solution to broaden the frequency range of enhanced dynamic behavior, minimize strain at the clamped beam end, and maximize the output voltage in a rectifier circuit. The results also find that the selection of tip mass may not be highly influential for the overall performance so long as the beam shape, beam length, and placement of nonlinearity-induced magnets are appropriately chosen.

Keywords: laminated piezoelectric beam, multi-objective optimization, vibration energy harvesting, genetic algorithm

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(Some figures may appear in colour only in the online journal)

#### 1. Introduction

With an accelerating development of the Internet-of-things (IoT), sustainably powering the vast fleet of low-power wireless devices in our world is on the verge of crisis [1]. With a projected growth of IoT devices upwards of tens of billions by 2020 [2], the excessive reliance on chemical batteries threatens the environment and resilience of the global energy economy. Yet, vibrational energy harvesting suggests a promising solution to meet a portion of the accelerating power supply demand [3]. Optimizing such vibration energy harvester platforms to ensure robust direct current (DC) power when subjected to broadband vibration excitation, and

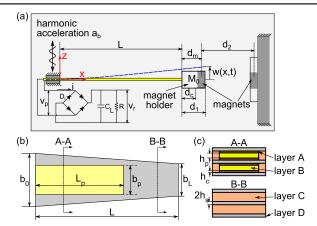
over a long time horizon, is therefore central to the vision of a sustainable IoT [4].

The fundamental requirement of a vibration energy harvester is to ensure the necessary DC power is delivered to an electrical load. Optimization strategies therefore seek to enable this requirement. A straightforward method to maximize output power is to increase the electromechanical coupling coefficient for a piezoelectric harvester platform, which enhances the conversion efficiency between vibration and electrical energies. Cho *et al* [5] uncovered the influences of residual stress, layering thicknesses, and electrode coverage on the resulting electromechanical coupling coefficient for an optimized thin-film PZT membrane design. Wang and

Wu [6] examined the effects of the piezoelectric patch positioning and size for application on a cantilever beam, to show how power harvesting efficiency could be optimized. Lü et al [7] proposed an optimization method by introducing an intrinsic power density scale to reduce the number of parameters for optimizing energy conversion efficiency. Qin et al [8] investigated the influence of dimensions on the PZT-5A material and arrived at an optimal electromechanical coupling coefficient  $(k_{15})$  for the shear vibration mode of energy harvesting. In addition, after identifying the superiority of a trapezoidal piezoelectric beam shape for mechanical robustness and long service life [9], alternative tapered beam shapes have been studied to enhance the uniformity of strain distribution along the beam length [10–14]. Dietl and Garcia [15] developed an optimization tool on the basis of such trends and proposed an optimal curved piezoelectric beam shape for more uniform strain distribution. Further approaches to the beam configuration have been considered for the sake of maximizing electrical power generation, for example a tapered beam with cavity [16], right-angle piezoelectric cantilevers having auxiliary beams [17], beams with initial curvature [18], and piezoelectric energy harvesters having cellular honeycomb structures [19].

Because linear dynamic response only ensures high output power around the primary resonant frequencies, robust electrical power delivery must involve an energy harvester exhibiting large amplitude dynamic behaviors when subjected to broadband frequency vibration excitation. One well-known method to broaden the frequency range of a piezoelectric energy harvester is to introduce nonlinear magnetic effects [20-22]. Erturk and Inman [23] suggested as much as an 800% increase in output power could be achieved for a bistable magnetopiezoelastic harvester under harmonic vibration when the platform was interfaced with a purely resistive electrical load. Comparatively, Ferrari et al [24] reported a 250% increase in output power for a magnetoelastic energy harvester subjected to wideband stochastic excitation. The frequency range for large amplitude response and large output power has been extended also by adopting adaptive bistable harvester designs to more often realize the snap-through behavior [25-27]. In this spirit, Zhou et al [28] proposed a flexible bistable energy harvester with two elastic beams to create a variable repulsive magnetic force, which may reduce the potential barrier that must be overcome to ensure snapthrough oscillation. Recently instead of focusing on the bistable system, multi-stable systems, such as tri-stable energy harvester, have been introduced to enhance the frequency range and output power achieved for the large amplitude dynamic behaviors [29, 30]. Because the alternating current output from the harvester must be rectified to DC to power electronic devices, nonlinear harvester platforms interfaced with rectifier circuits [31] or power management circuits [32] have also demonstrated broadband frequency response despite interfacing with such nonlinear electrical circuits.

Yet, enhanced dynamic behavior may correspondingly be detrimental to the long working life of the vibration energy harvester. In particular, the PZT material is exceedingly brittle such that it is vulnerable to damage caused by events resulting



**Figure 1.** (a) Schematic of nonlinear energy harvester system with repulsive magnets and rectification circuit interface. (b) Dimensions of the piezoelectric beam. (c) Cross-section of the piezoelectric beam among the several layers.

in high bending strain. Without cautious design, a piezo-electric energy harvester could fail in hours, or even in minutes [33–35]. Gundimeda *et al* [36] have helped quantify the mechanical robustness and design flexibility of laminated beams when compared with conventional metal substrates used for piezoelectric beams. Li *et al* [37] further demonstrated the superiority of laminated piezoelectric beams to yield high output power. Yet, strain conditions or laminated beams are seldom considered when optimizing the energy harvester, especially if nonlinearity is introduced.

Given the state-of-the-art in optimization efforts for piezoelectric energy harvesters, a need exists to develop an optimization approach that exploits nonlinearity in the design of custom-shaped harvesters having excellent DC output power, broad frequency range of operation, and exceptional mechanical robustness. This research meets the need via an analytical model and integrated genetic algorithm (GA) optimization framework. The following sections introduce the analytical method, numerically verify the efficacy of the analysis, and then establish the optimization framework. Optimized piezoelectric energy harvesters are then examined and cross-compared, after which an experimental sequence confirms the validity of the theoretical predictions through laboratory examinations. A summary of main discoveries from this research are presented to conclude this report.

#### 2. Analytical model overview

2.1. Analytical model formulation for a laminated piezoelectric beam with magnetic nonlinearity

Figure 1 presents a schematic of the nonlinear energy harvesting system considered in this research. A laminated piezoelectric cantilever is interfaced with a rectifier bridge  $D_i$ , smoothing capacitor  $C_L$ , and load resistance R, which is a standard energy extraction circuit. The lamination sequence is intricate. As shown in figure 1(b), along the total length L of the clamped beam, the width at the free end  $b_L$  and the width

at the fixed end  $b_0$  are the layers D, composed from a glassreinforced epoxy laminate (FR4). The thickness of the FR4 is  $2h_s$ . Since ceramic materials are exceedingly difficult to arbitrarily shape due to extreme brittleness, here we consider a piezoelectric PZT-5H layer labeled as layer B that adopts a fixed and commercially available rectangular shape. The subsequent optimization of this report considers other beam design parameters as candidates for optimization, in contrast to change of the PZT-5H layer shape. The length, width, and thickness of the piezoelectric layer considered here are  $L_p$ ,  $b_p$ , and  $h_n$ . The two piezoelectric layers shown in figure 1(c) are connected in parallel. The layer A corresponds to a copper electrode layer and has the same length and width as the piezoelectric layer for full electrode coverage. The thickness of the copper layer is  $h_c$ . Layer C denotes polyimide that fills the remaining laminate beam volume in the absence of the copper or piezoelectric layers. Repulsive magnets are included to introduce nonlinearity for tuning the frequency response when the cantilever is subjected to the harmonic base acceleration  $a_b$ . The total tip mass is taken as  $M_0$ , which constitutes both the magnet as well as the holder required to secure the magnet to the cantilever free tip. The other opposing repulsive magnet is attached to the moving base. In the following derivation, the magnet holder shown in figure 1(a) is assumed to be a rigid extension from the beam tip. The length of the holder is  $d_1$ . The distance between two magnet centers is  $d_2$ .

When the transverse displacement at the free end of the beam is w(x, t), the strain distribution including the non-linearity [38] caused by the large amplitude vibration is written

$$S_1 = -z[w_{xx}(1 - w_x^2)^{-1/2}] \simeq -zw_{xx}\left(1 + \frac{1}{2}w_x^2\right).$$
 (1)

The  $w_x$  represents for  $\frac{\partial w}{\partial x}$ ,  $w_{xx}$  is the corresponding second derivative, z is the distance away from the neutral axis, and  $S_1$  indicates the strain in the x direction caused by deflection in the z axis.

For the piezoelectric layers, the electrical potential  $\varphi(z, t)$  is assumed to be a linear function through the respective piezoelectric layer thickness [39, 40]. Therefore

$$E_3 = -\frac{\partial \varphi}{\partial z} = -\varphi_z,\tag{2}$$

where  $E_3$  is the electric field in the z direction. Then, the coupling between mechanical and electrical responses for the piezoelectric layers are

$$\begin{bmatrix} D_3 \\ T_1 \end{bmatrix} = \begin{bmatrix} \varepsilon_{33}^s & e_{31} \\ -e_{31} & E_p \end{bmatrix} \begin{bmatrix} E_3 \\ S_1 \end{bmatrix}. \tag{3}$$

The  $D_3$  is the electric displacement through the piezoelectric layer thickness,  $T_1$  and  $S_1$  are the stress and strain in x direction,  $e_{31}$  refers to the coupling between the electric field in the z direction and the stress in the x direction,  $\varepsilon_{33}^s$  is the piezoelectric permittivity at constant strain,  $E_p$  is the Young's modulus of the PZT-5H.

Since the rigid magnet holder has a finite length, the kinetic energy of the tip mass  $T_m$  is calculated based on the

velocity at the mass center. The distance from the laminated beam end to the mass center of the holder is  $d_c$  as shown in figure 1(a)

$$T_m = \frac{1}{2} M_0 [\dot{w}(L + d_c, t) + v_b]^2$$
  
=  $\frac{1}{2} M_0 [\dot{w}(L, t) + d_c \cdot \dot{w}_x(L, t) + v_b]^2$ . (4)

The  $\dot{w}$  is the first derivative of the displacement with respect to the time t,  $v_b$  is the velocity of the base.

Therefore, the total kinetic energy of the nonlinear system shown in figure 1(a) is given in (5)

$$T = \frac{1}{2} \int_{V_s} \rho_s [\dot{w}(x, t) + v_b]^2 dV + \frac{1}{2} \int_{V_p} \rho_p [\dot{w}(x, t) + v_b]^2 dV + \frac{1}{2} \int_{V_c} \rho_c [\dot{w}(x, t) + v_b]^2 dV + \frac{1}{2} \int_{V_e} \rho_e [\dot{w}(x, t) + v_b]^2 dV + T_m.$$
 (5)

Here and after, the subscripts s, p, c, and e refer to the material of FR4, PZT-5H, copper, and polyimide. These subscripts are then used for the following parameters. The  $\rho$  and V denote the corresponding material density and the total volume for each material.

The magnetic potential energy caused by the magnets [21, 24, 41] is

$$U_{m} = -\int F_{mag} dz = -\frac{1}{2} k_{1} w^{2} (L + d_{mc}, t)$$
$$-\frac{1}{4} k_{3} w^{4} (L + d_{mc}, t), \tag{6}$$

where  $d_{mc}$  is the distance between the magnet center installed on the beam to the laminated beam end.

The magnet pair shown in figure 1(a) include identical magnets. As such, the coefficients in (6) are

$$k_1 = f_0 d_2^{-5}, k_3 = -2.5 f_0 d_2^{-7}, f_0 = 3\tau m^2/(2\pi),$$
  
 $m = M_a \cdot V_{mag}.$  (7)

The  $\tau$  is the permeability constant in vacuum, m is the effective magnetic moment,  $d_2$  is the center-to-center distance between the two repulsive magnets,  $M_a$  is the magnetization of the magnet,  $V_{mag}$  is the magnet volume.

The total potential energy for the system is supposed to be

$$U = \int_{V_s} \frac{1}{2} S_1 T_1 dV + \int_{V_c} \frac{1}{2} S_1 T_1 dV + \int_{V_p} \frac{1}{2} S_1 T_1 dV + \int_{V_p} \frac{1}{2} S_1 T_1 dV + \int_{V_p} \frac{1}{2} E_3 D_3 dV + U_m,$$
(8)

where the term  $T_1$  refers to the stress in the x direction.

Substituting equations (1), (3) and (6) into (8), the total potential energy is then given by

$$U = \int_{V_s} \frac{1}{2} E_s[z^2(w_{xx}^2 + w_{xx}^2 w_x^2)] dV$$

$$+ \int_{V_c} \frac{1}{2} E_c[z^2(w_{xx}^2 + w_{xx}^2 w_x^2)] dV$$

$$+ \int_{V_p} \frac{1}{2} E_p[z^2(w_{xx}^2 + w_{xx}^2 w_x^2)] dV$$

$$+ \int_{V_e} \frac{1}{2} E_e[z^2(w_{xx}^2 + w_{xx}^2 w_x^2)] dV$$

$$- \int_{V_p} \frac{1}{2} (E_3)^2 \varepsilon_{33}^s dV$$

$$- \int_{V_p} e_{31} E_3 \left[ z \left( w_{xx} + \frac{1}{2} w_{xx} w_x^2 \right) \right] dV$$

$$- \frac{1}{2} k_1 w^2 (L + d_{mc}, t) - \frac{1}{4} k_3 w^4 (L + d_{mc}, t).$$
(9)

The *E* indicates the Young's modulus, where the subscripts are those defined for the respective layers.

The external work is composed of two parts: mechanical work due to the external base acceleration  $a_b$  and electrical work caused by the electrical field inside the piezoelectric layers

$$W = \int_{V_s} \rho_s a_b w dV + \int_{V_c} \rho_c a_b w dV + \int_{V_p} \rho_p a_b w dV + \int_{V_e} \rho_e a_b w dV + M_0 a_b w (L + d_c) - \sum_{q=1}^{2} Q_q \cdot v_q.$$
 (10)

The  $Q_q$  is the charge in piezoelectric layer q,  $v_q$  represents the voltage at the piezoelectric electrodes. Since there are two piezoelectric layers, q ranges from 1 to 2 in this study.

Although there is discontinuity along the beam axis where the piezoelectric layers end, only the fundamental out-of-plane bending mode is dominate in the responses analysis as justified in section 2.2. This motivates the use of the Ritz method in the process to obtain the Euler–Lagrange equations of motion. Suitable trial functions in the assumed solution by the Ritz method ensure that the essential boundary conditions are satisfied for convergence of the approximate solution to the accurate result, including the discontinuity [42].

Assuming that the harmonically forced mechanical and electrical behaviors are respectively separable in space and time, the system dynamics may be approximated as a linear combination of linearly independent trial functions and generalized coordinates. For the mechanical responses, it is assumed that

$$w(x, t) = \psi(x)\mathbf{r}(t) = \begin{bmatrix} \psi_1(x) & \cdots & \psi_M(x) \end{bmatrix} \begin{bmatrix} r_1(t) \\ \vdots \\ r_M(t) \end{bmatrix}$$
$$= \sum_{i=1}^{M} \psi_i r_i, \tag{11}$$

where M is the total number of trial functions assumed in the summation (which is 3 in the following model development), and  $r_i(t)$  is the unknown generalized coordinate. The trial functions  $\psi_i(x)$  are approximated by the normal modes of a clamped beam with tip mass given in equation (12)

$$\psi_i(x) = \cos \frac{\lambda_i}{L} x - \cosh \frac{\lambda_i}{L} x + \zeta_i \left( \sin \frac{\lambda_i}{L} x - \sinh \frac{\lambda_i}{L} x \right)$$
(12)

$$\zeta_i = \frac{\sin \lambda_i - \sinh \lambda_i + \lambda_i \frac{M_0}{M_b} (\cos \lambda_i - \cosh \lambda_i)}{\cos \lambda_i + \cosh \lambda_i - \lambda_i \frac{M_0}{M_b} (\sin \lambda_i - \sinh \lambda_i)}.$$
 (13)

The  $M_b$  is the mass of the beam,  $\lambda_i$  is related to the natural frequency of the system, which can be solved by the transcendental equation (14)

$$1 + \cos \lambda \cosh \lambda + \lambda \frac{M_0}{M_b} (\cos \lambda \sinh \lambda - \sin \lambda \cosh \lambda)$$
$$= 0.$$
 (14)

The electrical potential is assumed to vary linearly through the thickness of a piezoelectric layer and become zero-valued outside the piezoelectric domains [15, 40]. The normalized voltage at the electrodes existing at  $z = h_s + h_p + h_c$  and  $z = -(h_s + h_p + h_c)$  are assumed to be 1 and -1, respectively, given the mirrored positions of the electrodes considering the laminate sequence about the middle plane of the beam. Consequently, the normalized linear distribution of electrical potential through the thickness of the piezoelectric layers is given by equations (16) and (17). The Ritz expansion of the electrical potential for the laminated beam is therefore given in equation (15). The signs of the trial functions of (16) and (17) are opposite since the locations of the piezoelectric layers are mirrored about the middle plane of the laminated beam

$$\varphi(z,t) = \mathbf{\Phi}(\mathbf{z})\mathbf{v}(\mathbf{t}) = \begin{bmatrix} \Phi_1(z) & \Phi_2(z) \end{bmatrix} \begin{bmatrix} v_1(t) \\ v_2(t) \end{bmatrix} = \sum_{q=1}^2 \Phi_q v_q$$
(15)

$$\Phi_{\rm I}(z) = \begin{cases} \frac{z - h_s - h_c}{h_p}; & h_s + h_c \leqslant z \leqslant h_s + h_p + h_c \\ 0 & \text{elsewhere} \end{cases}$$
(16)

$$\Phi_2(z) = \begin{cases} -\frac{z + h_s + h_c}{h_p}; (h_s + h_p + h_c) \leqslant z \leqslant -h_s - h_c \\ 0 & \text{elsewhere.} \end{cases}$$
(17)

Here  $v_q$  represents the voltage at the electrode of the piezo-electric layer q.

Substituting equations (2), (11), and (15) into the (5), (9), and (10), and then applying the Euler–Lagrange equation (18) with respect to the mechanical or electrical generalized coordinates, which are represented by  $\xi_m$ , the governing equations for the nonlinear energy harvesting system are

obtained as shown in equation (19). Here, Rayleigh proportional damping is assumed and nonlinear coupling between the mechanical and electrical dynamics are neglected

$$\frac{\partial}{\partial t} \left[ \frac{\partial (T - U + W)}{\partial \dot{\xi}_m} \right] - \frac{\partial (T - U + W)}{\partial \xi_m} = 0$$
 (18)

$$[M]\ddot{\mathbf{r}} + [D]\dot{\mathbf{r}} + [K]\mathbf{r} + [K_1]\mathbf{r} + \mathbf{F}_{NL} + [\Theta]\mathbf{v} = \mathbf{f}_{\mathbf{m}} \quad (19a)$$

$$-[\Theta]\dot{\mathbf{r}} + [C_p]\dot{\mathbf{v}} + [i_p] = 0. \tag{19b}$$

The nonlinear term is given in (20) and  $[i_p]$  is the current through the piezoelectric layer.

$$[\mathbf{F}_{NL}]_m = \sum_{i=1}^{M} \sum_{j=1}^{M} \sum_{k=1}^{M} [K_3]_{mijk} r_{mi} r_{mj} r_{mk}.$$
 (20)

The matrices in equation (19) are listed in equation (21)

$$[M]_{mi} = \int_{V_s} \rho_s \psi_m \psi_i dV + \int_{V_c} \rho_c \psi_m \psi_i dV + \int_{V_p} \rho_p \psi_m \psi_i dV + \int_{V_e} \rho_e \psi_m \psi_i dV + M_0 \left( \psi_m |_{x=L} + d_c \frac{\partial \psi_m}{\partial x} |_{x=L} \right) \times \left( \psi_i |_{x=L} + d_c \frac{\partial \psi_i}{\partial x} |_{x=L} \right)$$
(21a)

$$[K]_{mi} = \int_{V_s} E_s z^2 \left[ \frac{d^2 \psi_m}{dx^2} \frac{d^2 \psi_i}{dx^2} \right] dV$$

$$+ \int_{V_c} E_c z^2 \left[ \frac{d^2 \psi_m}{dx^2} \frac{d^2 \psi_i}{dx^2} \right] dV$$

$$+ \int_{V_p} E_p z^2 \left[ \frac{d^2 \psi_m}{dx^2} \frac{d^2 \psi_i}{dx^2} \right] dV$$

$$+ \int_{V_c} E_e z^2 \left[ \frac{d^2 \psi_m}{dx^2} \frac{d^2 \psi_i}{dx^2} \right] dV$$
(21b)

$$[D]_{mi} = \alpha [M]_{mi} + \beta [K]_{mi} \tag{21c}$$

$$[K_{1}]_{mi} = -k_{1} \left( \psi_{i}|_{x=L} + d_{mc} \frac{\partial \psi_{i}}{\partial x} \Big|_{x=L} \right) \times \left( \psi_{m}|_{x=L} + d_{mc} \frac{\partial \psi_{m}}{\partial x} \Big|_{x=L} \right)$$
(21d)

$$[\Theta]_{mq} = -\int_{V_p} e_{31} z \frac{d\Phi_q}{dz} \frac{d^2 \psi_m}{dx^2} dV$$
 (21e)

$$[K_{3}]_{mijk} = \int_{V_{s}} E_{s}z^{2} \left( \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d\psi_{j}}{dx} \frac{d\psi_{k}}{dx} \frac{d^{2}\psi_{m}}{dx^{2}} \right) dV$$

$$+ \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d^{2}\psi_{j}}{dx^{2}} \frac{d\psi_{k}}{dx} \frac{d\psi_{m}}{dx} \right) dV$$

$$+ \int_{V_{c}} E_{c}z^{2} \left( \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d\psi_{j}}{dx} \frac{d\psi_{k}}{dx} \frac{d^{2}\psi_{m}}{dx^{2}} \right) dV$$

$$+ \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d^{2}\psi_{j}}{dx^{2}} \frac{d\psi_{k}}{dx} \frac{d\psi_{k}}{dx} \frac{d^{2}\psi_{m}}{dx^{2}} dx^{2}$$

$$+ \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d^{2}\psi_{j}}{dx^{2}} \frac{d\psi_{k}}{dx} \frac{d\psi_{m}}{dx} \frac{d\psi_{m}}{dx} dx^{2}$$

$$+ \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d^{2}\psi_{j}}{dx^{2}} \frac{d\psi_{k}}{dx} \frac{d\psi_{m}}{dx} \frac{d\psi_{m}}{dx} dx^{2}$$

$$+ \frac{d^{2}\psi_{i}}{dx^{2}} \frac{d^{2}\psi_{j}}{dx^{2}} \frac{d\psi_{m}}{dx} \frac{d\psi_{m}}{dx} \frac{d\psi_{m}}{dx} \frac{d\psi_{m}}{dx} \frac{d\psi_{$$

$$[\mathbf{f_m}]_m = \int_{V_s} \rho_s a_b \psi_m dV + \int_{V_c} \rho_c a_b \psi_m dV + \int_{V_p} \rho_p a_b \psi_m dV + \int_{V_e} \rho_e a_b \psi_m dV + M_0 a_b \left( \psi_m |_{x=L} + d_c \frac{\partial \psi_m}{\partial x} |_{x=L} \right)$$
(21g)

$$[C_p]_{nq} = \varepsilon_3^s \int_{V_p} \frac{d\Phi_q}{dz} \frac{d\Phi_n}{dz} dV. \tag{21h}$$

From the governing equations (19), the terms  $[K_1]$  and  $\mathbf{F}_{NL}$  primarily result from the direct influence of the repulsive magnets. By changing the magnet gap  $d_2$ , the system may take on a monostable or bistable configuration. For a monostable configuration, there is one statically stable equilibrium. According to the influence of the magnet gap on the matrix  $[K_1]$ , the resonance frequency reduces with decrease of the magnet gap. This trend continues to a point. For still smaller magnet gap  $d_2$ , two stable equilibria occur and the harvester becomes bistable. In this way, the dynamic responses that may occur depend on the amplitude and frequency of the base acceleration. These dynamics responses may be snap-through, aperiodic, and intrawell vibration. The differences among these three kinds vibration are described in [22].

#### 2.2. Solutions to the nonlinear system of governing equations

The two piezoelectric layers are in parallel. As a result, the voltage of each piezoelectric layer satisfies

$$v_1 = v_2 = v_p. (22)$$

The governing equations in equation (19) further simplify to be

$$[M]\ddot{\mathbf{r}} + [D]\dot{\mathbf{r}} + [K]\mathbf{r} + [K_1]\mathbf{r} + \mathbf{F}_{NL} + [\Theta_1]v_p = \mathbf{f}_{\mathbf{m}}$$
(23a)

$$-[\Theta_1]^T \dot{\mathbf{r}} + C_{p1} \dot{\mathbf{v}}_p = -i, \tag{23b}$$

where  $[\Theta_1]$  is electromechanical coupling associated with the generalized coordinates,  $C_{p1}$  is the internal capacitance of the piezoelectric beam, i and  $v_p$  are the current and voltage in the harvesting circuit as shown in figure 1(a).

The base acceleration that drives the energy harvester is assumed to have moderate amplitude such that nonlinear harmonics are weakly induced. Thus, the displacement response frequency of the beam is assumed to coincide with the base acceleration frequency  $\omega$  [31, 32]. Therefore, the mechanical response are expressed

$$\mathbf{r}(t) = \mathbf{k}(t) + \mathbf{h}(t)\sin \omega t + \mathbf{g}(t)\cos \omega t. \tag{24}$$

Principles of harmonic or stochastic linearization are then utilized to linearize the governing equations (23), [43, 44]. The linearized governing equations are shown in equation (25)

$$[M]\ddot{\mathbf{r}} + [D]\dot{\mathbf{r}} + ([K] + [K_1] + [K_e]^{(k_3)})\mathbf{r} + [\Theta_1]v_p = \mathbf{f_m}$$
(25a)

$$-[\Theta_1]^T \dot{\mathbf{r}} + C_{p1} \dot{\mathbf{v}}_p = -i. \tag{25b}$$

The equivalent linear stiffness matrix accounting for the nonlinearity is given in (26)

$$[K_{e}]_{mh}^{(k_{3})} = \left\langle \frac{\partial [\tilde{\mathbf{F}}_{NL1}]_{m}}{\partial r_{h}} \right\rangle = \sum_{j=1}^{M} \sum_{k=1}^{M} ([K_{3}]_{mhjk} + [K_{3}]_{mjhk} + [K_{3}]_{mkjh}) \times \left[ k_{j}k_{k} + \frac{1}{2} (h_{j}h_{k} + g_{j}g_{k}) \right].$$
(26)

The  $\langle \rangle$  indicates the mathematical expectation operator.

When a rectifier with an RC circuit interfaces with a forced piezoelectric beam, the voltage across the piezoelectric layer electrodes is calculated by (27) [31]

$$v = \left[ \frac{-[\Theta_1]^T \mathbf{g}}{C_{p_1} \pi} \sin^2 \beta + \frac{[\Theta_1]^T \mathbf{h}}{2C_{p_1} \pi} (2\beta - \sin 2\beta) \right] \sin \omega t$$
$$+ \left[ \frac{[\Theta_1]^T \mathbf{h}}{C_{p_1} \pi} \sin^2 \beta + \frac{[\Theta_1]^T \mathbf{g}}{2C_{p_1} \pi} (2\beta - \sin 2\beta) \right] \cos \omega t$$
(27)

$$\beta = \arccos\left(\frac{\pi - 2\omega C_{p1}R}{\pi + 2\omega C_{p1}R}\right). \tag{28}$$

Substituting equations (24) and (27) into the governing equation (25), the coefficients for the sinusoidal terms in equation (24) are found from the equations (29) and (30).

$$-\omega^{2}[M]\mathbf{h} - \omega[D]\mathbf{g} + ([K] + [K_{1}] + [K_{e}]^{(k_{3})})\mathbf{h}$$
$$-\frac{\sin^{2}\beta}{C_{p1}\pi}[\Theta_{1}][\Theta_{1}]^{T}\mathbf{g} + \frac{(2\beta - \sin 2\beta)}{2C_{p1}\pi}[\Theta_{1}][\Theta_{1}]^{T}\mathbf{h} = 0$$
(29)

$$-\omega^{2}[M]\mathbf{g} + \omega[D]\mathbf{h} + ([K] + [K_{l}] + [K_{e}]^{(k_{3})})\mathbf{g}$$

$$+ \frac{\sin^{2}\beta}{C_{p_{1}}\pi}[\Theta_{l}][\Theta_{l}]^{T}\mathbf{h} + \frac{(2\beta - \sin 2\beta)}{2C_{p_{1}}\pi}[\Theta_{l}][\Theta_{l}]^{T}\mathbf{h} = \mathbf{f_{m}}.$$
(30)

By integrating the governing equations (25) over one period of the harmonic excitation force  $2\pi/\omega$ , the equations for the constant terms **k** in equation (24) are obtained, as shown in equation (31)

$$\{[K] + [K_1]\}\mathbf{k} + \kappa^{(k_3)} = 0$$

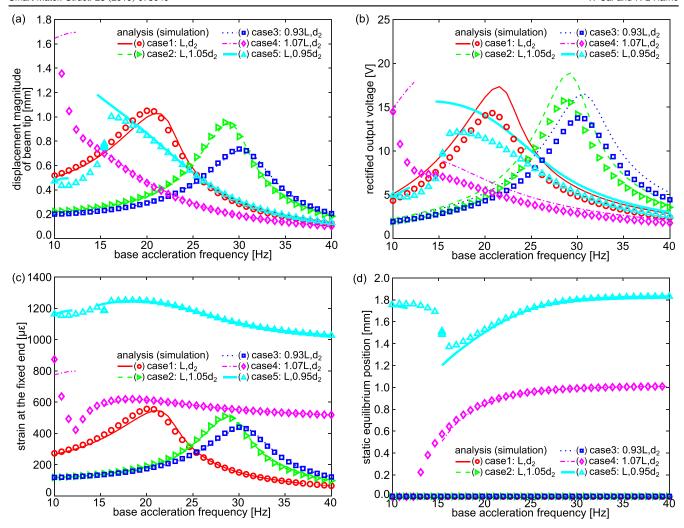
$$[\kappa^{(k_3)}]_m = \sum_{i=1}^M \sum_{j=1}^M \sum_{k=1}^M [K_3]_{mijk} k_i k_j k_k + \frac{1}{2} ([K_3]_{mijk} + [K_3]_{mjik} + [K_3]_{mjki}) \times k_i [g_j g_k + h_j h_k].$$
(32)

By simultaneously solving the equations (29)–(31), the generalized coordinates are found enabling determination of the physical responses via equations (11) and (15).

For other nonlinear interface circuits, such as buck-boost converter or synchronized switch harvesting on inductor electronic interface, with an expression of the output voltage  $v_p$  across the piezoelectric beam determined through the respective derivation [32, 45, 46], the procedures described above may be applied for the responses analysis.

#### 3. Numerical verification and comparison to analysis

In order to verify the accuracy of the approximate analytical solution to the nonlinear system of governing equations, fourthorder Runge-Kutta numerical integration is undertaken for the equation system (23). Due to the potential for multiple steadystate dynamic behaviors, 10 separate simulations are taken using normally distributed and randomly selected initial conditions (i.e. displacements, velocities, and voltages) for each combination of base acceleration and system parameters. The simulation duration is set to be 400 periods of the harmonic excitation to ensure steady-state conditions develop. The parameters of the system are given in tables 1 and 2. The beam length L and gap between repulsive magnets  $d_2$  are taken as the outcomes of the optimization, discussed in detail in section 4, but it is sufficient at this stage to use these key parameters for contrast to the simulation. Cubic magnets with side lengths of 6.35 [mm] are used to introduce nonlinearity. The mass of the magnet and mass of the holder contribute to the total mass at the cantilever tip, which is 15 g. Table 2 provides the properties for layers of the laminate beam. In addition to the nominally optimal parameter combination of harvester beam design studied, additional parameter combinations are examined here to comprehensively verify the analytical model. These four cases involve a combination of 7% increase or decrease in beam length L and 5% increase or decrease in magnet gap  $d_2$ .



**Figure 2.** Frequency responses of (a) beam tip displacement amplitude, (b) rectified output voltage across the resistor, (c) strain of the beam at the fixed end, and (d) static equilibrium position.

Table 1. Parameters for system design and excitation conditions.

L (mm)	$b_0, b_L(mm)$	$h_s$ (mm)	$h_p$ (mm)	$h_c$ (mm)	$d_1$ (mm)	$d_c$ (mm)	$d_{mc}$ (mm)	
37.95 d <sub>2</sub> (mm)	$32$ $M_0$ (g)	$0.04$ $M_a \text{ (MA m}^{-1}\text{)}$	$0.19 V_{mag} \text{ (cm}^3\text{)}$	$a_b \text{ (m s}^{-2}\text{)}$	$\begin{array}{c} 10 \\ L_p \text{ (mm)} \end{array}$	6.58 <i>b<sub>p</sub></i> (mm)	8.9 $R (k\Omega)$	$C_L (\mu F)$
10.2	15	1.61	0.768	5	27.8	18	100	10

**Table 2.** Material properties.

$E_s(GPa)$	$E_p(GPa)$	$E_c(GPa)$	$E_e(GPa)$	$\rho_s ~(\mathrm{kg}~\mathrm{m}^{-3})$	$\rho_p~({\rm kg~m}^{-3})$	$\rho_c ~(\mathrm{kg}~\mathrm{m}^{-3})$	$\rho_e~({\rm kg~m^{-3}})$	$\varepsilon_{33} \; (\mathrm{nF}  \mathrm{m}^{-1})$	$e_{31} ({\rm C m}^{-2})$
26	60.6	128	3.5	1900	7800	8940	1540	13.16	-16.6

Given the  $5 \,\mathrm{m\,s^{-2}}$  amplitude harmonic base acceleration identified in table 1, figure 2 shows the piezoelectric cantilever responses in terms of (a) displacement amplitude at the beam tip, (b) rectified output voltage, (c) strain at the fixed end on the bottom beam surface, and (d) static equilibrium position. Analytical results are shown as curves whereas simulation results are the open markers. From figure 2(d), when the magnet gap decreases (case 5) or beam length

increases (case 4) from the nominal case 1 design parameters, non-zero static equilibria appear. This indicates that for cases 4 and 5 an increase in beam length and a decrease in magnet gap result in a bistable energy harvester configuration. At higher frequencies of base acceleration, the residual strain caused by the non-zero static equilibria contributes to the greater total strain shown in figure 2(c), despite exhibiting small amplitudes of displacement, figure 2(a).

**Table 3.** Analytical and simulation results comparison.

	Case 1		Case 2		Case 3	
	Analysis	Simulation	Analysis	Simulation	Analysis	Simulation
Bandwidth (Hz)	11.86	12.08	9.46	9.52	11.74	11.79
RMS value of rectified voltage (V) RMS value of strain ( $\mu \varepsilon$ )	13.87 420.84	10.7 427.65	13.87 376.27	11.52 378.11	12.43 332.62	10.52 333.9

The table 3 quantitatively compares the analytical and simulation results from figure 2. The frequency bandwidth shown in table 3 is determined by the frequency range in the harmonic voltage responses over which the voltage amplitude is greater than or equal to one-half of the maximum computed voltage amplitude. The corresponding root mean square (RMS) values are calculated inside this frequency bandwidth.

As shown in table 3, the differences between the analytical and simulation results for the frequency bandwidth of large amplitude dynamics and RMS strain at the clamped end are less than 2%. Based on the derivation in section 2, only the fundamental harmonic term is employed to approximate the system responses, thus neglecting diffusion of energy to higher order harmonics, which contributes to the overprediction of the rectified RMS voltage by the analysis as seen in table 3. Despite this discrepancy, from figure 2(b) the voltage responses of the first three cases show good agreement for the influence of 5% decrease in magnet gap and 7% decrease in beam length. On the other hand, for the bistable cases (cases 4 and 5) in figure 2(a), the harmonic response and steady state assumptions of analysis lead to less agreement with simulation due to the chaotic responses found by numerical simulation that are unable to be reproduced by the harmonic analysis. At higher frequencies of the base acceleration, figure 2 uniformly reveals good quantitative agreement between simulation and analysis, where the significance of nonlinear response is relatively low.

Overall, comparing with the simulation results, the analytical model captures the main characteristics of the non-linear mechanical and electrical system for both monostable and bistable configurations resulting from the key parameter shifts of beam length and magnet gap.

### 4. Optimization for a robust energy harvester system via GA

Multiple attributes of an energy harvesting system contribute to a robust system design for persistent DC power delivery. To understand how to design parameters distinctly participate to yield exceptional overall performance for the laminated piezoelectric cantilever, here a multi-objective, GA optimization method is established and then utilized to scrutinize what combinations of design elements give rise to optimality.

#### 4.1. Multi-objective optimization constraints and objectives

To consider a practical design problem, several overall constraints are set in the exploration of feasible design

parameters. The space to install energy harvesters in application is likely confined [47]. The maximum length  $L_{\rm max}$ , width  $b_{\rm max}$ , and thickness  $t_{\rm max}$  of the maximum operating volume are set to be 70 mm, 40 mm, and 4 mm, respectively. The thickness extent is the most limited since the beam vibrates in this axis. To generally minimize dynamic strain and prolong working life, the peak amplitude of displacement must be limited. Specifically, the global constraints based on these limitations for the optimization are as follow:

- 1. The sum of the beam length L, holder extension  $d_1$ , and magnet gap  $d_2$  must be less than maximum length  $L_{\text{max}}$ .
- 2. The beam length must be greater than 30 mm and less than 60 mm to warrant assumptions of vibration like a beam in contrast to a plate.
- 3. The displacement amplitude at the beam tip must be less than  $t_{\text{max}}/2$  to not exceed the maximum permitted thickness of the operating volume.
- 4. The magnet gap  $d_2$  must be less than 20 mm and greater than 5 mm, to minimum unnecessary cases of extreme nonlinearity.
- 5. The displacement amplitude at the beam tip must be greater than 1 mm, otherwise GA searches use excessive time for irrelevant and poor-performing combinations of design parameters.

The following three objectives are utilized for the multiobjective optimization considered in this report.

- 1. Frequency range  $f_{range}$ . The  $f_{range}$  is calculated from the voltage amplitude based on the concept of frequency bandwidth defined in section 3. For a bistable configuration of the energy harvester, only the snapthrough response is considered to determine the frequency range.
- 2. Total RMS of rectified output voltage  $V_{rms}$  in  $f_{range}$ . For a monostable system, the values inside the frequency range  $f_{range}$  are included in the calculation. For a bistable system, only the values associated with snapthrough inside the frequency range contribute to the cost function of  $V_{rms}$ .
- 3. Total RMS of strain at the clamped beam end on the bottom surface  $S_{rms}$  in  $f_{range}$ . The calculation is similar to the procedures described in the objective 2. The difference occurs when the system becomes bistable. Because the existence of non-zero static equilibrium positions will cause residual strain in the whole frequency range. The RMS strain for bistable configurations include the contributions from the residual strain and the dynamic strain amplitude.

The multi-objective optimization is formally defined by a weighted sum of the individual objectives [48]. A non-dimensionalization is applied before the summation of components. The linear response for a piezoelectric cantilever is adopted to obtain the nondimensionalization constants. Since the bandwidth for a linear response is narrow when compared to the nonlinear dynamic behavior, the resonance frequency  $f_0$  of the linear piezoelectric cantilever is used to normalize the  $f_{range}$ . The voltage  $V_0$  at resonance for the linear piezoelectric cantilever is used to normalize  $V_{rms}$ . Finally, the strain  $S_0$  at the fixed end on the bottom surface of the linear piezoelectric cantilever at resonance is used to normalize  $S_{rms}$ . These values are found to be

$$f_0 = 25 \text{ Hz}, V_0 = 65 \text{ V}, S_0 = 1365 \,\mu\varepsilon.$$
 (33)

The total objective employed for optimization is

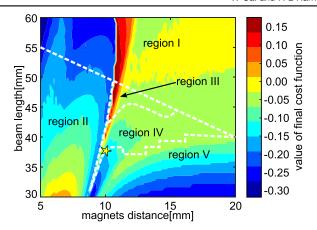
$$cost = -w_f \frac{f_{range}}{f_0} - w_v \frac{V_{rms}}{V_0} + w_s \frac{S_{rms}}{S_0}$$
 (34)

$$w_f + w_v + w_s = 1. (35)$$

The  $w_f$ ,  $w_v$ , and  $w_s$  are respectively weights for the objectives of frequency range, rectified output voltage, and strain condition. Since an energy harvester can deliver high output power in a wide frequency range with a reasonable low strain condition is preferred, with the total objective defined in equation (34), the smallest value of the total objective indicates the optimal design.

The GA optimization developed for this research is inspired through the specific strategies described in Haupt and Haupt [48]. A total of 100 generations including populations with 80 individuals are used for broad evaluation of the parameter space. Each individual is a combination of the design parameters defined for the several optimization studies considered in the following sections. Each individual is evaluated for the three distinct objectives: frequency range, RMS voltage, and RMS strain. The cost function for each individual is then computed by equation (34) according to the multi-objective weighting. The initial population of 80 individuals is generated through random selection inside the constraint range of each design parameter. After evaluating and ranking each individual in the population on the basis of cost function value, the best 50% of the population is selected as parents to generate 40 offspring for the next generation. The remaining 50% of the next generation is the best 50% of the prior population. A mutation rate of 20% is also included to randomly mutate 20% of the next generation to ensure the diversity inside each population.

In the following sections unless otherwise indicated, the parameters given in tables 1 and 2 are employed. The base acceleration frequency range is from 10 to 40 Hz with a constant amplitude 5 m s<sup>-2</sup>, which is a characteristic of ambient vibrations in heavy industry or automotive applications. Three optimization studies are undertaken to uncover the origins of optimal robust DC power delivery from the laminated piezoelectric cantilever to a fixed load condition. The detailed influence of the resistance on the performance of the piezoelectric energy harvester may be found in [32, 49].



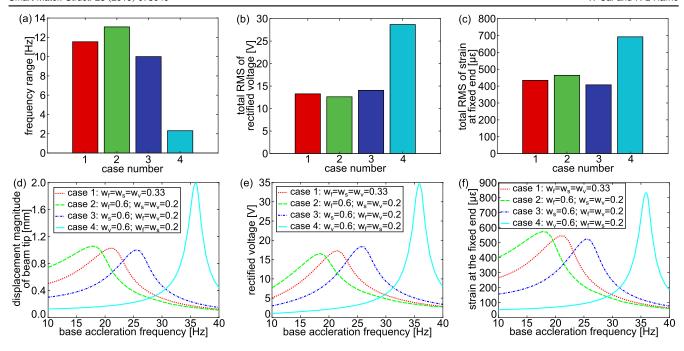
**Figure 3.** Cost function values of different combination of beam length and magnet gap.

#### 4.2. Optimization results and discussion

4.2.1. Optimization 1: beam length L and magnet gap  $d_2$ . A first optimization problem with the design parameters of beam length L and distance between repulsive magnets  $d_2$  is considered to understand how the nonlinear forces associated with magnetic repulsion influence the ideal energy harvesting beam design.

When the multi-objective cost function weights  $w_f$ ,  $w_v$ , and  $w_s$  are the same value (1/3), a system with beam length L = 37.95 mm and magnet gap  $d_2 = 10.17$  mm is identified as optimal. This parameter combination is shown as the star marker in figure 3. Figure 3 presents the values of the cost function as the shading for a wide range of combinations of beam length and magnet gap. Region I conflicts with constraint 1. Regions II and V are neglected because they violate constraint 5. For harvesters designs in region II, with the decrease of the magnet gap, more energy is demanded to pass the potential barrier to realize snap-through behavior. Therefore, in region II only small amplitude of intrawell behaviors are possible. Because of the residual stress caused by the non-zero static equilibria, the intrawell responses result in large strain and low output voltage. For designs inside region V, the resonance frequency of the system lies sufficiently outside the frequency range of interest in this research, which results in low rectified output voltage flow. In contrast, region III includes the designs with displacement amplitude larger than 2 mm, which compromises longevity of the harvester and conflicts with the volumetric space constraints. The star label resides in the region IV, where all combinations of the magnet distance and beam length satisfy the optimization constraints.

The results of this optimization may be assessed in relation to the findings from the numerical verification of the analytical model in figure 2. The case 1 in figure 2 provided the optimal result obtained in this section 4.2.1 optimization when all weights are the same. For cases 4 and 5, the length of the laminated cantilever is increased or the magnet gap is decreased, enhancing the nonlinear interaction between repulsive magnets and causing bistability. The bistability is seen to result in high strain at the clamped end of the



**Figure 4.** Cost function values of (a) frequency range, (b) total RMS of the rectified output voltage inside the frequency range, and (c) total RMS of the strain at the fixed end on the bottom surface inside the frequency range. Corresponding responses of (d) displacement amplitude of the beam tip, (e) rectified output voltage, and (f) strain at the fixed end on the bottom surface.

cantilever figure 2(c), which practically reduces the working life of the harvester due to the susceptibility to early failure.

The optimal design case 1 and the sub-optimal designs of cases 2 and 3 are monostable, such that the resting equilibria have zero displacement of the beam tip. The findings of figure 2 thus show that a decrease in beam length or increase in magnet gap is desirable to sustain a monostable, and therefore long-life platform design, if the optimal design parameters are unable to be perfectly achieved.

In order to study the influence of weights in this optimization problem, three more optimization problems are considered. Here, the weights assigned to compute the cost function values are differed. Case 2 corresponds to the case that  $w_f$ ,  $w_v$ , and  $w_s$  are respectively 0.6, 0.2, and 0.2. Case 3 corresponds to the case that  $w_f$ ,  $w_v$ , and  $w_s$  are respectively 0.2, 0.2, and 0.6 for  $w_f$ ,  $w_v$ , and  $w_s$ . Finally, case 4 corresponds to the case that  $w_f$ ,  $w_v$ , and  $w_s$  are respectively 0.2, 0.6, and 0.2. These unique cases make the individual cost function objectives relatively dominant: (case 2) frequency range  $f_{range}$ , (case 3) strain condition  $S_{rms}$ , and (case 4) rectified voltage  $V_{rms}$ .

The optimal designs identified for the three cases are respectively  $L=37.37\,$  mm,  $d_2=9.94\,$  mm for case 2,  $L=38.38\,$  mm,  $d_2=10.51\,$  mm for case 3, and  $L=43.86\,$  mm,  $d_2=16.13\,$  mm for case 4. Figures 4(a)–(c) present the cost function values computed for harvesters optimized with the different weighted cases. Figures 4(d)–(f) respectively show the responses of beam tip displacement, rectified voltage, and strain at the cantilever end of four cases. Since the magnet gap is more influential in introducing nonlinearity, when the frequency range is dominant (case 2) in the optimization, a smaller magnet gap is optimized to introduce greater nonlinearity that leads to a wider frequency range. Comparatively, when the strain condition is

dominant (case 3), the magnet gap increases to realize the nonlinear resonant behaviors at a comparatively higher frequency as shown in figure 4(d). The responses of the cases 1, 2, and 3 in figures 4(e) and (f) indicate an increase in output voltage at high frequencies of base acceleration due to a shifting of the nonlinear resonance to higher frequencies even with a smaller strain at the cantilever end. This explains why the optimal magnet gap in case 3 is greater than those for cases 1 and 2. For a similar reason, when the voltage is dominant (case 4), the magnet gap is further increased with a much longer beam length to highly increase the rectified output voltage by increased resonant frequency. In comparison with the other three cases, the case 4 optimization demonstrates a tremendous drop in frequency range and increase in strain and output voltage in figures 4(a)-(c). From the responses shown in figure 4(d), the optimal design in case 4 is more similar to a linear configuration for its narrow bandwidth, which illustrates the sensitivity of the voltage objective to the weights change.

4.2.2. Optimization 2: beam length L, magnet gap  $d_2$ , and beam width at free end  $b_L$ . Multiple investigations have concluded that a trapezoidal shape for a piezoelectric cantilever results in more uniform strain distribution and often increase in output power [9, 15]. Therefore, the beam width at the free end  $b_L$  of the laminated piezoelectric cantilever is introduced as an additional design variable for the optimization. The additional constraint for  $b_L$  is that it may vary from zero to the maximum width value, which is 40 mm as stated in section 4.1. Four optimization cases, having the same variation of objective weights employed in section 4.2.1, are considered here. As in section 4.2.1, these

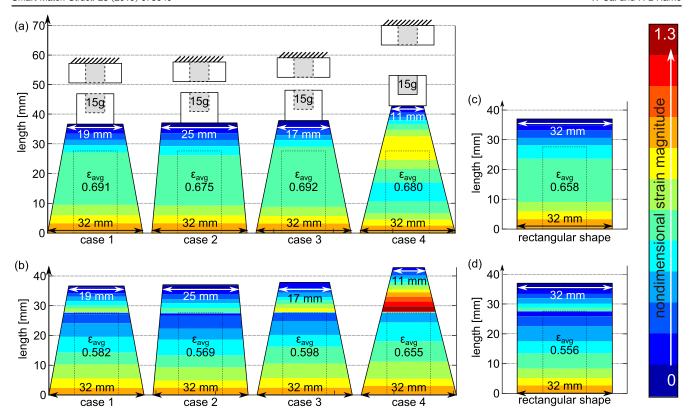


Figure 5. (a) Strain contour plot from analysis and (b) strain contour plot from finite element simulation of optimal designs. (c) Strain contour plot from analysis and (d) strain contour plot from finite element simulation of a rectangular shape for comparison.

cases are also referred to as cases 1, 2, 3, and 4 based on the weights used in the optimization.

The optimal designs obtained for the four cases including the beam length, beam width, and magnet gap are schematically shown in figure 5(a). The black dotted curve overlaid on each beam shape schematic is the outline of the piezoelectric layer. The grey squares represent the positions of the magnets, and the white rectangles are the magnet holders. The clamp location for the beams corresponds to the bottom of each respective schematic, so that the holder and magnet are installed at the opposite end. The repulsive magnets are positioned opposite of the magnet holder and the magnet gap is identified in the figure 5 according to the spacing shown. The mass labeled is the total tip mass, which is the combination of the magnet and the holder mass. The strain distribution of the cantilever on the bottom surface is also shown as the shaded contour in figure 5. The magnitude of the strain is nondimensionalized by the strain at the fixed end. The average values of the nondimensional strain  $\varepsilon_{avg}$  are reported on the surfaces and correspond to the average nondimensional strain over the area covered by the PZT-5H layer.

Figure 5(a) presents the analytical reconstruction of the strain distributions whereas figure 5(b) presents strain results obtained from finite element simulations (ABAQUS) of the piezoelectric cantilevers having the same optimized geometries, layering, and material properties. The schematics of the magnets in figure 5(b) are omitted for sake of brevity. Since in the analysis the strain is assumed to be uniform through the

beam width and equal to the axial strain, the finite element results are also presented with the axial strain mapped equally over the beam width. Here, both the analytical and finite element strain distribution for a rectangular beam shape are respectively provided in figures 5(c) and (d) for comparison. The dimensions of the rectangular beam include a constant beam width of 32 mm along the beam axis.

Comparing the finite element and analytical results in figures 5(a) and (b), the strain distribution around the end of the PZT cannot be exactly captured in the analysis due to the use of a limited number of trial functions. This is especially evident for case 4 in figure 5(a), where the finite element simulation figure 5(b) predicts a greater nondimensional strain at the position of the discontinuity than that suggested by the analysis in figure 5(a). Despite the relatively minor discrepancies, both analytical and finite element results demonstrate that a more uniform strain distribution is achieved over the tapered beam shape. The analysis and simulations both agree that the tapered beam strain distributions, figures 5(a) and (b), are more uniform when compared to the corresponding rectangular beams, figures 5(c) and (d). A greater average nondimensional strain over the piezoelectric layers indicates that the tapered beams deliver comparatively higher output electrical power to the loads than the rectangular beams. Therefore, considering the four cases of optimization, whether all objectives are equally weighted (case 1) or the individual objectives are set to be dominant (cases 2, 3, and 4), the optimization results show that tapered beams with strategic beam length and magnet gap

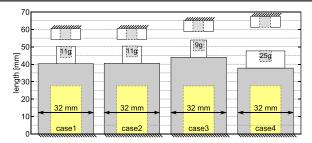


Figure 6. Optimal designs for optimization 3.

deliver improved global performance. When the strain condition (case 3) or rectified voltage (case 4) objective is dominant in the optimization, a higher taper ratio defined as  $b_L/b_0$  is optimized. This further supports the conclusion that tapered beam shapes deliver higher power by the more uniform strain distribution, even though the piezoelectric layers only cover part of the beam. In addition, comparing with the optimal designs of cases 1 and 2, the magnet gap increases for the cases 3 and 4. This is especially evident for case 4, where the magnet gap is greatly increased to reduce nonlinearity and promote more linear response behavior. This agrees with the conclusion in section 4.2.1.

4.2.3. Optimization 3: beam length L, magnet gap  $d_2$ , and tip mass  $M_0$ . Since applications often envision collocation of the harvester with the powered microelectronics, it is desirable to limit the harvester mass to lead to a compact, lightweight energy harvesting platform for system integration. In the third optimization study, the beam length, magnet gap, and tip mass are taken as design parameters to optimize. Here a maximum tip mass 25 g is taken as additional constraint. Four cases with different weights setting as stated in section 4.2.1 are studied.

The optimal designs for the four cases are schematically shown in figure 6. As described in section 4.2.2, the light grey square and the white rectangle represent the magnet and the holder. The mass labels are the total magnet and holder mass. Since the extended length of the magnet holder at beam free end are constant, the shape or the material of the holder may change to ensure the optimal tip mass. In the schematics, the holder shape is changed to illustrate a change of optimized tip mass. The additional characteristics schematically shown in figure 6 are similar to those in figure 5. As first observed through the optimization of section 4.2.1, the magnet gap greatly increases for case 4 to promote linear dynamic behavior and greatest RMS voltage in a narrow frequency range. For the cases 1, 2, and 3 in figure 6, sufficient nonlinearity results from the repulsive magnet interaction to lead to a monostable configuration. Additionally, since the tip mass is directly related to the external force induced at the beam tip, when considering the tip mass as a design variable a larger mass corresponds to a greater tip force that increases the strain along the beam length and correspondingly delivers high output power. Therefore, when emphasizing the minimization of strain by case 3, the optimal design converges to a small tip mass shown in figure 6. In contrast, when the rectified output voltage becomes the dominant factor in the optimization shown by case 4, the maximum allowable tip mass is selected with a short beam length to lead to a higher frequency of resonance, as similarly observed in section 4.2.1. For the cases 1 and 2, the optimized tip mass results from the respective balance of the multiple objectives in the cost function evaluations.

From the corresponding cost function values shown in figure 7, the change in the tip mass causes an up to 27 V change in the RMS voltage, while the RMS strain increases up to 900  $\mu\varepsilon$  for case 4. This confirms the significant influence of the tip mass selection on the output voltage and strain at the cantilever fixed end. In contrast, the influence of tip mass on optimizing the frequency range is weak since the beam length and magnet gap are primarily shifted to promote a broader frequency range of the nonlinear resonant behavior. These results show that it is valuable to include the tip mass as a design variable in optimization to ensure that a combination of harvester design parameters is optimized having overall greatest lifetime and best power delivery.

From all optimal designs for case 4 in the three optimization problems considered in this section 4, when the voltage objective is dominant in the optimization extreme design conditions are reached. In other words, case 4 results in the maximum permitted beam length in optimization 2 and the maximum tip mass in optimization 3. The frequency responses of the optimal designs for the optimized results for case 4 are similar to linear responses, as seen by the examples in figures 4(d)-(f). Based on the narrowband resonant behavior, such linear systems may easily lose superiority of performance when excitation conditions change. Therefore, in an optimization of an energy harvesting system, a strict optimization on the basis of maximizing output voltage is not recommended. Comparatively, preferring optimizing on the basis of frequency range, strain at the fixed beam end, or the more balanced multi-objective optimization lead to robust energy harvesters that may serve in practical environments.

#### 5. Experimental system description

The key findings from the analytical model and optimization investigations are then validated through controlled experiments. Based on the optimal designs of the laminated piezoelectric cantilevers determined through cases 1–3 in section 4.2, three beams shown in figure 8(a) are cut from off-the-shelf piezoelectric energy harvesters by Midé Technology. The shapes are cut using a CNC router. Due to limitations of creating the cut-out cantilever shapes, the optimal geometries are unable to be exactly achieved although they are closely emulated. The beams 1, 2, and 3 shown in figure 8(a) have experimentally identified parameters given in table 4, while the remaining parameters not given are provided in table 2.

Figure 8(b) presents the experimental setup used to examine the three piezoelectric cantilevers. For each experiment, the piezoelectric beam is clamped to an aluminum frame mounted to an electrodynamic shaker table (APS Dynamics 400). At the free end of the piezoelectric beam, an aluminum magnet holder is installed. An opposing magnet holder is affixed to the shaker table at a location immediately adjacent to the beam tip along the

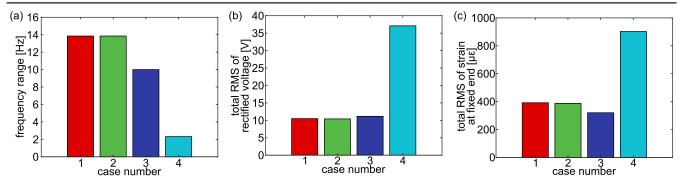


Figure 7. Cost function values of four optimal designs. (a) Frequency range. (b) Total RMS of the output voltage. (c) Total RMS of the strain at the fixed end

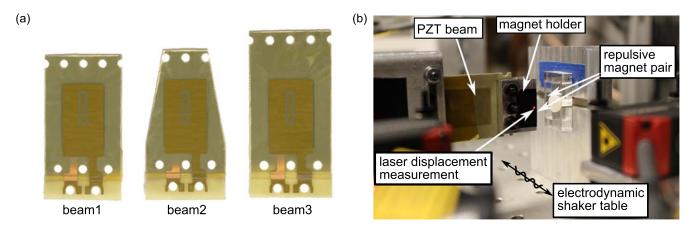


Figure 8. (a) Three beams shapes for experiments. (b) Photograph of the experiment platform.

**Table 4.** Parameters for three beams identified from experiments.

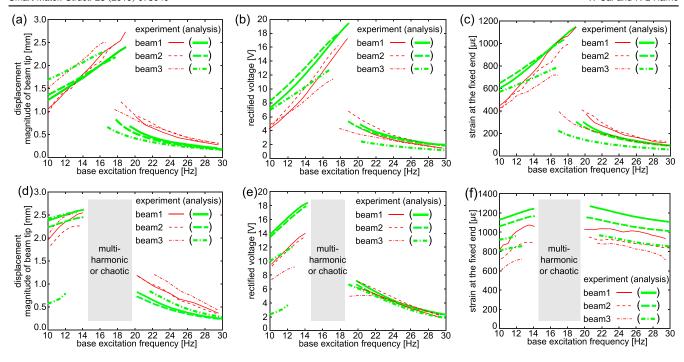
		$d_2$ (mm)							
	L (mm)	$b_L$ (mm)	$h_s$ (mm)	$h_c$ (mm)	Monostable case	Bistable case	$\varepsilon_{33} \; (\mathrm{nF}  \mathrm{m}^{-1})$	$e_{31} ({\rm C m}^{-2})$	$M_0$ (g)
Beam 1	37.4	32	0.034	0.02	10.8	10.4	30.15	-14.54	16.6
Beam 2	37.4	18	0.04	0.02	10.6	10.3	30.15	-14.54	16.6
Beam 3	45	32	0.034	0.02	11.8	11.6	30.15	-14.54	7.0

beam axis, as shown in figure 8(b). The shaker table is driven by a controller (Vibration Research Controller VR9500) and amplifier (Crown XLS 2500) with an accelerometer (PCB Piezotronics 333B40) to provide control signal feedback to ensure the base acceleration amplitude remains  $5\,\mathrm{m\,s^{-2}}$  at all frequencies examined. Two laser displacement sensors (Micro-Epsilon ILD-1420) measure the absolute displacement of the beam tip and the shaker table. A bridge rectifier (1N4148 diodes) with a smoothing capacitor  $C_r$  and resistive load R are connected to the piezoelectric beam to quantify the converted electric energy.

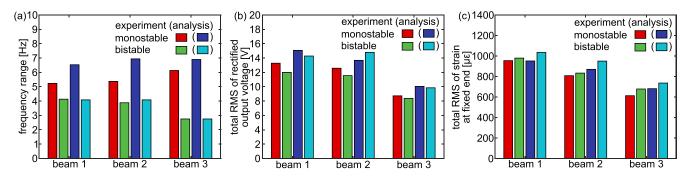
#### 6. Experiment validation

The experiments examine both monostable and bistable configurations of beams 1, 2, and 3. Figures 9(a) and (b) respectively show the displacement amplitude of the beam tip and rectified output voltage of the three beams for a

monostable configuration. Figures 9(d) and (e) are the corresponding responses for a bistable configuration of each beam. Analytical predictions are shown as the thick curves whereas the experimental results are presented using thin curves in the same respective line style. Because the strain at the clamped ends of the beams cannot be measured without influencing the laminated beam mechanical properties, the strain responses shown in figures 9(c) and (f) are calculated from finite element analysis (FEA) using ABAOUS. In the FEA, a solid model of the exact beam composition is used, where all parameters employed are provided in tables 2 and 4. Rigid elements are included to model the aluminum magnet holder, and a mass element is used to model the beam tip mass. After establishing the finite element model, the displacement amplitudes shown in figures 9(a) and (d) are applied as boundary conditions for the FEA to compute a static load analysis that provides values of the clamped end strain, using the strain at the middle of the clamp.



**Figure 9.** Frequency responses for three beams. (a) Displacement amplitude at the beam tip, (b) rectified output voltage, and (c) strain at the fixed end for monostable configurations. (d)–(f) Corresponding responses for bistable configuration.



**Figure 10.** Comparison of cost function values between experiment and analysis for three beam designs. (a) Frequency range. (b) Total RMS of the rectified output voltage. (c) Total RMS of the strain at the fixed end.

From the displacement responses shown in figures 9(a) and (d), the large displacement amplitudes at the bifurcations are in good agreement between the analysis and experimentation. Since the stiffness of the beam is increased due to the rigid crosssection assumption in Euler-Bernoulli beam theory, the analytical strain predictions in figures 9(c) and (f) are relatively greater than the FEA results using the experimental data of displacement amplitude. This in turn increases the predicted analytical output voltage in figures 9(b) and (e) compared to the experimental levels. Electrical component losses and higher order harmonics are also neglected in analysis, which may also contribute to the overprediction of rectified voltage. The grey shaded areas in figures 9(d)-(f) correspond to multi-harmonic or chaotic dynamic behaviors that are not predicted by analysis due to steady-state assumptions in the solution formulation. Despite these discrepancies, the overall quantitative agreement between the experimental and analytical results in figure 9 exemplifies that the model formulation accurately replicates the electromechanical behaviors of the piezoelectric laminated energy harvester.

The results of figure 10 establish the efficacy of the optimization created through this research. The results of monostable cases of all three beams suggest a higher output voltage with a lower strain level and a wider frequency range (red or blue bars) when compared to the bistable configurations of the energy harvesters (green or cyan bars). This emphasizes the global superiority of monostable platforms over bistable system configurations to achieve broadly robust energy harvesting performance and agrees with results obtained throughout section 4. In addition, the tapered beam 2 exhibits quantitatively less strain at the clamped end than the rectangular beam 1 of the same length, figure 10(c), while the beam 2 is within 1 V RMS of the rectified output voltage of beam 1, figure 10(b). This result agrees with the findings of section 4.2.2. Furthermore, when the tip mass decreases to 7 g for beam 3, the frequency range may still be broad by

introducing sufficient nonlinearity as shown in figure 10(a). Yet, the output voltage and strain for beam 3 in figures 10(b) and (c) are decreased comparing with the results of the other two beams, which supports the influences uncovered in section 4.2.3. Thus, considering the design constraints examined here, piezoelectric laminated cantilevers with tapered shapes, monostable nonlinearity, and moderate lengths deliver the most performance robust DC power delivery.

#### 7. Conclusions

This research provides the first illumination of the coupled influences of nonlinearity and beam shape design on the resulting mechanical robustness and versatile electrical power generation of a nonlinear energy harvesting system. Here, an analytical model is established and verified through numerical simulations and experimental examinations for key influences of nonlinearity, beam shape, and tip mass select result in robust system designs. Multi-objective optimization guides attention to strategic combinations of design parameters based on the specific optimization variables and relative weights given to the objective functions. Trapezoidal beam shapes are found to promote large output voltage without yielding high strain at the clamped end, which promotes practical longevity of the system. Bistable implementations of the nonlinear energy harvesters are found to be undesirable for sake of large residual strain at the clamped end, even though they may deliver wide frequency range of operation and high output voltages. Therefore, both the optimization results using the new analysis and experiment validations indicate that a trapezoidal beam in a monostable configuration is globally preferred.

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