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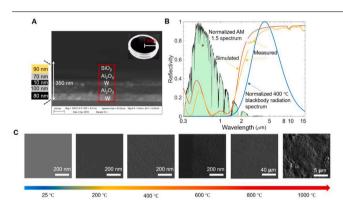


# Scalable-manufactured metal-insulator-metal based selective solar absorbers with excellent high-temperature insensitivity

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#### GRAPHICAL ABSTRACT



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#### ABSTRACT

Solar absorbers, harvesting solar irradiance in the form of heat, are extensively applied in the solar hot water systems and concentrated solar thermal systems such as concentrated solar power plants, solar thermoelectric generators, and solar thermophotovoltaics. It is of great significance to incorporate spectrally selective solar absorbers into solar thermal systems, especially at high operational temperatures to depress the thermal loss due to the thermal re-emission of high-temperature solar absorbers. This work computationally and experimentally demonstrates a new spectrally selective solar absorber consisting of a multilayered stack made of silica/alumina/tungsten/alumina/tungsten based on metal-insulator-metal resonance structures and fabricated by the magnetron sputtering method, which are angular insensitive and polarization-independent. The relationship between solar conversion efficiency, cut-off wavelength, operational temperatures, and concentration factor is theoretically investigated. An overall absorptance of 88.1% at solar irradiance wavelength, a low emittance of 7.0% at infrared thermal wavelength, and a high solar-to-heat efficiency of 82.5% are identified. Additionally, it shows the annealed samples maintain an extremely high absorption in solar radiation regime over at least 800 °C and a high concentration factor of over 100. The SEM topography images of the absorbers after thermal annealing at various temperatures demonstrates that the surface blisters and cracks result in the thermal degradation of the absorbers due to the dissimilarity between thermal expansion coefficients of tungsten and silica. The high-temperature insensitivity of the multilayer metal-insulator-metalbased selective solar absorbers will shed light on an alternative novel photonic metamaterial structure that can be scalable-manufactured to improve the energy conversion efficiency of solar thermal engineering.

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#### 1. Introduction

The exploration of alternatives to traditional fossil fuels is invigorated over the past decades because of the energy crisis and increasing global warming, among which solar thermal energy is of intense interest due to its affluence and environmental sustainability. Solar thermal energy is a technology for harnessing solar energy as heat to be applied in the residential, commercial, and industrial sectors. It gains much more attention recently in industrial heating [1], air condition [2], water desalination [3], and electricity generation [4,5]. However, it is strongly impeded to larger-scale engineering applications due to the constrained solar-to-heat conversion efficiency coming from thermal reemission of the blackbody radiation when solar absorbers are heated to high temperatures. Selective solar absorber, a key component that harvests solar radiations and converts it into thermal energy, greatly affects the thermal performance and efficiency of concentrated solar power (CSP) [6], solar thermoelectric generator (STEG) [4], and solar thermophotovoltaic (STPV) systems [7,8]. Ideally, solar absorber should have highly sharp spectral selectivity, which means it possesses a unity absorptance and omnidirectional, polarization-independent nature [9,10] in the solar irradiance range (visible and near-infrared regions) and shows no spontaneous thermal emittance losses in the midinfrared region due to blackbody radiation. The cut-off wavelength of a solar absorber, at which its absorptance spectrum changes sharply, is temperature-dependent owing to the wavelength shifting of blackbody radiation according to Wien's displacement law, so it is meaningful to investigate how the cut-off wavelength, the operational temperature, and solar concentration factor affect the energy conversion of CSP systems. Furthermore, excellent high-temperature thermal stability is also highly desired for assuring solar absorbers to operate with high solar-toheat conversion efficiency at fluctuated high temperatures. Therefore, spectrally selective solar absorbers that are thermally stable, angularand polarization-insensitive will extensively promote the large-scale applications of solar thermal engineering.

Numerous approaches have been discovered to obtain selective broadband absorbers, including both natural existing materials and micro/nanoscale patterned metamaterials. Natural materials based solar absorber, such as black carbon paint, black chrome [11,12], and Pyromark [13] intrinsically exhibit high absorptance in the visible and near-infrared regions, as well as cermet [14,15]. However, they are not ideal for high-temperature applications, because they show high thermal leakage due to high emittance in the mid-infrared wavelength region. Moreover, the tunability of their spectral selectivity is relatively low, limiting their feasibility for diverse applications at different operational temperature or solar concentration factor. Besides natural materials, metamaterial with micro/nanoscale artificial structures displays spectral selectivity that cannot be achieved in naturally occurring materials [16,17]. The cut-off wavelength of high visible/nearinfrared absorptance and low mid-infrared emittance can also be tuned through adjusting the geometric parameters of metamaterials. Plentiful selective metamaterials absorber, such as one- [18]/two-dimensional surface gratings [19-21], nanoparticles embedded dielectrics [22,23], cross-bar [24] or nano-disk arrays [25], and photonic crystals [26-30] have been developed. However, these selective absorbers rely on complicated nanofabrication process, such as electron beam lithography (EBL) or focused ion beam milling (FIB), which is high-cost and time-consuming and makes it hard for large-scale industrial fabrication. The EBL and FIB techniques both rely on expensive equipment and materials supplies which make them stay on lab-scale fabrication and difficult to meet the industrial manufacturing requirements. Furthermore, the samples rely on the EBL and FIB techniques that are still not on the wafer-scale, which makes it time-consuming to expand to the industrial level. The fabrication process of photolithography is complicated and needs to be conducted in a cleanroom atmosphere. Compared with the abovementioned technique, vacuum deposition methods, either physical vapor deposition (PVD) or chemical vapor

deposition (CVD), can fabricate large-scale samples and the cleanroom atmosphere is not mandatory in the industrial fabrication. The thermal stress of these structured based absorbers at elevated high temperatures will cause the unrecoverable damage of the surface topography and yields the loss of spectral selectivity. Hence, selective solar absorber that is lithography-free and holds high tunability of spectral selectivity and high-temperature stability are highly required in solar thermal energy engineering.

Recently, the multilayered thin-film selective solar absorbers have gained increasing attention owing to their strong sunlight absorption with simple configurations and facile fabrication procedures. Li et al. [31] manufactured an all-ceramic titanium nitride (TiN) /TiNO/ zirconium dioxide (ZrO<sub>2</sub>)/silica (SiO<sub>2</sub>) thin film with 92.2% solar absorptance and 17% thermal emittance at 1000 K using a magnetron sputtering system. Khoza et al. [32] fabricated a new ZrO<sub>x</sub>/zirconium (Zr)/ZrOx/AlxOv multilayered selective solar absorber using electronbeam vacuum evaporation and achieved a solar absorptance of 93% and low thermal emittance of 11% at 100 °C. Gao et al. [33] deposited a solar selective coating of zirconium diboride (ZrB<sub>2</sub>)/alumina (Al<sub>2</sub>O<sub>3</sub>) by a magnetron sputtering system on stainless steel (SS) substrate and the coating exhibited a solar absorptance of 92% and an emittance of 11% that are stable at 500 °C for 100 h. Gao et al. [34] fabricated a selective coating of SS/Titanium carbide (TiC)-ZrB<sub>2</sub>/ZrB<sub>2</sub>/Al<sub>2</sub>O<sub>3</sub>, which has a solar absorptance of 92% and thermal emittance of 10% and exhibits thermal stability at 600 °C for 100 h in vacuum. Metal-insulator-metal (MIM) based novel absorbers consisting of two metallic layers with a thin dielectric layer sandwiched in between. The incident light is absorbed due to strong optical interactions at the resonant wavelengths and is converted to heat in lossy metallic nanostructures [35]. MIM based absorber is among the most promising composite for spectrally selective solar absorber with excellent selectivity and high-temperature thermal stability. It has been demonstrated as an alternative approach to achieve wavelength selectivity [36-38] and can be fabricated by simple, cost/time effective, and large-scale vacuum deposition methods, such as sputtering, evaporation or chemical vapor deposition [39]. The scalable-manufactured potential of the MIM metamaterials makes it more attractive than the one-/two-dimensional photonic crystals and patterned metasurfaces that rely on complicated photolithography methods. It also makes it not suitable for large-scale solar thermal engineering applications. Chirumamilla et al. [36] reported a MIM based solar absorber with Al<sub>2</sub>O<sub>2</sub> and tungsten (W) thin-film structures using electron beam evaporation and direct-current (DC) sputtering methods and examined the thermal stability of fabricated absorbers at 600 °C in air for 4 h. Langlais et al. [37] fabricated the five layers stacked with magnesium oxide, titanium dioxide, and W thin-film solar absorbers with 85% solar conversion efficiency at 1000 suns. Nuru et al. [40] manufactured an Al<sub>x</sub>O<sub>y</sub>/Pt/Al<sub>x</sub>O<sub>y</sub> multilayered solar absorber with an electron beam evaporation method and experimentally validate its thermal stability in the air up to 500 °C for 2 h. Though the high solar absorptance can be achieved in the partial or full solar wavelengths, the maximum thermal annealing temperature and time are limited to be 600 °C or 4 h in these reported studies.

The high-temperature stability of MIM resonance absorber coated with an anti-reflection layer because of possible thermal stress and material oxidation requires further investigations as well as its thermal performance at different operational temperatures and solar concentration factor. Refractory materials, such as W, Al $_2$ O $_3$ , and SiO $_2$  (with bulk materials melting points of 3422 °C, 2072 °C, and 1710 °C, respectively) [36], based multilayer broadband absorber can be a promising candidate with high-temperature stability. In addition, W, Al $_2$ O $_3$ , and SiO $_2$  have low thermal expansion coefficients of 4.2  $\times$  10 $^{-6}$  m/(m K), 5.4  $\times$  10 $^{-6}$  m/(m K), and 0.55  $\times$  10 $^{-6}$  m/(m K) [36] at room temperature, respectively and these values are at the same order, which helps maintain the surface topography after thermal annealing though the thermal expansion coefficient is temperature-dependent. Additionally, W is a good solar radiation absorber in the visible region since

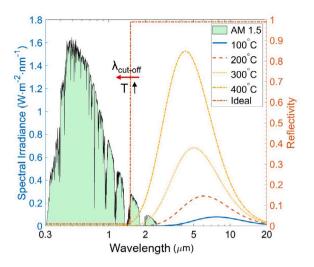


Fig. 1. Solar spectral irradiance (AM 1.5, global tilt), radiative heat flux of blackbody thermal radiation, and reflectivity spectrum of ideal selective solar absorber.

the real part of the dielectric permittivity is positive below 900 nm, among which over 50% of solar radiation power distributes, and results straightly to high visible light absorption. Simultaneously, W shares the similar properties of common metals, such as silver, aluminum, and gold, that are highly reflective in the mid-infrared wavelength region. The Al<sub>2</sub>O<sub>2</sub> and W thin layers arrange alternatively forming a MIM resonator and thereby exhibit enhanced absorptance of visible and near-infrared light. The thin SiO2 layer on top of the MIM resonator serves as an anti-reflection layer to enhance the visible light absorption. Both Al<sub>2</sub>O<sub>3</sub> and SiO<sub>2</sub> layers sandwich the easy-oxidized W layer and serve as a protective layer to ensure the high-temperature stability.

In this work, we theoretically design and experimentally fabricated an ultrathin selective solar absorber based MIM resonance and SiO2 anti-reflection effects. The angular and polarization-independent spectral reflectivity is theoretically identified. The reflection of both transverse electric (TE) and transverse magnetic (TM) polarizations remains low at incident angles of up to 85° within the visible and near-infrared region. High-temperature thermal treatment is further experimentally investigated to prove its thermal insensitivity. Scanning electron microscope (SEM) is employed to explore the topography variations after thermal annealing at different temperatures to shed light on how the topography of samples affects their spectral selectivity. The simulations of thermal performance for the fabricated absorber under a one-day sunlight cycle indicate that its highest stagnation temperatures of 273 °C and 675 °C under 1 sun and 20 suns, respectively. This elucidates the potential engineering applications of the mid-temperature industrial applications such as processing heat supply and high-temperature concentrated solar power plants, STPV, and STEG systems. Compared with the recent work about the MIM based multilayered selective solar absorber listed above, we proposed a novel scalable-manufactured multilayered stack consisting of SiO<sub>2</sub>/Al<sub>2</sub>O<sub>3</sub>/W/Al<sub>2</sub>O<sub>3</sub>/W with excellent spectral selectivity and high-temperature thermal stability up to 800 °C.

#### 2. Fundamental theory

#### 2.1. Energy conversion efficiency analysis of ideal solar absorber

To approach the perfect design of selective solar absorber, the affecting parameters of solar-to-heat conversion efficiency for the ideal selective solar absorber are theoretically investigated. Fig. 1 shows the spectral irradiance distributions of solar radiative heat flux, as well as the thermal radiative heat flux of a blackbody at various temperatures.

It can be found that most of the solar irradiance is distributed in the visible and near-infrared region (0.3  $\mu$ m  $\sim$  2.5  $\mu$ m), while most of the thermal radiation from a blackbody spreads within the mid-infrared region according to Wien's displacement law. When a blackbody is at 400 °C, the thermal emission is at the same order of solar heat flux. Therefore, it is crucial to maximizing the absorption of solar radiation and minimizing the energy loss from the thermal re-emission in the mid-infrared spectral region. Additionally, the distribution wavelength region of blackbody radiation moves to a shorter wavelength as its temperature increases, as illustrated by thermal radiation curves of the blackbody at 100 °C, 200 °C, 300 °C, and 400 °C in Fig. 1. The optimal cut-off wavelength,  $\lambda_{\text{cut-off}}$  shifts to a shorter wavelength to reduce the blackbody thermal re-emission to approach a maximum efficiency. Fig. 1 also exhibits the reflectivity spectrum of an ideal solar absorber, which shows zero reflectivity below  $\lambda_{\text{cut-off}}$  to increase the absorption of solar radiation and unity reflectivity beyond the  $\lambda_{\rm cut-off}$  to reduce the thermal emission. Moreover, the principle differs between unconcentrated and concentrated solar power applications, considering that the concentration factor of a focusing lens can be designed up to several thousands of times for now. Note that, the operational temperature of the solar absorber varies greatly when the circulation rate of working fluids or thermal loads differs. The energy conversion efficiency of solar absorber alters as a function of the cut-off wavelength, the concentration factor of solar light, and the operational temperature of the solar absorber. Hence, it is a trade-off to design a nearly perfect selective solar absorber for diverse engineering

The solar-to-heat conversion efficiency of solar absorber is defined by this following equation, assuming no heating conduction and con-

$$\eta = \frac{\alpha_{\rm abs} CF \cdot Q - \epsilon_{\rm abs} \left( \sigma T_{\rm abs}^4 - \sigma T_{\rm amb}^4 \right)}{CF \cdot Q} \tag{1}$$

where CF is the concentration factor, i.e., the ratio of the solar collector aperture area and the solar absorber area, O is the solar radiative heat flux at AM 1.5 (global tilt) [41].  $\sigma$  is the Stefan–Boltzmann constant.  $T_{
m abs}$  is the operational temperature of absorber, and  $T_{
m amb}$  is set to be 25 °C as the ambient temperature.  $\alpha_{\rm abs}$  and  $\epsilon_{\rm abs}$  is the total absorptance and emittance of the solar absorber, respectively, which are expressed as the following:

$$\alpha_{\rm abs} = \int_{0}^{\infty} \alpha'_{\lambda, \rm abs} I_{\rm AM1.5}(\lambda) d\lambda / \int_{0}^{\infty} I_{\rm AM1.5}(\lambda) d\lambda \tag{2}$$

$$\epsilon_{\rm abs} = \int_{0}^{\infty} \epsilon'_{\lambda, \rm abs} I_{\rm BB} \left( \lambda, T_{\rm abs} \right) d\lambda / \int_{0}^{\infty} I_{\rm BB} \left( \lambda, T_{\rm abs} \right) d\lambda \tag{3}$$

$$\epsilon_{\rm abs} = \int_0^\infty \epsilon_{\lambda, \rm abs}' I_{\rm BB} \left( \lambda, T_{\rm abs} \right) d\lambda / \int_0^\infty I_{\rm BB} \left( \lambda, T_{\rm abs} \right) d\lambda \tag{3}$$

where  $\alpha'_{\lambda, \mathrm{abs}}$  and  $\epsilon'_{\lambda, \mathrm{abs}}$  are the wavelength-dependent absorptance and emittance for the solar absorber at the room temperature, respectively.  $I_{\text{AM1.5}}(\lambda)$  is the spectral irradiance intensity of solar radiation at AM 1.5 and  $I_{BB}(\lambda, T_{abs})$  is the spectral blackbody radiative intensity at a certain operational temperature,  $T_{\rm abs}$ . For the theoretical calculation of  $\alpha'_{1 \text{ abs}}$  the spectral integration range is limited within 0.25 µm ~ 4.0 µm, where the available AM 1.5 data covers, while the integration range lies between 2.5  $\mu m$  and 20  $\mu m$  for the calculation of  $\epsilon'_{\lambda,abs}$ .

Fig. 2 illustrates the energy conversion efficiency,  $\eta_{abs}$ , as a function of the absorber operational temperature,  $T_{\rm abs}$ , and the cut-off wavelength,  $\lambda_{\rm abs,cut-off}$  at different concentration factor, CF, 10 suns, 50 suns, and 100 suns, in which the red color represents higher efficiency, and the blue means lower efficiency. Note that, these energy efficiency contour plots are obtained by analyzing Eqs. (1), (2), and (3). First of all, the solar concentration factor is set to be 10, 50, and 100, and then treat the operational temperature  $(T_{abs})$  and the optimal cut-off as two variables to calculate the solar-to-heat conversion efficiency of the solar absorbers. It can be observed in Fig. 2A, B, and C that the efficiency curve shares the similar potential of increasing to maximum from zero and then decreasing as  $\lambda_{\text{cut-off}}$  sweeps from 0.25 µm to 4.0 µm. For the absorber with a temperature of 600 °C under 50 suns, the efficiency is

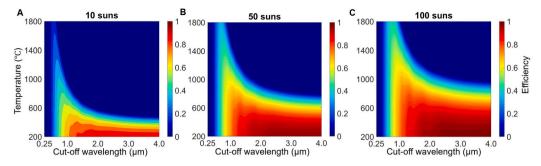


Fig. 2. (A), (B), and (C) Simulated solar-to-heat energy conversion efficiency of solar absorber contour plotted against the absorber operational temperature and cut-off wavelength at different solar concentration factor of 10, 50, and 100, respectively.

18.9% at  $\lambda_{cut\text{-}off}$  of 0.5  $\mu m,$  and increases to a maximum of 94.3% at  $\lambda_{cut\text{-off}}$  of 1.8 µm, then decreases to 72.8% at  $\lambda_{cut\text{-off}}$  of 4 µm. According to Wien's displacement law, the thermal radiative heat flux of 600 °C blackbody is distributed within 1 µm to 16 µm and reaches its maximum at 3.2 µm. Therefore, the solar radiation matters at a shorter wavelength  $(0.25 \mu m \sim 1.8 \mu m)$ , while the thermal losses due to blackbody reemission become prominent at a longer wavelength (1.8  $\mu$ m  $\sim$  4.0  $\mu$ m). The maximum efficiency decreases and the corresponding  $\lambda_{\text{cut-off}}$  shifts to a shorter wavelength at 50 suns as the temperature increases. The energy conversion efficiency at other concentration factor follows the same rules, as can be seen in Fig. 2. The maximum efficiency under 50 suns happens at 1.8  $\mu m$  with 94.3% at 600 °C, at 1.3  $\mu m$  with 79.6% at 1000 °C, and at 0.8  $\mu m$  with 47.9% at 1500 °C. It is reasonable that  $\lambda_{\text{cut-off}}$  shifts to the left, since the main distribution wavelength of blackbody radiation moves to a shorter wavelength as the temperature increases. The corresponding maximum efficiency decrease from 94.3% to 47.9% as the temperature increases from 600 °C to 1500 °C, since the overlap between the distribution area of the solar irradiance and blackbody radiation becomes larger, considering that the wavelength region of solar radiation keeps unchanged, while the thermal radiation regime of the blackbody shifts to the left and the magnitude of thermal radiative heat flux increases along with the increasing of absorber temperature. The maximum efficiency of the absorber at 800 °C happens at 1.3  $\mu m$  with 81.3% under 10 suns, at 1.4  $\mu m$  with 86.9% at 50 suns, and at 1.8  $\mu m$  with 91.1% at 100 suns. It can be found that the  $\lambda_{cut-off}$ moves to a longer wavelength and the maximum efficiency increases as well when the concentration factor increase from 10 suns to 100 suns. It is easy to understand that the absorbed heat flux of solar power increases while the distribution wavelength region and the radiation rate of the blackbody keep unchanged at a certain temperature, and the solar radiation becomes dominant at a high CF.

#### 2.2. Thermal performance of the selective solar absorbers

To demonstrate the absorption capability of the multilayer solar absorber under direct solar radiation with various CF, the temperature variations are simulated by solving the thermal balance equation as expressed by the following [42,43]:

$$\begin{aligned} Q_{\text{total}}(T_{\text{abs}}, T_{\text{amb}}) = & Q_{\text{sun}}(T_{\text{abs}}) + Q_{\text{amb}}(T_{\text{amb}}) \\ & - Q_{\text{re-emit}}(T_{\text{abs}}) \end{aligned} \tag{4}$$

It is supposed that the backside of the solar absorber is thermally insulated (i.e., no thermal load is connected to the absorber), so the heat transfer is considered only between the solar absorber and air. Here,  $Q_{\rm sun}$  is the heating power of solar absorber from solar radiation,  $Q_{\rm amb}$  is incident thermal radiation from ambient,  $Q_{\rm re-emit}$  stands for the heat flux through thermal re-emission from the solar absorber surface, and  $Q_{\rm total}$  is the net heating power of the solar absorber.

Solar radiation absorbed by the absorber,  $Q_{sun}$ , is given by  $Q_{sun}$  ( $T_{abs}$ ):

$$Q_{\text{sun}}(T_{\text{abs}}) = A \cdot CF \int_{0}^{\infty} d\lambda I_{\text{AM1.5}}(\lambda) \alpha(\lambda, \theta_{\text{sun}}, T_{\text{abs}})$$
 (5)

Here, A is the area of the solar absorber.  $\alpha(\lambda, \theta_{\text{sun}}, T_{\text{abs}})$  is the temperature, wavelength-dependent, and angular sensitive absorptance of the solar absorber, however, as discussed above, the absorptance of the designed solar absorber is angular and temperature-independent below 600 °C. Hence, it is rational to take the measured data at room temperature into calculations.

The absorbed power of incident thermal radiation from atmosphere  $Q_{
m amb}$ 

 $(T_{\rm amb})$  can be expressed as follows:

$$Q_{\rm amb}(T_{\rm amb}) = A \int_0^\infty d\lambda I_{\rm BB}(T_{\rm amb}, \lambda) \alpha(\lambda, \theta, \phi, T_{\rm abs}) \times \epsilon(\lambda, \theta, \phi)$$
(6)

where  $I_{\rm BB}(T_{\rm amb},\lambda)=2hc^5\lambda^{-5}\exp(hc/\lambda k_BT-1)^{-1}$  defines the spectral radiance of a blackbody at a certain temperature T, where h is the Planck's constant,  $k_B$  is the Boltzmann constant, and  $\lambda$  is the wavelength.  $\alpha(\lambda,\theta,\phi,T_{\rm abs})=\frac{1}{\pi}\int_0^{2\pi}{\rm d}\phi\int_0^{\pi/2}\epsilon_\lambda\cos\theta\sin\theta{\rm d}\theta$  is the temperature-dependent absorptance of solar absorber [44]. Here, we take it as temperature-independent after demonstrations of high-temperature stability test. The emissivity of the air,  $\epsilon(\lambda,\theta,\phi)$ , is given by 1- $t(\lambda,\theta,\phi)$ .  $t(\lambda,\theta,\phi)$  is the transmittance value of atmosphere obtained from MOD-TRAN 4 [45].

The heat flux through thermal re-emission from the solar absorber surface is determined as follows:

$$Q_{\text{re-emit}}(T_{\text{abs}}) = A \int_{0}^{\infty} d\lambda I_{\text{BB}}(T_{\text{abs}}, \lambda) \epsilon(\lambda, \theta, \phi, T_{\text{abs}})$$
 (7)

where,  $I_{\rm BB}(T_{\rm abs},\lambda)$  is the thermal radiation of a blackbody at a certain temperature.  $\epsilon(\lambda,\theta,\phi,T_{\rm abs})=\alpha(\lambda,\theta,\phi,T_{\rm abs})$  is the emissivity of the solar absorber according to Kirchhoff's law of thermal radiation [46].

The time-dependent temperature variations of the absorber can be obtained by solving the following equation:

$$C_{\text{abs}} \frac{dT}{dt} = Q_{\text{total}}(T_{\text{abs}}, T_{\text{amb}}) \tag{8}$$

Since the multilayer structure of the absorber is only 350 nm thick, it is reasonable to neglect its thermal resistance. Therefore, the heat capacitance of the absorber,  $C_{\rm abs}$ , considers here equal to the heat capacitance of 300  $\mu \rm m$  silicon wafer on top of which the multilayer absorbers are fabricated. The thermal performance analysis of the selective solar absorbers will be discussed in Section 5.

#### 3. Methods

#### 3.1. Sample fabrications and SEM topography characterizations

The selective solar absorber samples are deposited on a 2'' silicon wafer using a home-built high vacuum magnetron sputtering system, details about this machine are described in this publication [8]. The base pressure before the deposition is  $3.4\times10^{-7}$  Torr. The fabricated parameters including deposition rate, base pressure, and sputtering power are specified in Table 1. The SiO $_2$  subscript layer is grown from

**Table 1**Deposition methods and parameters of different layers of the designed absorber.

Material	Sputtering	Layer	Deposition	Argon base	Sputtering
	deposition	thickness	rate	pressure	power
	method	(nm)	(Å/s)	(Pa)	(W)
SiO <sub>2</sub> coating	RF	90	0.98	0.38	150
Al <sub>2</sub> O <sub>3</sub> top layer	RF	70	1.09	0.38	15
W top layer	DC	10	0.72	0.08	150
Al <sub>2</sub> O <sub>3</sub> top layer	RF	100	1.09	0.38	15
W substrate	DC	80	0.72	0.08	150

the silicon target using reactive radio frequency (RF) sputtering with argon and oxygen gas supplied during deposition. The total thickness of the fabricated absorber is 350 nm, which is shown in Fig. 3B. The inset of Fig. 3B shows a photo of the multilayer absorber sample is placed in a 2-inch single wafer carrier case. The black surface elucidates its high absorptance in the visible spectral region. The 80 nm W substrate layer is thick enough to block all the incident wavelengths of interest from 0.3  $\mu m$  to 15  $\mu m$ , so it can be treated as optically opaque. The cross-section topography of the absorber sample is characterized by SIGMA VP Field Emission-Scanning Electron Microscope (FE-SEM).

#### 3.2. Optical and radiative properties measurements

Reflectance measurements in the ultraviolet, visible, and near-infrared regions are performed on an Agilent Cary 5000 spectrometer with a 150 mm PTFE based integrating sphere. Reflectance measurements are taken with a wavelength scan step of 1 nm at a normal incident angle and normalized to a labsphere spectralon reflectance standard. The near/mid-infrared reflectance measurements are completed on Jasco 6600 FTIR spectrometer together with a Pike 3 inches golden integrating sphere. The angle of the incident beam from the FTIR spectrometer is fixed at 12°. Spectra are taken at a scan rate of 64 with a wavelength resolution of 0.4 cm<sup>-1</sup>. Details about FTIR spectrum measurements can be referred to Tian et al.'s recent article [8].

#### 3.3. High-temperature insensitivity test

High-temperature stability tests are done in a tube oven at an argon protective atmosphere with an alumina tube of 5 cm diameter and 80 cm length. The tube is connected to a rotary vane vacuum pump, a vacuum gauge, and an argon tank at one end. Samples are placed in an alumina crucible boat (100 mm  $\times$  30 mm  $\times$  20 mm) and positioned in the center of the tube. First, the rotary van vacuum pump will vacuum down to  $1.5\times10^{-2}$  Torr, then open the argon tank valve with a partial pressure of 70 psi and introduce argon into the alumina tube for 30 s. These processes are repeated three times for each test. The temperature controller is set to be 200 °C, 400 °C, 600 °C, 800 °C, and 1000 °C for a 6-h cycle. The reflectance measurements are carried out after each thermal treatment.

#### 4. Results

#### 4.1. Design and spectrometric characterizations of absorber samples

The thickness for different layer stacks of the solar absorber is optimized using MATLAB optimizations toolbox. The materials of different layers are chosen as  $\mathrm{SiO}_2$ ,  $\mathrm{Al}_2\mathrm{O}_3$ , W,  $\mathrm{Al}_2\mathrm{O}_3$ , and W from the top to bottom. Their optimized thicknesses of this five-layer stack are obtained by tuning their respective thickness variables to reduce the discrepancy between the calculated spectra and the spectra of a desired ideal selective solar absorber. This process involves an optimization procedure that aims to minimize the error between the calculated and desired spectra. The error between these two spectra is given by  $\delta = \sum_{i=1}^n \left[ r_c - r_i \right]^2.$  Here,  $r_c$  and  $r_i$  are the calculated and desired reflectivity spectra of the solar absorbers. Index i refers to the ith

**Table 2**Calculated energy conversion efficiency of the designed solar absorber at different operational temperature under 100 suns using the simulated reflectivity spectra.

Temperature ( $T_{abs}$ , °C )	200	400	500	600
Efficiency (η, %)	81.0	82.3	81.5	79.8

wavelength (0.3 µm  $\sim$  20 µm) over which computational analyses are conducted. The proposed planer structure is fabricated by employing the magnetic sputtering technique. Fig. 3A illustrates the 3-D schematic of proposed multilayer stack consisting of five layers (i.e., SiO<sub>2</sub>, Al<sub>2</sub>O<sub>3</sub>, W, Al<sub>2</sub>O<sub>3</sub> and W from top to bottom, successively). The thickness of each layer for the multilayer stack is shown in the 2-D schematic of Fig. 3B. The thickness of different layers for the fabricated sample is confirmed by the cross-section view of FE-SEM shown in Fig. 3B. Considering the thickness of the bottom W layer, it is reasonable to consider the sample is opaque at the wavelength of interest and the solar absorptance of the absorber,  $\alpha_{\rm abs} = 1 - \rho_{\rm abs}$ . Table 2 shows the calculated energy conversion efficiency of the designed solar absorber under 100 suns calculated using the simulated reflectivity spectrum at different temperatures. The solar-to-heat efficiency of the absorber reaches a maximum at the temperature of 400 °C.

### 4.2. Diffuse and polarization-independent behaviors of designed solar absorbers

Generally, sunlight concentrates into beams with high energy density and it is absorbed by solar absorbers, while the incident angle of the concentrated beams is not always normal to the surface of solar absorbers. Therefore, the optical and thermal radiative properties at oblique angles of the solar absorber are vital for efficiently harvesting direct sunlight coming from various directions after an optical concentrator [47]. Besides, polarization independence is also crucial for a perfect absorber to maximize solar energy absorption since solar radiation is randomly polarized.

Here, Fig. 4 manifests the simulated contour plot of the spectral reflectivity for the multilayer solar absorber as a function of incident angle,  $\theta$ , and wavelength at TE, TM polarized waves, and unpolarized waves, in which the red color means higher reflectivity, while the blue color represents lower reflectivity. It is exhibited that the reflectivity remains lower than 0.1 with visible and near-infrared (from 0.25  $\mu m$ to 1.7  $\mu$ m) in Fig. 4A, B. At a wavelength,  $\lambda = 0.55 \mu$ m in which the irradiation peak of solar lies, the reflectivity is 0.129 at normal incidence, and it decreases slightly to 0.117 at  $\theta = 30^{\circ}$ , then reduces to 0.109 at  $\theta = 60^{\circ}$  for TE waves. Even at  $\theta = 75^{\circ}$ , the reflectivity is as low as 0.179 for TE waves. For TM waves, the reflectivity is 0.129 at normal incidence, and it decreases slightly to 0.093 at  $\theta =$  $30^{\circ}$ , then reduces to 0.044 at  $\theta = 60^{\circ}$ . The reflectivity maintains at 0.113 even at  $\theta = 75^{\circ}$  for TM waves at  $\lambda = 0.55$  µm. The reflectivity across the infrared region (from  $0.8~\mu m$  to  $1.7~\mu m$ ) follows similar rules. It proves the multilayer solar absorber remains low reflectivity over a broad spectral range for both polarizations. The angular-insensitivity spectral reflectivity for unpolarized waves is shown in Fig. 4C. It can be seen that the reflectivity is insensitive to the incident angle over a large range, and low reflectivity exists over the visible and nearinfrared region. Note that, although the reflectivity shows a dip at  $\lambda$  =  $8 \mu m$ , the peak wavelength of 400 °C blackbody thermal radiation lies at 3.2 µm. For higher temperature applications, the peak wavelength shows blue shifts to a shorter wavelength, so it is reasonable that this dip does not affect the thermal performance of the designed solar absorber. The overall absorptance of the fabricated samples at the solar irradiance wavelength region is 87.19%, while it shows a low emittance of 7.13% at a thermal wavelength with the measured spectral reflectivity, assuming that the operational temperature of absorbers is 100 °C.

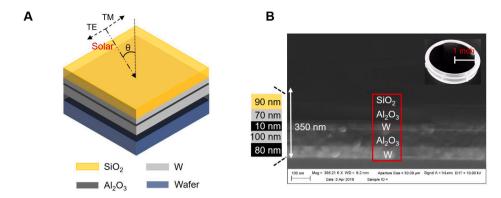


Fig. 3. (A) 3-D schematic of a multilayer stack consisting of W,  $Al_2O_3$ , and  $SiO_2$ . The incidence angle,  $\theta$ , is defined as the angle between solar incident radiation and the surface normal. (B) A cross-section SEM micrograph of the fabricated sample, the 2-D schematic shows the thickness of each layer for the multilayer stack, and the insect is an image of a sample fabricated on a 2-inch silicon wafer staying in a wafer carrier case.

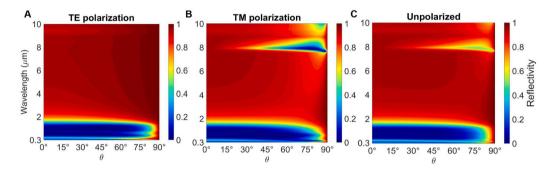


Fig. 4. (A), (B), and (C) Simulated angle dependent TE polarization, TM polarization, and unpolarized reflectivity of selective solar absorber contour plotted against wavelength and angle of incidence,  $\theta$ .

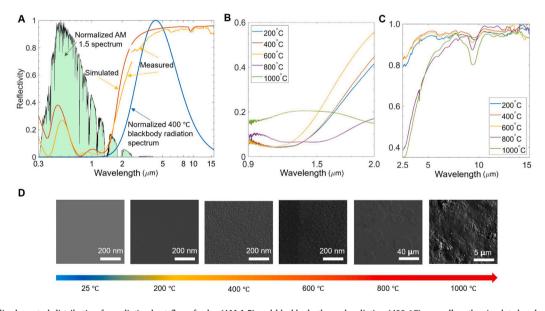


Fig. 5. (A) Normalized spectral distribution for radiative heat flux of solar (AM 1.5) and blackbody thermal radiation (400 °C), as well as the simulated and measured reflectivity spectra of a multilayer solar absorber; (B) Near-infrared and (C) mid-infrared spectral reflectivity of the multilayer absorber measured by FTIR spectrometer at room temperature (25 °C) after thermal treatment at 200 °C, 400 °C, 600 °C, 800 °C, and 1000 °C for 6 h. (D) SEM topographic images of the fabricated multilayer solar absorber sample after thermal annealing at 200 °C, 400 °C, 600 °C, 800 °C, and 1000 °C for 6 h.

#### ${\it 4.3. \ Spectral \ selectivity \ of \ the \ fabricated \ sample \ at \ room \ temperature}$

The hemispherical reflectivity of the fabricated sample is characterized by UV/VIS/NIR spectrophotometer in the visible and near-infrared region (from 0.3  $\mu m$  to 2.5  $\mu m$ ) and FTIR spectrometer in the midinfrared region (from 2.5  $\mu m$  to 16  $\mu m$ ) at room temperature. The measured spectral reflectivity breaks at 2.5  $\mu m$  since the incident angle

of the light beam for these two spectrometers is different and the measurement mechanism of the embedded detector differs, as discussed in the following method section. The simulated spectral reflectivity is also interrupted at 2.5  $\mu m$  to match the measured spectrum. The simulation of reflectivity spectrum is executed at  $\theta=0^\circ$  and  $12^\circ$  at visible/near-infrared and mid-infrared region, respectively. A good match between the simulated and measured spectra is seen in the near-infrared and

Table 3 Overall thermal absorptance and thermal emittance as fabricated and after thermal annealing for 6 h at different temperatures ( $T_{abs}$ = 100 °C).

Temperature (°C)	$lpha_{ m abs}$	$\epsilon_{ m abs}$
25	0.872	0.071
200	0.843	0.082
400	0.838	0.063
600	0.809	0.160
800	0.892	0.190
1000	0.812	0.211

mid-infrared spectral region, while the measured reflectance is lower from 0.3  $\mu m$  to 0.5  $\mu m$ , which is desired for enhanced solar radiation absorption.

### 4.4. Thermal insensitivity test and thermal failure mechanism from SEM characterizations

Consistent spectral performance of the selective solar absorbers at high temperatures is significant, especially for CSP systems to maintain a high conversion efficiency under concentrated solar radiation. To evaluate the radiative properties of the absorber at different temperatures, we use the FTIR spectrometer, which covers  $0.9~\mu m$  to  $15~\mu m$ together with a gold integrating sphere, to measure the hemispheric reflectivity after thermal annealing at various temperatures for 6 h. Fig. 5B and C exhibits the spectral reflectivity of the solar absorber after 6 h thermal annealing at various temperatures. It can be seen that the spectral reflectivity of the tested samples barely changes from 200 °C to 600 °C, indicating its excellent high-temperature stability. The nearinfrared reflectivity starts to increase and the mid-infrared reflectivity begins to decrease dramatically when the annealing temperature is increased to 800 °C and 1000 °C. It indicates that spectral selectivity starts to fail possibly due to physical or chemical damages at high temperatures. Table 3 lists the overall thermal absorptance and thermal emittance as fabricated and after thermal annealing for 6 h at different temperatures by integrating the wavelength from 0.9 µm to 15 µm. It is shown that both thermal absorptance,  $\alpha_{\rm abs}$ , and emittance,  $\epsilon_{\rm abs}$ , changes greatly and  $\alpha_{abs}$  shows a potential of increase after 800 °C and 1000 °C annealing, which is also displayed in Fig. 5B and C that the cut-off wavelength begins to red-shift to a longer wavelength. Simultaneously,  $\epsilon_{\mathrm{abs}}$  increases with increasing of thermal annealing temperature, which indicates that the selectivity of the fabricated absorber becomes weaker because of the damages from high-temperature annealing. However, the fabricated samples still keep relatively low reflectivity (around 0.1 after 800 °C annealing, and around 0.2 after 1000 °C annealing) in the solar radiation region, and the reflectivity between 2.5  $\mu m$  and 7.0  $\mu m$ maintains at a low value. It indicates that the spectral selectivity has failed partially, but the solar absorber still works in a CSP system with a high CF (over 100) which dominates the energy conversion efficiency.

To illustrate the mechanism that causes the degradation at 800 °C, the samples are characterized under FE-SEM (SIGMA VP) before and after thermal annealing. Fig. 5D shows the SEM topographic images of the sample surface before and after being heated from 200 °C to 1000 °C for 6 h. From 25 °C to 200 °C, it shows no apparent changes. When the temperature keeps going up from 400 °C to 600 °C, the granulated protrusions with a diameter of about 20 nm show up at the samples surface, which can also be demonstrated in the reflectivity spectrum as seen in Fig. 5B and C. However, when the absorber is further heated to 800 °C, blisters with diameters around 40 µm are formed and cracks appear at 1000 °C, possibly due to the thermal press arising from the difference of thermal expansion coefficients of SiO<sub>2</sub> and W. Here, it leaves room for an improvement to approach to a prefect solar absorber by employing other materials with more similar thermal expansion coefficients with Al<sub>2</sub>O<sub>3</sub> as an anti-reflection coating to avoid the thermal stress.

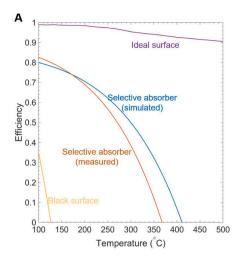
#### 5. Discussion

### 5.1. Solar conversion efficiency calculation for fabricated multilayer solar absorber

In order to quantitatively evaluate the thermal performance of the designed solar absorber, the solar-to-heat conversion efficiency is theoretically analyzed here according to Eq. (1). Since the designed structure of solar absorber is demonstrated to be angular-insensitive as well as thermally stable up to 600 °C, we use the reflectivity obtained from the UV/VIS/NIR spectrophotometer and FTIR spectrometer at room temperature (25 °C) for efficiency calculation, as shown in Fig. 5A. The  $\alpha'_{\lambda,abs}$  and  $\epsilon'_{\lambda,abs}$  in Eqs. (2) and (3) are assumed to be independent of temperature as observed in Fig. 5B. The spectral integration for  $\alpha'_{\lambda,abs}$  and  $\epsilon'_{\lambda,abs}$  is performed over wavelengths region from 0.3 µm to 16 µm, which cover the 97% of the solar radiation wavelength region. Only 5% of the blackbody thermal radiation fall outside this defined spectral region for a 400 °C blackbody.

Fig. 6A shows the solar-to-heat conversion as a function of absorber temperature,  $T_{\rm abs}$  under unconcentrated solar radiation for an ideal absorber surface, the multilayer solar absorber with spectral reflectivity taken from measurements or simulation, and a black surface (with unity absorptance over the entire wavelength of interest). The cut-off wavelength of an ideal solar absorber differs and is optimized at different operational temperatures to maximize the energy conversion efficiency, which indicates an upper limit of efficiency. The reflectivity of a black surface is zero over the entire wavelength region and shows no spectral selectivity.

It has been shown that the conversion efficiency of the solar absorber is 82.5% and 79.9% at an absorber temperature  $T_{\rm abs} = 100 \, ^{\circ}{\rm C}$ with the simulated or measured spectral reflectivity. The efficiency drops gradually to zero at the stagnation temperature of 410 °C and 367 °C for the solar absorber using simulated and measured optical properties, where absorbed solar energy equals to blackbody reemission energy (i.e., no solar thermal energy is harvested). The energy curves of solar absorber with measured and simulated radiative properties meet at 171 °C. The fabricated solar absorber has a higher efficiency than the designed one below 171 °C because the reflectivity of the fabricated absorber is lower from 0.3 µm to 0.5 µm, where nearly half of the solar radiation distributes than the simulation values, which is shown in Fig. 5A. Above 171 °C, the efficiency of as designed solar absorber exhibits more advantages of the conversion efficiency than the fabricated one. It has a higher stagnation temperature due to its higher reflectivity in the mid-infrared region for enhanced minimization of thermal re-emission. As a reference, the black surface converts 34.8% solar energy to heat at  $T_{\rm abs}$ , and its efficiency goes down to zero at 126 °C very quickly, which further demonstrates the significance of wavelength selectivity in maximizing solar-to-heat energy conversion efficiency. On the other hand, the efficiency of the multilayer absorber is 17% and 20% lower than the ideal surface at  $T_{\rm abs} = 100$  °C with measured optical properties and simulated radiative spectrum data, respectively. It mainly results from the larger thermal emittance in the mid-infrared region. Additionally, the reflectivity of the selective absorber within the solar radiation spectrum is higher than that of the ideal one. The difference between the designed solar absorber and the ideal surface becomes even larger with the increasing of the absorber operational temperature. It is because that the cut-off wavelength of the ideal solar absorber at a certain temperature is optimized according to its operational temperature. This yields that the ideal solar absorber can achieve a balance between the absorption of solar energy and the depression of the spontaneous blackbody irradiance. For example, when the operational temperature rises, the cut-off wavelength of the ideal solar absorber becomes smaller accordingly to lower the thermal irradiation from the blackbody irradiance since the blackbody radiation spectrum moves to shorter wavelengths. However, the cut-off wavelength of the proposed selective absorber keeps unchanged at  $1.7~\mu m$ .



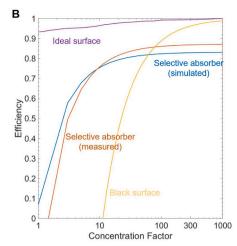


Fig. 6. (A) The calculated solar-to-heat energy conversion efficiency of an ideal selective absorber, the multilayer solar absorber with radiative properties of measured or simulated, and a black surface as a function of absorber operational temperature,  $T_{\rm abs}$ , under unconcentrated solar light; (B) solar-to-heat conversion efficiency for abovementioned four absorber surfaces as a function of concentration factor, CF, at an absorber operational temperature of  $T_{\rm abs} = 400\,^{\circ}{\rm C}$ .

Thus, when the operational temperature increases, the thermal loss arising from the blackbody radiance of the selective solar absorber before 1.7  $\mu m$  goes up because of the left-shift of the blackbody radiation at a higher temperature, and thus the solar-to-heat efficiency drops. Simultaneously, the reflectivity spectrum of the ideal surface changes more sharply at the cut-off point than the multilayer solar absorber, as shown in Fig. 5A. Therefore, the geometry parameters of the multilayer solar absorber need to be optimized to make the cut-off wavelength perfectly matched to the operational temperature.

The solar-to-heat energy conversion efficiency also varies with the concentration factor at CSP systems. Fig. 6B plots the efficiency as a function of concentration factor, CF, from 1 to 1000, at an absorber temperature,  $T_{\rm abs}$  = 400 °C, for a medium-temperature solar thermal application. The cut-off wavelength of the ideal selective absorber is optimized according to different CFs, and the corresponding efficiency goes up from 93.8% to near unity when the CF reaches 1000, which indicates an upper limit for the selective solar absorber performance. The energy efficiency of both multilayer solar absorber with measured or simulated optical properties and black surface keeps going up with an increase of CF. The efficiency of solar absorber with measured reflective spectrum is lower than with simulated radiative properties below 10 suns, it is because the simulated data has a higher emittance than the measured one, however, when the CF gets larger, the solar energy input will play a main role in the efficiency calculations and also the absorber with measured optical has lower reflectivity with solar radiation spectrum. For the black surface, its energy conversion efficiency becomes greater than zero at around 12 suns and climbs up approaching the thermal performance of ideal selective absorber when the CF = 1000. The solar radiation heat flux is much larger than the 600 °C blackbody thermal radiation under 1000 concentrated sunlight. From Fig. 6B, it can be concluded that the selective solar absorber has the advantage over the black surface below around 100 suns, while the selective solar absorber will fail when the CF becomes larger.

## 5.2. Thermal performance investigations under different concentration factor in a one-day sunlight cycle

The transient temperature fluctuations of the solar absorber under different concentration factor are simulated by solving Eq. (8), which is integrated over time to obtain the temperature evolution of solar absorber, as shown in Fig. 7. For each simulation, the initial temperature of the solar absorber is assumed to be the same as the ambient temperature.

Fig. 7 shows the transient temperature variations of the solar absorber under 1 sun (i.e. no optical concentration) and 20 suns, respectively, for fabricated selective absorber and black surface over a one-day sunlight cycle on July 10, 2018, in Boston, Massachusetts [48], which is a typical summer climate. Using the ambient temperature [48] and the solar irradiation data [49] of July 10, 2018, as the input data of Eq. (4), the temperature variations of the fabricated selective absorber and black surface are simulated from sunrise (5:00 a.m.) to one hour after sunset (8:00 p.m.). It can be observed that the highest temperature of the selective solar absorber is 273 °C and 675 °C for 1 sun and 20 suns, respectively, while the highest temperature of the black surface is 72 °C and 361 °C for 1 sun and 20 suns, respectively. It indicates a difference of their highest temperature between the selective and black solar absorber is 201 °C and 311 °C under 1 sun and 20 suns, respectively. It is obvious that the thermal performance of the selective solar absorber is overwhelmingly better than the black surface at any time under sunshine, and it reveals the significance of selective solar absorber in the CSP systems. It is worth to mention that, the temperature of the black surface starts to drop below the ambient temperature at about half-hour after the sunrise (5:30 a.m.) or halfhour before sunset (6:30 p.m.) and even drop below 0 °C at one hour after sunset (8:00 p.m.). It can be attributed to the radiative cooling properties of the black surface, considering that the black surface has unity emittance over the atmospheric transparent window (from 7.9 µm to 13 µm) and it has been discussed here [42,43].

#### 6. Conclusion

In this article, a multilayer selective solar absorber consists of SiO<sub>2</sub>-Al<sub>2</sub>O<sub>3</sub>-W-Al<sub>2</sub>O<sub>3</sub>-W stacks is theoretically designed, experimentally fabricated, optically characterized, and theoretically investigated its thermal performance. UV/VIS/NIR spectrophotometer and FTIR spectrometer measurements display excellent spectral selectivity, which demonstrates 0.12 reflectivity in the visible (0.38  $\sim$  0.74  $\mu$ m) and near-infrared (0.74  $\sim$  2.5  $\mu$ m) region and 0.93 reflectivity in the midinfrared (2.5  $\sim$  20  $\mu$ m) thermal region. Oblique reflectivity is also theoretically characterized to show low reflectivity over solar radiation region for both transverse electric and transverse magnetic polarization, which demonstrating its angular and polarization insensitivity. High-temperature stability test after annealing is investigated at 200 °C, 400 °C, 600 °C, 800 °C and 1000 °C for 6 h shows that the fabricated solar absorber keeps its wavelength selectivity at even 600 °C. The fabricated samples still maintain low reflectivity at the solar radiation regime but lose partial spectral selectivity at the mid-infrared region.

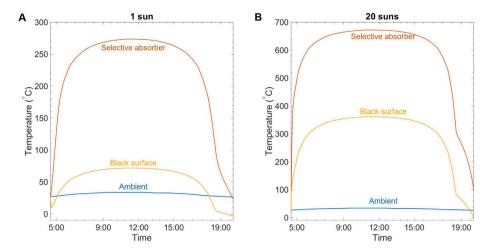


Fig. 7. (A) and (B) Thermal performance of the selective absorber (orange curve) and the black surface (yellow curve) over a one-day sunlight cycle from sunrise (5:00 a.m.) to one hour after sunset (8:00 p.m.) at varying ambient temperature (blue curve) two different concentration factor (1 sun and 20 suns).

This solar absorber after thermal annealing at 800 °C and 1000 °C can still work at a concentrated solar power system with an even higher concentration factor (over 100). The SEM topography images depict that the blisters and cracks appear in the samples after the high-temperature thermal treatment at, e.g., 800 °C and 1000 °C, which is attributed to the dissimilarity between the thermal expansion coefficients of  $\rm SiO_2$  and W. Theoretical efficiency analysis yields that the fabricated solar absorbers have a solar-to-heat conversion efficiency of over 82.5% and a high stagnation temperature of 675 °C under 20 suns. The potential high solar-to-heat conversion efficiency and high stagnation temperature further validate that the designed selective solar absorber provides a promising way to harness the solar energy for high-temperature concentrated solar power plants, solar thermophotovoltaic, and solar thermoelectric generator systems.

#### CRediT authorship contribution statement

Yanpei Tian: Conceptualization, Methodology, Validation, Writing. Xiaojie Liu: Writing - review & editing. Alok Ghanekar: Conceptualization, Methodology. Yi Zheng: Supervision, Writing - review & editing.

#### Declaration of competing interest

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

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