# Fine-Grained Trajectory Optimization of Multiple UAVs for Efficient Data Gathering from WSNs 

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#### Abstract

The increasing availability of autonomous smallsize Unmanned Aerial Vehicles (UAVs) has provided a promising way for data gathering from Wireless Sensor Networks (WSNs) with the advantages of high mobility, flexibility, and good speed. However, few works considered the situations that multiple UAVs are collaboratively used and the fine-grained trajectory plans of multiple UAVs are devised for collecting data from network including detailed traveling and hovering plans of them in the continuous space. In this paper, we investigate the problem of the Fine-grained Trajectory Plan for multi-UAVs (FTP), in which $m$ UAVs are used to collect data from a given WSN, where $m \geq 1$. The problem entails not only to find the flight paths of multiple UAVs but also to design the detailed hovering and traveling plans on their paths for efficient data gathering from WSN. The objective of the problem is to minimize the maximum flight time of UAVs such that all sensory data of WSN is collected by the UAVs and transported to the base station. We first propose a mathematical model of the FTP problem and prove that the problem is NP-hard. To solve the FTP problem, we first study a special case of the FTP problem when $m=1$, called FTP with Single UAV (FTPS) problem. Then we propose a constantfactor approximation algorithm for the FTPS problem. Based on the FTPS problem, an approximation algorithm for the general version of the FTP problem when $m>1$ is further proposed, which can guarantee a constant factor of the optimal solution. Afterwards, the proposed algorithms are verified by extensive simulations.


Index Terms- Unmanned Aerial Vehicle, Wireless Sensor Network, data gathering, mobile collector, trajectory optimization.

## I. Introduction

IN WIRELESS Sensor Networks (WSNs), sensors with limited battery resources are deployed on the detection areas to monitor the environment and their sensory data

[^0]needs to be collected to the base station [1], [2]. In the past decades, a huge amount of architectures tailored to Low Power Wide Area Networks (LPWANs), such as LORAWAN, SIGFOX, NB-IOT and LTE-M, have been prosperous in both urban and remote areas, in which the installation of few gateways over the territory allows to gather data even from sensors that are placed at different miles from the gateways. However, one of the prominent features, and subsequently one of the main problems with these architectures is that they rely heavily on infrastructure. Infrastructure-based networks tend to be susceptible to major damage by natural disasters and other catastrophic situations, such as hurricane, earthquake, volcanic, etc [3]. Therefore, in these situations, the fast and effective data collection methods from WSNs can effectively reduce the losses of lives and property. Due to the complexity of terrain and environment of the detection areas, data collection via multi-hop communication or ground mobile collectors faces many practical challenges. For examples, multi-hop communication makes the energy of sensors around the base station deplete much faster than others, which shortens the lifetime of the network; and the obstacles in the detection areas may inhibit ground mobile collectors to gather data from sensors, since the sensors are generally deployed in complex ground environments, especially in rugged and hilly terrain. The fast development of Unmanned Aerial Vehicles (UAVs) is providing an emerging solution to these challenging tasks due to their high maneuverability, good speed, flexibility, and increasing carrying capacity [4], [5]. The architecture of the UAV-based WSN is shown in Fig.1, in which sensors are deployed in the monitoring area to detect environment information. UAVs act as mobile collectors to gather data generated by sensors from WSN and transmit the data to the base station for data forwarding. Then the received data by the base station is transmitted to the users through the Internet or Satellite for further computational analysis to determine the appropriate response mechanism.

In recent years, there are many researches which proposed various problems and algorithms for the trajectory optimization of UAVs to effectively collect sensory data from WSNs, such as [6]-[9]. In [6], Kim et al. investigated the trajectory optimization problem of multiple UAVs, in which UAVs are employed to jointly collect sensory data from a given set of sensors to minimize the task completion time. However, their models of the problems are defined on the two-dimensional plane without considering flight altitude and data transmission expenditure for data gathering from sensors. In [7], Gong et al.


Fig. 1. The architecture of the UAV-based WSN.
studied the flight time minimization problem of UAV for data collection from WSN, in which they considered the case that the UAV can collect data from sensors during either cruising or hovering. However, they only gave the solution for the situation where the single UAV gathers sensory data from sensors located on a line. In [8], Ghorbel et al. proposed an energyefficient method to minimize the energy consumption of both the UAV and the sensors while accomplishing a tour to collect data from WSN. They took into account the total consumption of both traveling and hovering for data collection. But they only considered data gathering on the fixed hovering points and ignored the situation where the UAV can collect data during flight. In [9], Luo et al. investigated the Transportation and Communication Latency Optimization (TCLO) problem, which is to find the optimal flight plan of UAV such that all sensing data carried by sensors is collected by the UAV and transported to the base station, while the data collection latency of UAV is minimized. They considered the situation where single UAV collects data from sensors during flight, but they ignored the circumstance in which multiple UAVs are collaboratively used to gather sensory data from sensors.

To overcome the shortcomings of the above research works, in this paper, we focus on fine-grained trajectory plans of multiple UAVs in three-dimensional free space. It optimizes not only the paths of UAVs but also detailed hovering and traveling plans of UAVs for efficient data gathering from WSN. We aim at optimizing the flight trajectories of multiple UAVs such that all sensing data generated by sensors are transported to the base station and the maximum flight time of UAVs is minimized. The contributions of this paper are as below.
(1) We propose a multi-UAVs data gathering model in WSNs, which is called Fine-grained Trajectory Plan for multiUAVs ( $m$-UAVs) (FTP) problem, where $m \geq 1$. It is to find the optimal fine-grained flight plans of multiple UAVs to gather data from WSN such that the maximum time consumption of UAVs is minimized. Then we give a mathematical model for the FTP problem and prove that it is NP-hard.
(2) We first study a special case of the FTP problem when $m=1$, called FTP with Single UAV (FTPS), in which the single UAV is used to complete data gathering tasks from a given WSN. To solve the FTPS problem, we present another problem, namely the Path Plan of Single-UAV (PPS), which is to find a detailed flight plan of the UAV in the data collection area of a sensor. Then we propose an approximation
algorithm to solve the PPS problem. Based on the PPS problem, we devise an approximation algorithm FTPSA with the performance ratio $2+\varepsilon$ for the FTPS problem, where $0<\varepsilon<1$ is a constant.
(3) We extend the solution for the FTPS problem to the general FTP problem, where $m>1$, and propose an approximation algorithm FTPM with the performance ratio $3+\varepsilon$ for approximating the optimal solution of the FTP problem, where $0<\varepsilon<1$ is a constant.
(4) The extensive simulations are presented to verify the effectiveness of the proposed algorithm for the FTP problem.
The remainder of this paper is organized as follows. Section II introduces related works. In Section III, we introduce some models and definitions of the problem. In Section IV, we propose an approximation algorithm for solving the FTPS problem. Section V introduces an approximation algorithm for the FTP problem. Simulations are shown in Section VI. Section VII concludes this paper.

## II. Related Works

In this section, we briefly review the literature related to the trajectory optimization problems of UAVs as collectors for sensory data collection in WSNs. Based on the mobility of collectors, we classify the investigated problems into two different types: trajectory optimization of ground mobile collectors (e.g. robots and vehicles) in two-dimensional plane and trajectory optimization of UAVs in three-dimensional (3D) space.

Trajectory Optimization of Ground Mobile Collectors: In [10], Bhadauria and Isler introduced a path planning problem, called $k$-DGP, in which multiple robots are used to gather data from stationary devices with wireless communication capabilities in WSN. The objective of the problem is to compute tours of $k$ robots such that all data carried by sensors is collected by robots and the time cost of robots is minimized. In [11], Huang et al. investigated the data delivery delay problem in WSNs, in which mobile nodes attached to buses were used to collect data from sensors. The goal of the problem is to route delay sensitive data from sensors to mobile nodes within an allowed latency. In [12], Singh et al. proposed a scheme using an unequal fixed grid-based cluster along with a mobile data mule for data collection from the cluster heads in WSN, which could overcome the challenge of the high energy depletion rate in nodes near to the base station to maximize the lifetime of the network. In [13], Kumar and Dash investigated the data collection problem in WSN using a mobile collector, in which the mobile sink efficiently collects data from nearby sensors while moving along a pre-specified path with a constant speed such that the total data collected by the mobile collector is maximized with minimum energy consumption.
Trajectory Optimization of UAVs in 3D Space: In [14], Zeng et al. investigated the energy-efficient communication problem for a point-to-point link, in which a UAV is employed to communicate with a ground terminal for a finite time horizon, which is a new design framework that needs to jointly consider the communication throughput and the UAV's propulsion energy consumption. The objective of the problem aims at maximizing the energy efficiency in bits/Joule by optimizing the UAV's trajectory. In [15], Liu et al. studied
the problem of UAVs supported data collection for WSN and designed the flight paths for single UAV and multiple UAVs to maximize the capacity of sensors. However, they assumed that the flying paths of UAVs are fixed, which is not in line with the actual situation. In [16], Hamidullah et al. investigated the problem of path planning for multiple UAVs to collect data from several RoadSide Units (RSUs), whose goal is to find the time-optimal paths for these UAVs such that they can collectively visit all RSUs and gather all data from RSUs. However, they assumed that the altitude of the RSUs and UAVs are identical, which ignores the impact of flying height of UAVs on path optimization, and they used traditional genetic algorithm and harmony search algorithm to solve the problem. In [17], Liu et al. investigated the delay-tolerant sensory data gathering problem in UAV-aided WSNs, in which they consider both the sensor's transmission strategy and UAV's trajectory optimization to minimize the transmission energy consumption while guaranteeing the completed transmission within a given time. They considered the situations that the single UAV is used to collect data from sensors and divided the data transmission deadline into discrete time slots for designing UAV trajectory. In [18], Guo et al. studied jointly optimizing the UAV's time allocation between recharging and service, flight trajectory and transmit power allocation to maximize the minimum average rate among all ground users, in which the UAV can be recharged periodically at a fixed depot before providing communication service to ground users. In [19], Lee and Yu proposed the path planning optimization of rechargeable solar-powered UAV based on the gravitational potential energy to expand flight time without energy consumption. In [20], Natalizio et al. intrduced a novel trajectory planning problem for multiple UAVs that takes into account time and capacity constraints, such as airborne energy and limited computational resources. They solved these problem by leveraging the deployment of training and recharging areas (TRA) in the smart cities for providing sevices of securely recharging, updating and reconfiguration for UAVs. Then the 3D trajectory planning problem of UAVs moving through TRA was investigated by proposing an online approach. However, the method was only validated and tested through simulation without analyzing performance ratio.

In this paper, we investigate the FTP problem which is to optimize the fine-grained trajectories of multiple UAVs for efficient sensory data gathering from WSNs in 3D free space. It not only can overcome the challenges of data collection with ground mobile collectors such as rugged and hilly terrain of detection areas, low speed but also can conquer the shortcomings of UAV trajectory optimization problems in the above researches. Then we propose a constant factor approximation algorithm to solve the FTP problem, which optimizes not only the paths of UAVs but also detailed hovering and traveling plans of UAVs for efficient data gathering from WSN.

## III. Models and Definitions

In this section, we introduce some models and the definitions of the problem.


Fig. 2. UAVs act as mobile collectors to gather data from WSN.

## A. Network Model

As shown in Fig. 2, we consider a set of $n$ wireless sensors $S=\left\{s_{1}, s_{2}, \cdots, s_{n}\right\}$ located in the two-dimensional monitoring region $\Omega \subseteq \mathfrak{R}^{2}$ and they have the same three-dimensional transmission range $R$. Assume that each sensor $s_{i} \in S$ generates $V_{i}$ units of sensing data. For each $s_{i} \in S$, we use $T R\left(s_{i}\right)$ to denote the hemispheric region above the ground which is centered at $s_{i}$ and whose radius is $R$. There are several UAVs available to gather sensory data from sensors within the sensors' transmission range. Let $F=\left\{f_{1}, f_{2}, \cdots, f_{m}\right\}$ denote the set of $m$ UAVs, in which UAVs have the uniform horizontal flight speed $v_{f}$, the vertical flight speed $v_{h}$ and fly at a fixed altitude of $h$ when they fly in horizontal, where $h \leq R$. Let $h_{0}$ denote the minimum altitude of UAVs from the ground when they fly in vertical. In practice, $v_{f}$ and $v_{h}$ could correspond to the maximum horizontal speed and vertical speed required for minimizing the time consumption of UAVs, respectively and $h$ could correspond to the minimum altitude required for terrain or building avoidance without the need for frequent aircraft ascending and descending. In this paper, we do not consider other higher constraints such as acceleration, weight and steering angle of UAVs. The UAV $f_{k} \in F$ can collect sensory data from $s_{i} \in S$ if and only if it is in $T R\left(s_{i}\right)$. All UAVs will start from the stationary base station $s_{0}$ when performing their data collection duties and go back to $s_{0}$ after finishing their data collection tasks.

In this paper, we use the three-dimensional Cartesian coordinate system $X Y Z$ to mark the locations of sensors and UAVs, with all dimensions being measured in meters. Without loss of generality, we assume that all sensors in $S$ are randomly deployed in the first quadrant of the coordinate system and the $Z$ coordinates of them are zero, and let $\left(x_{i}, y_{i}, 0\right)$ denote the coordinates of sensor $s_{i} \in S$. For each sensor $s_{i} \in S$, since the horizontal flight altitude of UAV is $h$, the horizontal flying data collection area of UAV at flight altitude $h$ in $T R\left(s_{i}\right)$ is a circular area that is the cross-section between $\operatorname{TR}\left(s_{i}\right)$ and the plane $Z=h$ in the coordinate system. We use $N\left(s_{i}^{\prime}\right)$ to represent the circular area which is centered at $s_{i}^{\prime}$ and whose radius is $r=\sqrt{R^{2}-h^{2}}$, as the upper gray shaded area shown in Fig.3, where $s_{i}^{\prime}$ is the projection of $s_{i}$ on the $N\left(s_{i}^{\prime}\right)$ plane, and the $X$ and $Y$ coordinates of $s_{i}^{\prime}$ are the same as $s_{i}$ and its $Z$ coordinate is $h$. Let $D=\left\{N\left(s_{i}^{\prime}\right), N\left(s_{2}^{\prime}\right), \cdots, N\left(s_{n}^{\prime}\right)\right\}$. For each sensor $s_{i} \in S$, since the minimum flying altitude of UAVs is $h_{0}$ when they fly in vertical, the data collection area of UAV


Fig. 3. The data collection area $\Omega\left(s_{i}\right)$ is a cylinder whose top and bottom bases are $N\left(s_{i}^{\prime}\right)$ and $N\left(s_{i}^{\prime \prime}\right)$ respectively, and its altitude is $h-h_{0}$.
in $T R\left(s_{i}\right)$ is a cylinder $\Omega\left(s_{i}\right) \subset T R\left(s_{i}\right)$, one of whose bases is $N\left(s_{i}^{\prime}\right)$, which is formed by moving down the base $N\left(s_{i}^{\prime}\right)$ for the distance $h-h_{0}$. Let $N\left(s_{i}^{\prime \prime}\right)$ denote the other base of $\Omega\left(s_{i}\right)$ which is centered at $s_{i}^{\prime \prime}$ and whose radius is also $r$, where $s_{i}^{\prime \prime}$ is the projection of $s_{i}$ on $N\left(s_{i}^{\prime \prime}\right)$ and its $X$ and $Y$ coordinates are the same as $s_{i}$ and its $Z$ coordinate is $h_{0}$, as shown in Fig. 3. That means the UAV can collect data from $s_{i}$ only when it travels (or hovers) in $\Omega\left(s_{i}\right)$. Let $\Theta=\left\{\Omega\left(s_{1}\right), \Omega\left(s_{2}\right), \cdots, \Omega\left(s_{n}\right)\right\}$ be the set of the data collection areas of UAVs. For simplicity, we denote $\left\{x_{0}, y_{0}, z_{0}\right\}$ as the coordinate of $s_{0}$. If $U_{k}$ is the flight path of $f_{k} \in F$, then we use $L\left(U_{k}\right)$ to represent the length of $U_{k}$. For any two different points $u$ and $w$ on $U_{k}$, we use $P_{u, w}$ to denote the path between them on $U_{k}$ and $(u, w)$ to represent the line segment for connecting the two points. Let $L\left(P_{u, w}\right)$ stand for the length of $P_{u, w}$ and $d_{u, w}$ be the Euclidean distance between $u$ and $w$.

For any pair of sensors $s_{i}, s_{j} \in S$, if they are within each other's communication range, then they can communicate with each other. Therefore, the cluster head sensors can be selected by clustering algorithms from sensors in the network and could enable basic data gathering work in WSNs. For this reason, the UAVs can focus on gathering sensing data from cluster head sensors, such as [12], [21]. In this paper, we assume that any two data collection areas $\Omega\left(s_{i}\right) \in \Theta$ and $\Omega\left(s_{j}\right) \in \Theta$ are disjoint from each other.

## B. Communication Model

For each $f_{k} \in F$, it can gather data from sensor $s_{i} \in S$ only when it is in $\Omega\left(s_{i}\right)$. As the data transmission rate from $s_{i}$ to $f_{k}$ changes with the varying transmission distance under signal path loss model, in this paper, we employ the LOS ground-toair channel model between UAVs and sensors with path loss exponent $2 \leq \alpha<4$ that was adopted by [7], [14]. Therefore, the data transmission rate from $s_{i}$ to $f_{k}$ can be expressed as

$$
\begin{equation*}
C_{i k}=\frac{1}{2} W \log _{2}\left(1+\frac{\beta_{0} P}{\sigma^{2} d_{s_{i}, f_{k}}^{\alpha}}\right)=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{d_{s_{i}, f_{k}}^{\alpha}}\right), \tag{1}
\end{equation*}
$$

where $d_{s_{i}, f_{k}}$ is the Euclidean distance between $s_{i}$ and $f_{k}, W$ represents the channel bandwidth, $\beta_{0}$ denotes the channel power at the reference distance $d_{0}=1 \mathrm{~m}, \sigma^{2}$ is the Gaussian noise power at the UAVs, and $\gamma_{0}=\frac{\beta_{0} P}{\sigma^{2}}$ denotes the reference signal-to-noise ratio (SNR) at the reference distance $d_{0}=1 \mathrm{~m}$.

## C. Definition of the Problem

For each $f_{k} \in F$, let $U_{k}$ represent the flight tour of $f_{k}$ where $U_{k}$ is composed of the horizontal flight path $U_{k}^{f}$ and vertical flight path $U_{k}^{h}$, i.e., $U_{k}=U_{k}^{f} \cup U_{k}^{h}$. We use $H_{k}$ to denote the set of hovering points of $f_{k}$ on $U_{k}$ and $T_{k}$ to be the set of hovering
times of $f_{k}$ at the hovering points in $H_{k}$. For any hovering point $H P_{s_{i}}^{k} \in H_{k}$ of $f_{k}$ in $\Omega\left(s_{i}\right)$, there exists a corresponding hovering time $t_{s_{i}}^{k} \in T_{k}$. Suppose $\Phi(U, H, T)$ is a feasible flight plan of $m$ UAVs such that all sensory data of sensors can be collected by $m$ UAVs and transported to the base station, in which $U=\left\{U_{1}, U_{2}, \cdots, U_{m}\right\}, H=\left\{H_{1}, H_{2}, \cdots, H_{m}\right\}$ and $T=\left\{T_{1}, T_{2}, \cdots, T_{m}\right\}$. We use $\phi\left(U_{k}, H_{k}, T_{k}\right)$ to denote the flight plan of $f_{k}$ and $E_{\phi}^{k}=L\left(U_{k}^{f}\right) / v_{f}+L\left(U_{k}^{h}\right) / v_{h}+\sum_{t_{s_{i}} \in T_{k}} t_{s_{i}}^{k}$ to represent the time cost of $f_{k}$ when its flight plan $\phi\left(U_{k}, H_{k}, T_{k}\right)$ has been determined. In this paper, we aim at finding an optimal flight plan $\Phi(U, H, T)$ of $m$ UAVs such that the maximum time consumption $E_{\Phi}=\max \left\{E_{\phi}^{k} \mid f_{k} \in F\right\}$ is minimized.

We refer to the problem as a Fine-grained Trajectory Plan for multi-UAVs (FTP), whose detailed definition is shown as follows.

Definition 1 FTP: Given a set $S=\left\{s_{1}, s_{2}, \cdots, s_{n}\right\}$ of $n$ sensors in which each sensor $s_{i}$ has $V_{i}$ units of sensing data, a set of data collection areas $\Theta=\left\{\Omega\left(s_{1}\right), \Omega\left(s_{2}\right), \cdots, \Omega\left(s_{n}\right)\right\}$, a set of disks $D=\left\{N\left(s_{i}^{\prime}\right), N\left(s_{2}^{\prime}\right), \cdots, N\left(s_{n}^{\prime}\right)\right\}$, a set $F=$ $\left\{f_{1}, f_{2}, \cdots, f_{m}\right\}$ of $m$ UAVs in which all UAVs have uniform horizontal flight speed $v_{f}$, vertical flight speed $v_{h}$, flight altitude $h$ for horizontal flying, the minimum vertical flight altitude $h_{0}$ and the same initial location $s_{0}$, the Fine-grained Trajectory Plan for multi-UAVs (FTP) problem is to find a flight plan $\Phi(U, H, T)$ for $m$ UAVs such that
(1) each tour $U_{k} \in U$ starts from and ends at $s_{0}$,
(2) each $f_{k} \in F$ can collect data from $s_{i}$ when it flies in (or on the border of) $\Omega\left(s_{i}\right)$,
(3) for each $s_{i} \in S$, the UAVs can only fly vertically in the area $\Omega\left(s_{i}\right) \backslash N\left(s_{i}^{\prime}\right)$,
(4) for each $s_{i} \in S$, there exists at least a tour $U_{k} \in U$ passing through $\Omega\left(s_{i}\right)$ and a hovering point $H P_{s_{i}}^{k} \in H_{k} \cap \Omega\left(s_{i}\right)$ with hovering time $t_{s_{i}}^{k} \in T_{k}$ for some $f_{k} \in F$ and $V_{i}$ units of sensing data is transmitted to the base station, and
(5) $E_{\Phi}=\max \left\{E_{\phi}^{k} \mid f_{k} \in F\right\}$ is minimized, where $E_{\phi}^{k}=$ $L\left(U_{k}^{f}\right) / v_{f}+L\left(U_{k}^{h}\right) / v_{h}+\sum_{t_{s_{i}}^{k} \in T_{k}} t_{s_{i}}^{k}$.

Next, we will introduce the mathematical formulation for the FTP problem. We use $q_{i}^{k}$ to denote the projection point of $H P_{s_{i}}^{k}$ on $N\left(s_{i}^{\prime}\right)$. For simplicity, let $V_{0}=0$ denote the amount of data stored by $s_{0}, q_{0}^{k}=\left(x_{0}, y_{0}, h\right)$, and $H P_{s_{0}}^{k}=s_{0}$ for each $f_{k} \in F$. Let $V_{i}^{k}$ represent the amount of data collected from $s_{i}$ by $f_{k}$. We define binary variable $a_{i j k}$ as below.

$$
a_{i j k}= \begin{cases}1, & \text { if } f_{k} \text { visits } \Omega\left(s_{j}\right) \text { after } \Omega\left(s_{i}\right)  \tag{2}\\ 0, & \text { otherwise }\end{cases}
$$

We can obtain the following mathematical formulation of the FTP problem.

$$
\begin{equation*}
\text { Minimize } \max _{1 \leq k \leq m} \sum_{i=0}^{n} \sum_{\substack{j=0 \\ j \neq i}}^{n}\left(\frac{d_{q_{i}^{k}, q_{j}^{k}}}{v_{f}}+\frac{2 \cdot d_{q_{j}^{k}, H P_{s_{j}}^{k}}}{v_{h}}+t_{s_{j}}^{k}\right) \cdot a_{i j k} \tag{3}
\end{equation*}
$$

$$
\sum_{k=1}^{\substack{\text { s.t. } \\ m}} \sum_{j=1}^{n} a_{0 j k}=m
$$

$$
\begin{align*}
& \sum_{k=1}^{m} \sum_{i=1}^{n} a_{i 0 k}=m  \tag{5}\\
& 1 \leq \sum_{k=1}^{m} \sum_{i=0}^{n} a_{i j k} \leq m \quad j=1,2, \cdots, n  \tag{6}\\
& \sum_{i=0}^{n} a_{i p k}-\sum_{j=0}^{n} a_{p j k}=0 \quad k=1,2, \cdots, m,  \tag{7}\\
& p=1,2, \cdots, n \\
& \sum_{i=0}^{n} \sum_{k=1}^{m} a_{i j k} \cdot V_{j}^{k}=V_{j} \quad j=1,2, \cdots, n  \tag{8}\\
& \sum_{s_{i} \in G s_{j} \in G} a_{i j k} \leq|G|-1 \quad \forall G \subset S, G \neq \emptyset,  \tag{9}\\
& k=1,2 \cdots, m \\
& a_{i j k} \in\{0,1\} \quad i=0,1, \cdots, n  \tag{10}\\
& j=0,1, \cdots, n, \quad i \neq j, \quad k=1,2, \cdots, m \\
& H P_{s_{i}}^{k} \in \Omega\left(s_{i}\right) \quad i=1,2, \cdots, n, k=1,2, \cdots, m  \tag{11}\\
& q_{i}^{k} \in N\left(s_{i}^{\prime}\right) \quad i=1,2, \cdots, n, k=1,2, \cdots, m \tag{12}
\end{align*}
$$

Constraints (4) and (5) express that each UAV goes from the depot $s_{0}$ to any data collection area and comes back to the depot. Constraint (6) states that each data collection area should be visited by at least one UAV and at most $m$ UAVs. Constraint (7) is the flow conservation constraint which ensures that once a UAV visits a data collection area, it must also depart from the same area. Constraint (8) ensures that the total amount of data collected from $s_{j}$ by the visited UAVs is $V_{j}$. Constraint (9) ensures connectivity requirement for the solution, i.e., prevents from formating subtours of cardinality $G$ not including the depot $s_{0}$, where $G$ is a subset of $S$. Constraint (10) defines the domain of the instance. Constraints (11) and (12) limit the position ranges of hovering point $H P_{s_{i}}^{k}$ and its projection point $q_{i}^{k}$ in each data collection area $\Omega\left(s_{i}\right)$ for any $f_{k} \in F$.

According to the definition of the FTP problem, the hovering time $t_{s_{j}}^{k}$ in the objective formula (3) can be computed as below. If $a_{i j k}=1$ and $a_{j p k}=1$, then we compute the intersection point $q_{b}^{j}$ between the line segment $\left(q_{i}^{k}, q_{j}^{k}\right)$ and $N\left(s_{j}^{\prime}\right)$ and compute the intersection point $q_{e}^{j}$ between the line segment $\left(q_{j}^{k}, q_{p}^{k}\right)$ and $N\left(s_{j}^{\prime}\right)$. Assume that the time of $f_{k}$ arriving at $q_{b}^{j}$ is $t_{0}$. Let $\Gamma=t_{0}+\frac{{ }_{q_{b}^{j}, q_{j}^{k}}}{v_{f}}+2 \cdot \frac{{ }^{d} q_{j}^{k}, H P_{s_{i}}^{k}}{v_{h}}+\frac{{ }_{q_{j}^{k}}{ }_{k} q_{e}^{j}}{v_{f}}$ and $t_{1}=\frac{{ }_{q_{q_{b}}, q_{j}^{k}}}{v_{f}}+\frac{{ }_{q_{j}^{k}}{ }_{j}, H P_{s_{i}^{k}}^{k}}{v_{h}}$. Assume the coordinate of $f_{k}$ at time $t \in\left[t_{0}, \Gamma\right]$ is $\left(x^{k}(t), y^{k}(t), z^{k}(t)\right)$. Then, at time $t \in\left[t_{0}, \Gamma\right]$, the data transmission rate from $s_{j}$ to $f_{k}$ can be expressed as

$$
\begin{equation*}
C_{j k}(t)=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{d_{s_{j}, f_{k}}^{\alpha}(t)}\right), \tag{13}
\end{equation*}
$$

where $d_{s_{j}, f_{k}}(t)=\sqrt{\left(x^{k}(t)-x_{j}\right)^{2}+\left(y^{k}(t)-y_{j}\right)^{2}+\left(z^{k}(t)-0\right)^{2}}$. Therefore, the hovering time of $f_{k}$ at $H P_{s_{j}}^{k}$ can be written as

$$
\begin{equation*}
t_{s_{j}}^{k}=\frac{V_{j}^{k}-\int_{t_{0}}^{\Gamma} C_{j k}(t) d t}{C_{j k}\left(t_{1}\right)} \tag{14}
\end{equation*}
$$

In the following theorem, we will prove that the FTP problem is NP-hard.


Fig. 4. An example of the flight path $U_{s_{i}}$ of UAV in $\Omega\left(s_{i}\right)$, which is composed of $U_{s_{i}}^{f}$ and $U_{s_{i}}^{h}$, where $U_{s_{i}}^{f}=\left(b_{s_{i}}, q_{i}^{\prime}\right) \cup\left(q_{i}^{\prime}, e_{s_{i}}\right)$ and $U_{s_{i}}^{h}=\left(q_{i}^{\prime}, H P_{s_{i}}\right) \cup\left(H P_{s_{i}}, q_{i}^{\prime}\right)$.

## Theorem 1: The problem FTP is NP-hard.

Proof: If we set $V_{i}=0$ for each sensor $s_{i} \in S, R=0$, $m=1$ and $h=h_{0}=0$, then the FTP problem can be reduced to the well-known traveling salesman problem (TSP), which is proved NP-hard [22]. Since a special case of the FTP problem is NP-hard, the FTP problem is also NP-hard.

In the FTP problem, a special case is that the single UAV (i.e., $m=1$ ) is used for gathering all sensory data from WSN, which is called the FTP with Single UAV (FTPS) problem. Based on Theorem 1, we can find that the FTPS problem is also NP-hard.

## IV. Algorithm for the FTPS Problem

In this section, we propose an approximation algorithm to solve the FTPS problem. According to the definition of the problem, we can find that the flight plan of UAV consists of two parts. The first is the path for connecting all data collection areas in $\Theta$. The second is the flight plan of UAV in every data collection area $\Omega\left(s_{i}\right)$ for gathering data from $s_{i} \in S$ including horizontal flight path, vertical flight path and hovering point with corresponding hovering time. Therefore, to solve the FTPS problem, we introduce two other problems, Euclidean TSP with Neighborhoods (TSPN) and Path Plan of Single-UAV (PPS), as shown in Definitions 2 and 3, which can be used as subroutines for the FTPS problem.
Definition 2 TSPN: Given a collection of $n$ disks, $D=$ $\left\{N\left(s_{i}^{\prime}\right), N\left(s_{2}^{\prime}\right), \cdots, N\left(s_{n}^{\prime}\right)\right\}$ where the disks are equal-size and disjoint each other, the TSPN problem aims to find a shortest tour $U_{f}^{\prime}$ that visits all disks in $D$.

The TSPN problem is proved NP-hard, and there exists a $(1+\varepsilon)$-approximation algorithm for the problem in [23], where $0<\varepsilon<1$.
The PPS problem aims at finding an optimal flight plan $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ of UAV in the data collection area $\Omega\left(s_{i}\right) \in \Theta$, as shown in Fig. 4, such that $V_{i}$ units of data carried by $s_{i}$ is collected by the UAV, where $U_{s_{i}}$ that consists of the horizontal flight path $U_{s_{i}}^{f}$ and vertical flight path $U_{s_{i}}^{h}$ is the traveling path of UAV in $\Omega\left(s_{i}\right)$, which starts from a given border point $b_{s_{i}}$ of $N\left(s_{i}^{\prime}\right)$ and ends at another border point $e_{s_{i}}$ of $N\left(s_{i}^{\prime}\right)$, and $H P_{s_{i}}$ represents the hovering point of UAV with hovering time $t_{s_{i}}$ on $U_{s_{i}}$. The objective of the problem is to minimize the time $\operatorname{cost} E_{\varphi}^{i}=L\left(U_{s_{i}}^{f}\right) / v_{f}+L\left(U_{s_{i}}^{h}\right) / v_{h}+t_{s_{i}}$ of UAV. More formally, we formulate this problem as below.
Definition 3 PPS: Given a sensor $s_{i}$ with $V_{i}$ units of sensing data, a data collection area $\Omega\left(s_{i}\right)$, a horizontal flying data collection area $N\left(s_{i}^{\prime}\right)$, a border point $b_{s_{i}}$ of $N\left(s_{i}^{\prime}\right)$ and a UAV with horizontal flight speed $v_{f}$, vertical flight speed $v_{h}$, horizontal flight altitude $h$, the minimum vertical flight altitude

```
Algorithm 1 PPSA
    Input: \(\Omega\left(s_{i}\right), N\left(s_{i}^{\prime}\right), b_{s_{i}}=\left(x_{i}^{b}, y_{i}^{b}, h\right), s_{i}^{\prime}=\left(x_{i}, y_{i}, h\right), V_{i}, r\),
            \(n, h, h_{0}, v_{f}, v_{h}, \mathrm{~W}, \gamma_{0}, M\)
    Output: \(\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)\)
    \(\delta=\frac{r}{6 n},\left(p_{0}, p_{1}, \cdots, p_{6 n}\right)=\left(b_{s_{i}}, s_{i}^{\prime}\right) ;\)
    \(\eta=\frac{\delta}{v_{f}} \cdot v_{h}, \tau=\left\lceil\frac{h-h_{0}}{\eta}\right\rceil ;\)
    For any \(0 \leq t \leq 6 n,\left(q_{t, 0}, q_{t, 1}, \cdots, q_{t, \tau}\right)=\left(p_{t}, p_{t}^{\prime}\right)\);
    for \(t\) from 0 to \(6 n\) do
        for \(l\) from 0 to \(\tau\) do
            \(C_{q_{t, l}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+(h-l \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right) ;\)
            if \(t=0\) then
                1) If \(l=0\), then \(V_{t, l}=0\);
                            2) If \(1 \leq l<\tau\), then \(V_{t, l}=\sum_{j=0}^{l-1} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}\);
            3 ) If \(l=\tau\), then
            \(V_{t, l}=\sum_{j=0}^{l-2} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}+\frac{2\left(h-h_{0}-(l-1) \eta\right)}{v_{h}} \cdot C_{q_{t, l-1}} ;\)
            else
                    1) If \(l=0\), then \(V_{t, l}=\sum_{k=0}^{t-1} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}\);
            2) If \(1 \leq l<\tau\), then
                    \(V_{t, l}=\sum_{k=0}^{t-1} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}+\sum_{j=0}^{l-1} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}} ;\)
            3) If \(l=\tau\), then
                    \(V_{t, l}=\sum_{k=0}^{t-1} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}+\sum_{j=0}^{l-2} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}+\)
                    \(\frac{2\left(h-h_{0}-(l-1) \eta\right)}{v_{h}} \cdot C_{q_{t, l-1}} ;\)
                end
            if \(V_{t, l} \geq V_{i}\) then
                \(E_{t, l}^{i}=M\);
            else
                \(E_{t, l}^{i}=2 \cdot\left(\frac{t \cdot \delta}{v_{f}}+\frac{l \cdot \eta}{v_{h}}\right)+\frac{V_{i}-V_{t, l}}{C_{q_{t, l}}} ;\)
                end
        end
    end
    \(E_{\varphi}^{i}=\min \left\{E_{t, l}^{i} \mid 0 \leq t \leq 6 n, 0 \leq l \leq \tau\right\} ;\)
    Return the current \(t\) and \(l, t_{s_{i}}=\frac{V_{i}-V_{t, l}}{C_{q_{t, l}}}\);
    if \(t=6 n\) then
        if \(l=\tau\) then
            \(H P_{s_{i}}=\left(x_{i}, y_{i}, h_{0}\right)\)
        else
            \(H P_{s_{i}}=\left(x_{i}, y_{i}, h-l \cdot \eta\right)\)
        end
    else
        \(\lambda=\frac{t \cdot \delta}{r-t \cdot \delta} ;\)
        if \(l=\tau\) then
            \(H P_{s_{i}}=\left(\frac{x_{i}^{b}+\lambda \cdot x_{i}}{1+\lambda}, \frac{y_{i}^{b}+\lambda \cdot y_{i}}{1+\lambda}, h-h_{0}\right) ;\)
        else
            \(H P_{s_{i}}=\left(\frac{x_{i}^{b}+\lambda \cdot x_{i}}{1+\lambda}, \frac{y_{i}^{b}+\lambda \cdot y_{i}}{1+\lambda}, h-l \cdot \eta\right) ;\)
        end
    end
\(42 U_{s_{i}}^{f}=\bigcup_{0 \leq k<t}\left(q_{k, 0}, q_{k+1,0}\right) \cup \bigcup_{0 \leq k<t}\left(q_{k+1,0}, q_{k, 0}\right)\);
\(43 U_{s_{i}}^{h}=\bigcup_{0 \leq j<l}\left(q_{t, j}, q_{t, j+1}\right) \cup \bigcup_{0 \leq j<l}\left(q_{t, j+1}, q_{t, j}\right)\);
\(44 U_{s_{i}}=U_{s_{i}}^{f} \cup U_{s_{i}}^{h}\);
```

$h_{0}$, the Path Plan of Single-UAV (PPS) problem is to find a flight plan $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ of UAV in $\Omega\left(s_{i}\right)$ such that
(1) $U_{s_{i}}$ starts from $b_{s_{i}}$ and ends at another border point $e_{s_{i}}$ (including $b_{s_{i}}$ ) of $N\left(s_{i}^{\prime}\right)$,
(2) the UAV can only fly vertically in the area $\Omega\left(s_{i}\right) \backslash N\left(s_{i}^{\prime}\right)$,
(3) UAV can gather data from $s_{i}$ during flying on $U_{s_{i}}$ and have a hovering point $H P_{s_{i}}$ with hovering time $t_{s_{i}}$ on $U_{s_{i}}$ for gathering the remaining data from $s_{i}$,
(4) $V_{i}$ units of sensing data is transmitted to the UAV, and
(5) $E_{\varphi}^{i}=L\left(U_{s_{i}}^{f}\right) / v_{f}+L\left(U_{s_{i}}^{h}\right) / v_{h}+t_{s_{i}}$ is minimized.

## A. Algorithm for the PPS Problem

In this subsection, we propose an approximation algorithm for solving the PPS problem, which is called PPSA. The objective of the algorithm is to find a flight plan $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ of single UAV in $\Omega\left(s_{i}\right)$ such that the total time cost of UAV in $\Omega\left(s_{i}\right)$

$$
E_{\varphi}^{i}=\frac{L\left(U_{s_{i}}^{f}\right)}{v_{f}}+\frac{L\left(U_{s_{i}}^{h}\right)}{v_{h}}+t_{s_{i}} \text { is minimized }
$$

where $U_{s_{i}}=U_{s_{i}}^{f} \cup U_{s_{i}}^{h}$.
Before describing the algorithm, we introduce some terms and notations. Suppose the coordinates of $b_{s_{i}}$ are $\left(x_{1}^{b}, y_{1}^{b}, h\right)$. Initially, we divide $r$ into $6 n$ equal parts and set $\delta=\frac{r}{6 n}$. Let $p_{0}=b_{s_{i}}$ and $p_{6 n}=s_{i}^{\prime}$. Afterwards, we use $\left(p_{0}, p_{1}, \cdots, p_{6 n}\right)$ to represent the line segment $\left(b_{s_{i}}, s_{i}^{\prime}\right)$, where $p_{t}$ is an equidistant point and $d_{p_{t}, p_{t+1}}=\delta$ for any $0 \leq t \leq 6 n-1$. For arbitrary point $p_{t} \in\left\{p_{0}, p_{1}, \cdots, p_{6 n}\right\}$, there exists a projection $p_{t}^{\prime}$ on $N\left(s_{i}^{\prime \prime}\right)$. Let $\eta=\frac{\delta}{v_{f}} \cdot v_{h}$. We divide the line segment $\left(p_{t}, p_{t}^{\prime}\right)$ into $\tau$ parts, where $\tau=\left\lceil\frac{h-h_{0}}{\eta}\right\rceil$. The first $\tau-1$ parts are equal and their length is $\eta$, and the length of the last part is less than or equal to $\eta$. For arbitrary $0 \leq t \leq 6 n$, we let $\left(q_{t, 0}, q_{t, 1}, \cdots, q_{t, \tau}\right)$ denote the line segment $\left(p_{t}, p_{t}^{\prime}\right)$, where $q_{t, l}$ is a breakpoint and $d_{q_{t, l}, q_{t, l+1}}=\eta$ for any $0 \leq l \leq \tau-2$, and $d_{q_{t, \tau-1}, q_{t . \tau}}=h-h_{0}-$ $(\tau-1) \cdot \eta$. Since the time complexity for calculating integral function grows exponentially, we use the amount of data collected by UAV during hovering at the starting point of a very short path to approximate the amount of data collected by UAV during flying on the path in the same time. For any $0 \leq t<6 n$, we use $\frac{2 \cdot d_{p_{t, 0}, p_{t+1,0}}}{v_{f}} \cdot C_{q_{t, 0}}$ to approximate the size of data collected by UAV during flying on the round trip of $\left(p_{t, 0}, p_{t+1,0}\right)$, where $C_{q_{t, 0}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)$ is the data transmission rate of UAV when it hovers at $q_{t, 0}$. By taking $t$, for any $0 \leq$ $l<\tau$, let $\frac{2 \cdot d_{q_{t, l}, q_{t, l+1}}}{v_{h}} \cdot C_{q_{t, l}}$ approximate the amount of data collected by UAV during flying on the round trip of $\left(q_{t, l}, q_{t, l+1}\right)$, where $C_{q_{t, l}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+(h-l \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)$ represents the data transmission rate of UAV when it hovers at $q_{t, l}$.

The PPSA algorithm consists of four steps as follows.
In the first step, we set $\delta=\frac{r}{6 n}$ and $\left(p_{0}, p_{1}, \cdots, p_{6 n}\right)=$ $\left(b_{s_{i}}, s_{i}^{\prime}\right)$. For each $p_{t} \in\left\{p_{0}, p_{1}, \cdots, p_{6 n}\right\}$, we compute its projection point $p_{t}^{\prime}$ on $N\left(s_{i}^{\prime \prime}\right)$. Let $\eta=\frac{\delta}{v_{f}} \cdot v_{h}$ and $\tau=\left\lceil\frac{h-h_{0}}{\eta}\right\rceil$. For arbitrary $0 \leq t \leq 6 n$, we set $\left(q_{t, 0}, q_{t, 1}, \cdots, q_{t, \tau}\right)=\left(p_{t}, p_{t}^{\prime}\right)$. For any $0 \leq l \leq \tau-2$, let $d_{q_{t, l}, q_{t, l+1}}=\eta$ and $d_{q_{t, \tau-1}, q_{t, \tau}}=$ $h-h_{0}-(\tau-1) \eta$.

In the second step, for any $0 \leq t \leq 6 n$ and $0 \leq l \leq \tau$, we compute the time consumption $E_{t, l}^{i}$ of UAV for gathering $V_{i}$ units of data from $s_{i}$ when its hovering point is located at $q_{t, l}$. Let $V_{t, l}$ denote the amount of data collected by UAV during traveling from $b_{s_{i}}$ to $q_{t, l}$ and from $q_{t, l}$ to $b_{s_{i}}$. We compute the value of $V_{t, l}$ in the light of the two cases: $t=0$ and $0<t \leq 6 n$. Then we compute the time cost $E_{t, l}^{i}$ of UAV for each of cases as below.
(1) $t=0$

- If $l=0$, then $V_{t, l}=0$.
- If $1 \leq l<\tau$, then $V_{t, l}=\sum_{j=0}^{l-1} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}$.
- If $l=\tau$, then $V_{t, l}=\sum_{j=0}^{l-2} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}+\frac{2\left(h-h_{0}-(l-1) \eta\right)}{v_{h}} \cdot C_{q_{t, l-1}}$.
(2) $0<t \leq 6 n$
- If $l=0$, then $V_{t, l}=\sum_{k=0}^{t-1} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}$.
- If $1 \leq l<\tau$, then $V_{t, l}=\sum_{k=0}^{t-1} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}+\sum_{j=0}^{l-1} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}$.
- If $l=\tau$, then $V_{t, l}=\sum_{k=0}^{t-1} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}+\sum_{j=0}^{l-2} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}+$ $\frac{2\left(h-h_{0}-(l-1) \eta\right)}{v_{h}} \cdot C_{q_{t, l-1}}$.
After obtaining the value of $V_{t, l}$, we compare $V_{t, l}$ with $V_{i}$. If $V_{t, l} \geq V_{i}$, then we set $E_{t, l}^{i}=M$, otherwise, $E_{t, l}^{i}=2 \cdot\left(\frac{t \cdot \delta}{v_{f}}+\frac{l \cdot \eta}{v_{h}}\right)+\frac{V_{i}-V_{t, l}}{C_{q_{t, l}}}$, where $M$ is a very large real number to control the end of the algorithm.

In the third step, the minimum time cost of UAV is computed as $E_{\varphi}^{i}=\min \left\{E_{t, l}^{i} \mid 0 \leq t \leq 6 n, 0 \leq l \leq \tau\right\}$, and the corresponding $t$ and $l$ are obtained.

In the fourth step, based on the values of $t$ and $l$, we compute the flight plan $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ of UAV in the area $\Omega\left(s_{i}\right)$. Firstly, for any $0 \leq t<6 n$, we let $\lambda=\frac{t \cdot \delta}{r-t \cdot \delta}$ and $\lambda=0$ for $t=6 n$. Then based on the different values of $l$ obtained from the above processes, we can compute the coordinates of $H P_{s_{i}}$ with the following cases.

- If $0 \leq l<\tau$, then $H P_{s_{i}}=\left(\frac{x_{i}^{b}+\lambda \cdot x_{i}}{1+\lambda}, \frac{y_{i}^{b}+\lambda \cdot y_{i}}{1+\lambda}, h-l \cdot \eta\right)$.
- If $l=\tau$, then $H P_{s_{i}}=\left(\frac{x_{i}^{b}+\lambda \cdot x_{i}}{1+\lambda}, \frac{y_{i}^{b}+\lambda \cdot y_{i}}{1+\lambda}, h-h_{0}\right)$.

Based on the coordinates of $H P_{s_{i}}$, we compute the traveling path $U_{s_{i}}$ and of UAV in $\Omega\left(s_{i}\right)$ and its hovering time $t_{s_{i}}$ on $H P_{s_{i}}$, where $U_{s_{i}}=\bigcup_{0 \leq k<t}\left(q_{k, 0}, q_{k+1,0}\right) \cup \bigcup_{0 \leq j<l}\left(q_{t, j}, q_{t, j+1}\right) \cup$ $\bigcup_{0 \leq k<t}\left(q_{k+1,0}, q_{k, 0}\right) \cup \bigcup_{0 \leq j<l}\left(q_{t, j+1}, q_{t, j}\right)$ and $t_{s_{i}}=\frac{V_{i}-V_{t, l}}{C_{q_{t, l}}}$.

Note that for any $t$ and $l$, the line segments $\left(q_{t, l}, q_{t, l+1}\right)$ and $\left(q_{t, l+1}, q_{t, l}\right)$ (or $\left(q_{t, l}, q_{t+1, l}\right)$ and $\left(q_{t+1, l}, q_{t, l}\right)$ ) represent the different traveling paths of UAV since the flight directions of UAV on the two paths are different. After executing the above four steps of the PPSA algorithm, a detailed flight plan of UAV $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ in $\Omega\left(s_{i}\right)$ is obtained. The pseudo-code of the algorithm is shown in Algorithm 1.

Now, we analyze the performance of the PPSA algorithm. Suppose $\varphi\left(U_{s_{i}}^{*}, H P_{s_{i}}^{*}, t_{s_{i}}^{*}\right)$ is an optimal flight plan of UAV for the PPS problem, where $U_{s_{i}}^{*}$ denotes the optimal traveling path of UAV in $\Omega\left(s_{i}\right)$ and $H P_{s_{i}}^{*}$ represents the optimal hovering point of UAV with hovering time $t_{s_{i}}^{*}$ on $U_{s_{i}}^{*}$. Assume $U_{s_{i}}^{*}$ is composed of $U_{s_{i}}^{f *}$ and $U_{s_{i}}^{h *}$, where $U_{s_{i}}^{f *}$ and $U_{s_{i}}^{h *}$ are the horizontal flight path and vertical flight path of UAV in $\Omega\left(s_{i}\right)$, respectively. Let $q_{i}^{*}$ denote the

(a) $U_{s_{i}}^{f *}$ is a flight tour of UAV which consists of two paths $P_{b_{s_{i}}, q_{i}^{*}}$ and $P_{q_{i}^{*}, b_{s_{i}}}$

(b) $U_{s_{i}}^{f *}$ is a flight path of UAV that starts from $b_{s_{i}}$ and ends at another border point $e_{s_{i}}$ of $N\left(s_{i}^{\prime}\right)$

Fig. 5. Two oppositions to the optimal horizontal flight path of UAV.
projection point of $H P_{s_{i}}^{*}$ on $N\left(s_{i}^{\prime}\right)$ and $d_{H P_{s_{i}}^{*}, q_{i}^{*}}=h^{*}$. We use $E_{\varphi}^{i *}=L\left(U_{s_{i}}^{f *}\right) / v_{f}+L\left(U_{s_{i}}^{h *}\right) / v_{h}+t_{s_{i}}^{*}$ to represent the time cost of UAV by executing the flight plan $\varphi\left(U_{s_{i}}^{*}, H P_{s_{i}}^{*}, t_{s_{i}}^{*}\right)$ and $C_{s_{i}}^{*}$ to be the data transmission rate of UAV on $H P_{s_{i}}^{*}$.

Lemma 1: The path $U_{s_{i}}^{f *}$ must be a tour which is on the line segment $\left(b_{s_{i}}, s_{i}^{\prime}\right)$ from $b_{s_{i}}$ to $q_{i}^{*}$ and from $q_{i}^{*}$ to $b_{s_{i}}$.

Proof: We use the proof by contradiction. Suppose $U_{s_{i}}^{f *}$ is not a tour on the line segment $\left(b_{s_{i}}, s_{i}^{\prime}\right)$, then two cases should be considered, as shown in Fig. 5. One is that both the starting point and ending point of $U_{s_{i}}^{f *}$ are $b_{s_{i}}$ but either of $P_{b_{s_{i}}, q_{i}^{*}}$ and $P_{q_{i}^{*}, b_{s_{i}}}$ or neither of them is on the line $\left(b_{s_{i}}, s_{i}^{\prime}\right)$, as shown in Fig. 5(a). The other is that $U_{s_{i}}^{f *}$ starts from $b_{s_{i}}$ and ends at another border point $e_{s_{i}}$ of $N\left(s_{i}^{\prime}\right)$, as shown in Fig. 5(b).

In the first case, we construct a new flight plan $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ of UAV in $\Omega\left(s_{i}\right)$. Let $H P_{s_{i}}=H P_{s_{i}}^{*}$ be the hovering point of UAV. Let $U_{s_{i}}^{f}=\left(b_{s_{i}}, q_{i}^{*}\right) \cup\left(q_{i}^{*}, b_{s_{i}}\right)$ and $U_{s_{i}}^{h}=\left(q_{i}^{*}, H P_{s_{i}}\right) \cup\left(H P_{s_{i}}, q_{i}^{*}\right)$ be the horizontal flight path and vertical flight path of UAV, respectively, and $U_{s_{i}}=U_{s_{i}}^{f} \cup U_{s_{i}}^{h}$. Then, we can obtain $L\left(U_{s_{i}}^{f *}\right)>L\left(U_{s_{i}}^{f}\right)$ and $L\left(U_{s_{i}}^{h *}\right)=L\left(U_{s_{i}}^{h}\right)$. Assume that the amount of data collected by UAV during flying on $U_{s_{i}}^{h}$ is $V_{i}^{h}$ and that the average data transmission rate of UAV during traveling on $U_{s_{i}}^{f}$ is $\overline{C_{s_{i}}^{f}}$. Thus, we can obtain $t_{s_{i}}=\frac{V_{i}-V_{i}^{h}-\frac{L\left(U_{s_{i}}^{f}\right)}{v_{f}} \cdot \overline{C_{s_{i}}^{f}}}{C_{s_{i}}^{*}}$, and $E_{\varphi}^{i}=\frac{L\left(U_{s_{i}}^{f}\right)}{v_{f}}+\frac{L\left(U_{s_{i}}^{h}\right)}{v_{h}}+\frac{V_{i}-V_{i}^{h}-\frac{L\left(U_{s_{s}}^{f}\right)}{v_{f}} \cdot \overline{C_{s_{i}}^{f}}}{C_{s_{i}}^{*}}$. According to the shapes of the curves $U_{s_{i}}^{f}$ and $U_{s_{i}}^{f *}$, we can find that for each point $p \in U_{s_{i}}^{f}$, there exists a point $p^{\prime} \in U_{s_{i}}^{f *}$ such that $d_{s_{i}^{\prime}, p}=d_{s_{i}^{\prime}, p^{\prime}}$, which means that the data transmission rate of UAV on $p$ is the same as on $p^{\prime}$ since $d\left(s_{i}, p\right)=\sqrt{h^{2}+d_{s_{i}^{\prime}, p}^{2}}=\sqrt{h^{2}+d_{s_{i}^{\prime}, p^{\prime}}^{2}}=d\left(s_{i}, p^{\prime}\right)$. Therefore, we can obtain that there exists a part of $U_{s_{i}}^{f *}$ such that the amount of data collected by UAV during flying on the part is $\frac{L\left(U_{s_{i}}^{f}\right)}{v_{f}} \cdot \overline{C_{s_{i}}^{f}}$. Suppose the average data transmission of UAV during flying on the path $U_{s_{i}}^{f *} \backslash U_{s_{i}}^{f}$ is $\overline{C_{s_{i}}^{*}}$. Since $\overline{C_{s_{i}}^{*}}<C_{s_{i}}^{*}$, we can obtain

$$
\begin{aligned}
& E_{\varphi}^{i *} \\
& =\frac{V_{i}-V_{i}^{h}-\frac{L\left(U_{s_{i}}^{f}\right)}{v_{f}} \cdot \overline{C_{s_{i}}^{f}}-\frac{L\left(U_{s_{i}}^{f *}\right)-L\left(U_{s_{i}}^{f}\right)}{v_{f}} \cdot \overline{C_{s_{i}}^{*}}}{C_{s_{j}}^{*}}+\frac{L\left(U_{s_{i}}^{f *}\right)}{v_{f}}+\frac{L\left(U_{s_{i}}^{h *}\right)}{v_{h}} \\
& =E_{\varphi}^{i}+\frac{L\left(U_{s_{i}}^{f *}\right)-L\left(U_{s_{i}}^{f}\right)}{v_{f}}-\frac{L\left(U_{s_{i}}^{f *}\right)-L\left(U_{s_{i}}^{f}\right)}{v_{f}} \cdot \frac{C_{s_{i}}^{*}}{C_{s_{i}}^{*}}>E_{\varphi}^{i},
\end{aligned}
$$

which is contradiction to the assumption that $E_{\varphi}^{i *}$ is an optimal solution.
In the second case, we construct another new flight plan $\varphi\left(U_{s_{i}}, H P_{s_{i}}, t_{s_{i}}\right)$ of UAV. We first connect $q_{i}^{*}$ with $s_{i}^{\prime}$ to obtain the line segment $\left(q_{i}^{*}, s_{i}^{\prime}\right)$. Then we select a point $q_{i}^{\prime}$ on the line segment $\left(b_{s_{i}}, s_{i}^{\prime}\right)$ such that $d_{q_{i}^{\prime}, s_{i}^{\prime}}=d_{q_{i}^{*}, s_{i}^{\prime}}$, as shown in Fig. 5(b). Let $q_{i}^{\prime \prime}$ represent the projection point of $q_{i}^{\prime}$ on $N\left(s_{i}^{\prime \prime}\right)$. We use $p_{i}^{\prime} \in\left(q_{i}^{\prime}, q_{i}^{\prime \prime}\right)$ to be the hovering point $H P_{s_{i}}$ of UAV in $\Omega\left(s_{i}\right)$, where $d_{q_{i}^{\prime}, p_{i}^{\prime}}=h^{*}$. We can obtain that the data transmission rate of UAV at $H P_{s_{i}}$ in the same as $H P_{s_{i}}^{*}$ since $d_{H P_{s_{i}}, s_{i}}=$ $\sqrt{\left(h-h^{*}\right)^{2}+\left(r-d_{q_{i}^{\prime}, s_{i}^{\prime}}\right)^{2}}=d_{H P_{s_{i}}^{*}, s_{i}}$. Afterwards, we let $U_{s_{i}}^{f}=$ $\left(b_{s_{i}}, q_{i}^{\prime}\right) \cup\left(q_{i}^{\prime}, b_{s_{i}}\right)$ and $U_{s_{i}}^{h}=\left(q_{i}^{\prime}, H P_{s_{i}}\right) \cup\left(H P_{s_{i}}, q_{i}^{\prime}\right)$ be the horizontal flight path and vertical flight path of UAV, respectively. Therefore, we have $L\left(U_{s_{i}}^{f *}\right)>L\left(U_{s_{i}}^{f}\right)$ and $L\left(U_{s_{i}}^{h *}\right)=L\left(U_{s_{i}}^{h}\right)$. According to the shapes of the curves $U_{s_{i}}^{f}$ and $U_{s_{i}}^{f *}$, we can obtain that for each point $p \in\left(b_{s_{i}}, q_{i}^{\prime}\right)$ (or $p \in\left(q_{i}^{\prime}, b_{s_{i}}\right)$ ), there exists a point $p^{\prime} \in P_{b_{s_{i}}, q_{i}^{*}}$ (or $p^{\prime} \in P_{q_{i}^{*}, e_{s_{i}}}$ ) such that $d_{s_{i}^{\prime}, p}=d_{s_{i}^{\prime}, p^{\prime}}$, which means that the data transmission rate of UAV on $p$ is the same as on $p^{\prime}$. The following proof is similar to the first case. Therefore, we can obtain $E_{\varphi}^{i}<E_{\varphi}^{i *}$, which is also contradiction to the assumption that $E_{\varphi}^{i *}$ is an optimal solution.

From above discussion, we can obtain that $U_{s_{i}}^{f *}$ is a tour on the line segment $\left(b_{s_{i}}, s_{i}^{\prime}\right)$, and both the starting point and ending point of $U_{s_{i}}^{f *}$ are $b_{s_{i}}$.

Theorem 2: We have $E_{\varphi}^{i} \leq E_{\varphi}^{i *}+\frac{r}{n \cdot v_{f}}$, where $E_{\varphi}^{i}$ is obtained by PPSA.

Proof: According to definition of the PPS problem and Lemma 1, we should consider three cases as follows:
(1) $H P_{s_{i}}^{*} \in\left(b_{s_{i}}, s_{i}^{\prime}\right)$;
(2) There exists a $0 \leq t \leq 6 n$ such that $H P_{s_{i}}^{*} \in\left(q_{t, 0}, q_{t, \tau}\right) \backslash$ $\left\{q_{t, 0}\right\}$;
(3) There exists a $0 \leq t \leq 6 n$ such that $q_{i}^{*} \in\left(q_{t, 0}, q_{t+1,0}\right) \backslash$ $\left\{q_{t, 0}, q_{t+1,0}\right\}$ and $H P_{s_{i}}^{*} \notin\left(b_{s_{i}}, s_{i}^{\prime}\right)$.
In the following, we will give the relationship between the time consumption of the optimal flight plan and the time consumption obtained by the PPSA algorithm for each of cases, and then obtain the performance ratio of the algorithm. For simplicity, we use $V_{U}^{*}$ to represent the amount of transmission data from $s_{i}$ to UAV when it flies on $U_{s_{i}}^{*}$.

In the first case, based on the Lemma 1, we know that there exists a $0 \leq t<6 n$ which satisfies

$$
\begin{equation*}
r-(t+1) \cdot \delta \leq d_{H P_{s_{i}}^{*}, s_{i}^{\prime}} \leq r-t \cdot \delta, \tag{15}
\end{equation*}
$$

as shown in Fig.6. then we can obtain

$$
\begin{align*}
E_{\varphi}^{i *} & =\frac{2\left(r-d_{\left.H P_{s_{i}^{*}}^{*}, s_{i}^{\prime}\right)}+\frac{2\left(V_{i}-V_{U}^{*}\right)}{v_{f}}+\log _{2}\left(1+\frac{\gamma_{0}}{\left(d_{\left.H P_{s_{i}, s_{i}^{\prime}}^{2}+h^{2}\right)^{\frac{\alpha}{2}}}^{2}\right.}\right)\right.}{} \\
& \geq 2 t \cdot \frac{\delta}{v_{f}}+\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)} . \tag{16}
\end{align*}
$$

By taking $t+1$, let $V_{U}=\sum_{j=0}^{t} \frac{\delta}{v_{f}} W \log \left(1+\frac{\gamma_{0}}{\left(\left(r-j \cdot \delta \delta^{2}+h^{2}\right)^{\frac{\alpha}{2}}\right.}\right)$, which is the amount of data transmitted from $s_{i}$ to UAV when it flies on the path $\left(b_{s_{i}}, q_{t+1,0}\right) \cup\left(q_{t+1,0}, b_{s_{i}}\right)$ obtained by PPSA.


Fig. 6. An example $t$ such that $r-(t+1) \cdot \delta<d_{H P_{S_{i}, ~}, s_{i}^{\prime}} \leq r-t \cdot \delta$.

Then, we have

$$
\begin{equation*}
V_{U}^{*} \leq V_{U}+\frac{\delta}{v_{f}} W \log \left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right) . \tag{17}
\end{equation*}
$$

Based on inequations (15)-(17) and algorithm PPSA, we can obtain

$$
\begin{aligned}
E_{\varphi}^{i} \leq & (t+1) \cdot \frac{2 \delta}{v_{f}}+\frac{2\left(V_{i}-V_{U}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
= & (2 t+4) \cdot \frac{\delta}{v_{f}} \\
& +\frac{2\left(V_{i}-V_{U}-\frac{\delta}{v_{f}} W \log \left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
\leq & (2 t+4) \cdot \frac{\delta}{v_{f}}+\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
\leq & E_{\varphi}^{i *}+\frac{4 \delta}{v_{f}}<E_{\varphi}^{i *}+\frac{r}{n \cdot v_{f}} .
\end{aligned}
$$

In the second case, we can obtain that there exists a $1 \leq l<\tau$ which satisfies

$$
\begin{equation*}
l \cdot \eta \leq d_{H P_{s_{i}}^{*}, q_{i}^{*}} \leq(l+1) \cdot \eta \tag{18}
\end{equation*}
$$

By taking $t$, according to the inequation (18), we can obtain

$$
\begin{align*}
E_{\varphi}^{i *} & =\frac{2 d_{b_{s_{i}}, q_{i}^{*}}}{v_{f}}+\frac{2 d_{q_{i}^{*}, H P_{s_{i}}^{*}}}{v_{h}}+\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{d_{H p_{i}}^{\alpha}, s_{i}}\right)} \\
& \geq 2 t \cdot \frac{\delta}{v_{f}}+2 l \cdot \frac{\eta}{v_{h}}+\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \delta)^{2}+(h-(l+1) \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)} . \tag{19}
\end{align*}
$$

By taking $l+1$, let $V_{U}=\sum_{k=0}^{t} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}+\sum_{j=1}^{l} \frac{2 \eta}{v_{h}} \cdot C_{q_{t, j}}$ be the amount of data transmitted from $s_{i}$ to UAV when it flies on the path $\left(b_{s_{i}}, q_{t, 0}\right) \cup\left(q_{t, 0}, q_{t, l+1}\right) \cup\left(q_{t, l+1}, q_{t, 0}\right) \cup\left(q_{t, 0}, b_{s_{i}}\right)$ obtained by PPSA, where $C_{q_{k, 0}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-k \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)$ for any $0 \leq k \leq t$ and $C_{q_{t, j}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+(h-j \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)$ for any $1 \leq j \leq l$. We can find that

$$
\begin{equation*}
V_{U}^{*} \leq V_{U}+\frac{\eta}{v_{h}} W \log \left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+(h-(l+1) \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right) . \tag{20}
\end{equation*}
$$

Since $\frac{\delta}{v_{f}}=\frac{\eta}{v_{h}}$, based on the algorithm PPSA and inequations (18)-(20), we can obtain

$$
\begin{aligned}
E_{\varphi}^{i} \leq & 2 t \cdot \frac{\delta}{v_{f}}+2(l+1) \cdot \frac{\eta}{v_{h}} \\
& +\frac{2\left(V_{i}-V_{U}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+(h-(l+1) \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
\leq & 2 t \cdot \frac{\delta}{v_{f}}+(2 l+4) \cdot \frac{\eta}{v_{h}} \\
& +\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-t \cdot \delta)^{2}+(h-(l+1) \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
\leq & E_{\varphi}^{i *}+\frac{4 \eta}{v_{h}}<E_{\varphi}^{i *}+\frac{r}{n \cdot v_{f}} .
\end{aligned}
$$

In the third case, we can obtain

$$
\begin{equation*}
r-(t+1) \cdot \delta<d_{q_{i}^{*}, s_{i}^{\prime}}<r-t \cdot \delta \tag{21}
\end{equation*}
$$

Suppose $d_{q_{t, 0}, q_{i}^{*}}=\varepsilon \cdot \delta$, where $0<\varepsilon<1$. Let $q_{i}^{\prime}$ be the projection point of $q_{i}^{*}$ on $N\left(s_{i}^{\prime \prime}\right)$. Then we divide $\left(q_{i}^{*}, q_{i}^{\prime}\right)$ into $\tau$ parts, and $\left(q_{t+\varepsilon, 0}, q_{t+\varepsilon, 1}, \cdots, q_{t+\varepsilon, \tau}\right)=\left(q_{i}^{*}, q_{i}^{\prime}\right)$, where $d_{q_{t+\varepsilon, l,}, q_{t+\varepsilon, l+1}}=\eta$ for any $0 \leq l<\tau-1$ and $d_{q_{t+\varepsilon, l}, q_{t+\varepsilon, l+1}}=h-$ $h_{0}-(\tau-1) \cdot \eta$ for $l=\tau-1$. Then, based on inequations (18) and (21), we can obtain

$$
\begin{align*}
& E_{\varphi}^{i *}= \frac{2 d_{b_{s_{i}}, q_{i}^{*}}}{v_{f}}+\frac{2 d_{q_{i}^{*}, H P_{s_{i}}^{*}}^{v_{h}}+\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{d_{H P_{s}, s_{i}}^{\alpha}}\right)}}{\geq} \begin{aligned}
v_{f} & \frac{2(t+\varepsilon) \cdot \delta}{v_{h}} \\
& +\frac{2 l \cdot \eta}{W \log _{2}\left(1+\frac{2\left(V_{i}-V_{U}^{*}\right)}{\left((r-(t+\varepsilon) \delta)^{2}+(h-(l+1) \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
\geq & 2 t \cdot \frac{\delta}{v_{f}}+\frac{2 l \cdot \eta}{v_{h}} \\
& +\frac{2\left(V_{i}-V_{U}^{*}\right)}{W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \delta)^{2}+(h-(l+1) \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)}
\end{aligned} . \\
& \\
& \\
&
\end{align*}
$$

By taking $t+1$ and $l+1$, let $V_{U}=\sum_{k=0}^{t} \frac{2 \delta}{v_{f}} \cdot C_{q_{k, 0}}+\sum_{j=0}^{l} \frac{2 \eta}{v_{h}}$. $C_{q_{t+1, j}}$ be the amount of data transmitted from $s_{i}$ to UAV when it flies on the path $\left(b_{s_{i}}, q_{t+1,0}\right) \cup\left(q_{t+1,0}, q_{t+1, l+1}\right) \cup$ $\left(q_{t+1, l+1}, q_{t+1,0}\right) \cup\left(q_{t+1,0}, b_{s_{i}}\right)$ obtained by PPSA, where $C_{q_{k, 0}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-k \cdot \delta)^{2}+h^{2}\right)^{\frac{\alpha}{2}}}\right)$ for any $0 \leq k \leq t$ and $C_{q_{t+1, j}}=\frac{1}{2} W \log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+(h-j \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)$ for any $0 \leq j \leq$ $l$. We have

$$
\begin{equation*}
V_{U}^{*} \leq V_{U}+\frac{\eta}{v_{h}} W \log \left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+(h-(l+1) \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right) . \tag{23}
\end{equation*}
$$

Based on inequations (21)-(23), we can obtain

$$
\begin{aligned}
E_{\varphi}^{i} \leq & \frac{2(t+1) \delta}{v_{f}}+\frac{2(l+1) \eta}{v_{h}} \\
& +\frac{2\left(V_{i}-V_{U}\right) \cdot \frac{1}{W}}{\log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+(h-(l+1) \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)}
\end{aligned}
$$

$$
\begin{aligned}
\leq & \frac{(2 t+2) \delta}{v_{f}}+\frac{(2 l+4) \eta}{v_{h}} \\
& +\frac{2\left(V_{i}-V_{U}^{*}\right) \cdot \frac{1}{W}}{\log _{2}\left(1+\frac{\gamma_{0}}{\left((r-(t+1) \cdot \delta)^{2}+(h-(l+1) \cdot \eta)^{2}\right)^{\frac{\alpha}{2}}}\right)} \\
\leq & E_{\varphi}^{i *}+\frac{2 \delta}{v_{f}}+\frac{4 \eta}{v_{h}} \leq E_{\varphi}^{i *}+\frac{r}{n \cdot v_{f}} .
\end{aligned}
$$

From what has been discussed, we have $E_{\varphi}^{i} \leq E_{\varphi}^{i *}+\frac{r}{n \cdot v_{f}}$.

## B. Algorithm for the FTPS Problem

In this subsection, we propose an approximation algorithm for solving the FTPS problem, which is called FTPSA. The objective of the algorithm is to find a flight plan $\Phi(U, H, T)$ of single UAV and

$$
\text { Minimize } \quad E_{\Phi}=\frac{L\left(U_{f}\right)}{v_{f}}+\frac{L\left(U_{h}\right)}{v_{h}}+t_{s_{i}} \text {, }
$$

where $U_{f}$ and $U_{h}$ represent the total horizontal flight path and total vertical flight path of UAV, respectively, and $U=U_{f} \cup U_{h}$.
The FTPSA consists of four steps. In the first step, we employ the $(1+\varepsilon)$-approximation algorithm for the TSPN problem proposed in [23] to compute a tour $U_{f}^{\prime}$ for $D$, and obtain the order of the data collection areas in $\Theta$ visited by $U_{f}^{\prime}$, which is denoted as $\Omega\left(s_{\rho_{1}}\right), \Omega\left(s_{\rho_{2}}\right), \cdots, \Omega\left(s_{\rho_{n}}\right)$, where $\Omega\left(s_{\rho_{i}}\right)$ is the $i$-th data collection area visited by $U_{f}^{\prime}$. In the second step, for each $\Omega\left(s_{\rho_{i}}\right) \in \Theta$, we compute the first intersection point $b_{s_{\rho_{i}}}$ between $U_{f}^{\prime}$ and $N\left(s_{\rho_{i}}^{\prime}\right)$. Let $b_{s_{\rho_{i}}}$ be the entry point of UAV to visit the data collection area $\Omega\left(s_{\rho_{i}}\right)$. We compute the flight plan $\varphi\left(U_{s_{\rho_{i}}}, H P_{S_{\rho_{i}}}, t_{S_{\rho_{i}}}\right)$ and the time cost $E_{\varphi}^{\rho_{i}}$ for UAV in $\Omega\left(s_{\rho_{i}}\right)$ by executing the PPSA algorithm, where $U_{s_{\rho_{i}}}$ is comprised of the horizontal flight path $U_{s_{p_{i}}}^{f}$ and vertical flight path $U_{s_{\rho_{i}}}^{h}$ of UAV. Then we compute the projection point $q_{\rho_{i}}$ of $H P_{s_{\rho_{i}}}$ on $N\left(s_{\rho_{i}}^{\prime}\right)$. Afterwards, we perform the operations $U_{h}=U_{h} \cup U_{s_{\rho_{i}}}^{h}, H=H \cup\left\{H P_{s_{\rho_{i}}}\right\}$ and $T=T \cup\left\{t_{s_{p_{i}}}\right\}$. In the third step, for any $1 \leq i \leq n$, we use the line segment $\left(q_{\rho_{i}}, q_{\rho_{i+1}}\right)$ to be the horizontal flight path of UAV which is from $q_{\rho_{i}}$ to $q_{\rho_{i+1}}$ (where $\left.s_{\rho_{n+1}}=s_{\rho_{1}}\right)$, and $U_{f}=U_{f} \cup\left(q_{\rho_{i}}, q_{\rho_{i+1}}\right)$. Finally, the complete flight tour of UAV $U=U_{f} \cup U_{h}$ is derived, and then the flight plan $\Phi=\{U, H, T\}$ and the total time cost $E_{\Phi}=L\left(U_{f}\right) / v_{f}+L\left(U_{h}\right) / v_{h}+\sum_{t_{s_{p_{i}}} \in T} t_{s_{\rho_{i}}}$ of UAV are obtained. The pseudo-code of the algorithm is shown in Algorithm 2.
Suppose $\Phi\left(U^{*}, H^{*}, T^{*}\right)$ is an optimal flight plan of the UAV for the FTPS problem, where $U^{*}$ is an optimal traveling tour of UAV, which consists of the optimal horizontal flight path $U_{f}^{*}$ and vertical flight path $U_{h}^{*}, H^{*}$ denotes an optimal set of hovering points of UAV on $U^{*}$ in which for each $H P_{s_{i}}^{*} \in H^{*}$, there is a corresponding hovering time $t_{s_{i}}^{*} \in T^{*}$. We use $E_{\Phi}^{*}$ to denote the time consumption of UAV for the flight plan $\Phi\left(U^{*}, H^{*}, T^{*}\right)$.

Theorem 3: We have $E_{\Phi} \leq(2+\varepsilon) \cdot E_{\Phi}^{*}+\frac{r}{v_{f}}$, where $E_{\Phi}$ is obtained by using the FTPSA algorithm.

Proof: Suppose $U_{t p}^{*}$ is an optimal tour for the TSPN problem. Since $U_{f}^{*}$ should visit all disks in $D$, we can obtain

```
Algorithm 2 FTPSA
    Input: \(S=\left\{s_{1}, s_{2}, \cdots, s_{n}\right\}, V_{i}\) for each \(s_{i} \in S, R, r\),
            \(D=\left\{N\left(s_{1}^{\prime}\right), N\left(s_{2}^{\prime}\right), \cdots, N\left(s_{n}^{\prime}\right)\right\}, W, \gamma_{0}, h_{0}\),
            \(\Theta=\left\{\Omega\left(s_{1}\right), \Omega\left(s_{2}\right), \cdots, \Omega\left(s_{n}\right)\right\}, h, v_{f}, v_{h}, s_{0}\)
```

    Output: A flight plan \(\Phi(U, H, T)\) of UAV and \(E_{\Phi}\)
    1 Using the \((1+\varepsilon)\)-approximation algorithm for the TSPN
    problem to compute a tour \(U_{f}^{\prime}\) for \(D\) [23], and the order
    of data collection areas in \(\Theta\) visited by \(U_{f}^{\prime}\), which is
    denoted as \(\Omega\left(s_{\rho_{1}}\right), \Omega\left(s_{\rho_{2}}\right), \cdots, \Omega\left(s_{\rho_{n}}\right)\);
    for each \(\Omega\left(s_{\rho_{i}}\right) \in \Theta\) do
        Compute the first intersection point \(b_{s_{\rho_{i}}}\) between \(U_{f}^{\prime}\)
        and \(N\left(s_{\rho_{i}}^{\prime}\right)\), where \(U_{f}^{\prime}\) is visited in counter clockwise
        order;
        Compute the flight plan \(\varphi\left(U_{s_{\rho_{i}}}, H P_{s_{\rho_{i}}}, t_{s_{\rho_{i}}}\right)\) and the
        time cost \(E_{\varphi}^{\rho_{i}}\) for UAV in the area \(\Omega\left(s_{\rho_{i}}\right)\) by executing
        the PPSA algorithm, where \(U_{s_{\rho_{i}}}=U_{s_{\rho_{i}}}^{f} \cup U_{s_{p_{i}}}^{h}\), and
        compute the projection point \(q_{\rho_{i}}\) of \(H P_{s_{\rho_{i}}}\) on \(N\left(s_{i}^{\prime}\right)\);
        \(U_{h}=U_{h} \cup U_{s_{p_{i}}}^{h}, H=H \cup H P_{s_{\rho_{i}}}, T=T \cup t_{s_{\rho_{i}}} ;\)
    end
    for \(i\) from 0 to \(n\) do
        Let \(\left(q_{\rho_{i}}, q_{\rho_{i+1}}\right)\) be the horizontal flight path of UAV
        which is from \(q_{\rho_{i}}\) to \(q_{\rho_{i+1}}\), and \(U_{f}=U_{f} \cup\left(q_{\rho_{i}}, q_{\rho_{i+1}}\right)\);
    9 end
    \({ }^{0} U=U_{f} \cup U_{h}, E_{\Phi}=L\left(U_{f}\right) / v_{f}+L\left(U_{h}\right) / v_{h}+\sum_{t_{s_{\rho_{i}}} \in T} t_{s_{\rho_{i}}}\);
    that $U_{f}^{*}$ is a feasible solution for the TSPN problem. Thus, we have

$$
\begin{equation*}
L\left(U_{t p}^{*}\right) \leq L\left(U_{f}^{*}\right) \tag{24}
\end{equation*}
$$

According to the algorithm for the TSPN problem, we can derive

$$
\begin{equation*}
L\left(U_{f}^{\prime}\right) \leq(1+\varepsilon) \cdot L\left(U_{t p}^{*}\right) \tag{25}
\end{equation*}
$$

Based on the definition of FTPS problem, we can obtain

$$
\begin{equation*}
E_{\Phi}^{*} \geq \frac{L\left(U_{f}^{*}\right)}{v_{f}}+\frac{L\left(U_{h}^{*}\right)}{v_{h}} \tag{26}
\end{equation*}
$$

and

$$
\begin{equation*}
E_{\Phi}^{*} \geq \sum_{i=1}^{n} E_{\varphi}^{\rho_{i}{ }^{*}} \tag{27}
\end{equation*}
$$

Since the shortest distance between any pair of points is the straight line for connecting them without any curves, we can get that for any $1 \leq i \leq n$,

$$
\begin{equation*}
d_{q_{\rho_{i}}, q_{P_{i+1}}} \leq \frac{1}{2} \cdot L\left(U_{s_{\rho_{i}}}^{f}\right)+\frac{1}{2} \cdot L\left(U_{s_{\rho_{i+1}}}^{f}\right)+L\left(P_{b_{s_{i}},}, b_{s_{p_{i+1}}}\right) . \tag{28}
\end{equation*}
$$

Based on the Theorem 2 and inequations (24)-(28), we can obtain

$$
\begin{aligned}
E_{\Phi} & =\frac{L\left(U_{f}\right)}{v_{f}}+\frac{L\left(U_{h}\right)}{v_{h}}+\sum_{t_{s \rho_{i}} \in T} t_{s_{\rho_{i}}} \\
& =\frac{\sum_{i=1}^{n} d_{q_{\rho_{i}}, q_{\rho_{i+1}}}}{v_{f}}+\frac{L\left(U_{h}\right)}{v_{h}}+\sum_{t_{s_{\rho_{i}}} \in T} t_{s_{\rho_{i}}}
\end{aligned}
$$

$$
\begin{aligned}
& \leq \frac{L\left(U_{f}^{\prime}\right)}{v_{f}}+\sum_{s_{\rho_{i}} \in S}\left(\frac{L\left(U_{s \rho_{i}}^{f}\right)}{v_{f}}+\frac{L\left(U_{s \rho_{i}}^{h}\right)}{v_{h}}+t_{s_{\rho_{i}}}\right) \\
& =\frac{L\left(U_{f}^{\prime}\right)}{v_{f}}+\sum_{i=1}^{n} E_{\varphi}^{\rho_{i}} \leq \frac{(1+\varepsilon) \cdot L\left(U_{f}^{*}\right)}{v_{f}}+\sum_{i=1}^{n}\left(E_{\varphi}^{\rho_{i} *}+\frac{r}{n \cdot v_{f}}\right) \\
& \leq(2+\varepsilon) \cdot E_{\Phi}^{*}+\frac{r}{v_{f}} .
\end{aligned}
$$

Hence, the theorem has been proved.

## V. Algorithm for the FTP Problem

In this section, we propose an approximation algorithm, called FTPM, to solve the general FTP problem. The objective of the FTPM algorithm aims at finding a flight plan $\Phi(U, H, T)$ of $m$ UAVs and

$$
\text { Minimize } \quad E_{\Phi}=\max _{1 \leq k \leq m} E_{\phi}^{k}
$$

where $E_{\phi}^{k}$ is the time cost of $f_{k}$.
The FTPM algorithm consists of five steps as follows.
In the first step, we compute the flight plan $\Phi\left(U_{t}, H_{t}, T_{t}\right)$ of the single UAV on $\Theta$ by executing the FTPSA algorithm, where $U_{t}$ consists of the horizontal flight path $U_{t}^{f}$ and vertical flight path $U_{t}^{h}$.

In the second step, for each $s_{i} \in S$, we create the virtual paths $P_{s_{i}^{b}, s_{i}^{e}}$ and $P_{s_{i}^{1}, s_{i}^{2}}$ to represent the horizontal flight path and vertical flight path of UAV with $t_{s_{i}}$ and $\frac{d_{q_{i}, H P_{s_{i}}}}{v_{h}}$ flying time, respectively, i.e., $L\left(P_{s_{i}^{b}, s_{i}}\right)=v_{f} \cdot t_{s_{i}}$ and $L\left(P_{s_{i}^{1}, s_{i}^{2}}\right)=\frac{d_{q_{i}} H P_{s_{i}}}{v_{h}} \cdot v_{f}$, where $s_{i}^{b}$ and $s_{i}^{e}$ are respectively the starting point and ending point of $P_{s_{i}^{b}, s_{i}^{e}}$, and $s_{i}^{1}$ and $s_{i}^{2}$ respectively denote the starting point and ending point of $P_{s_{i}^{1}, s_{i}}$. Then the paths $P_{s_{i}^{b}, s_{i}^{e}}, P_{s_{i}^{1}, s_{i}^{2}}$ and $P_{s_{i}^{2}, s_{i}^{1}}$ are added into $Q$, where $Q$ is a set of virtual paths.

In the third step, we combine $Q$ and $U_{t}^{f}$, and put the result into $U_{f}$, i.e., $U_{f}=U_{t}^{f} \cup Q$. Afterwards, we divide $U_{f}$ into $m$ paths $P_{1}, P_{2}, \cdots, P_{m}$ based on their counter-clockwise visiting sequence such that $L\left(P_{k}\right)=\frac{L\left(U_{f}\right)}{m}$ for any $1 \leq k \leq m$. Let $c_{k}$ denote the connection point between $P_{k}$ and $P_{k+1}$, where $1 \leq$ $k \leq m-1$. For simplicity, we use $c_{k}^{b}$ and $c_{k}^{e}$ to represent the starting point and ending point of $P_{k}$, respectively. Initially, we have $c_{1}^{b}=c_{m}^{e}=s_{0}, c_{k}^{e}=c_{k}$ for any $1 \leq k \leq m-1$ and $c_{k}^{b}=c_{k-1}$ for any $2 \leq k \leq m$.

In the fourth step, for each $s_{i} \in S$ that is visited by the path $P_{k}$, we design the detailed flight plan of $f_{k}$ in $\Omega\left(s_{i}\right)$ on the following two cases: $P_{s_{i}^{b}, s_{i}^{e}} \subset P_{k}$ and $c_{k} \in P_{s_{i}^{b}, s_{i}^{e}}$. In the former case, we let $H P_{s_{i}} \in H_{t}$ with the hovering time $L\left(P_{s_{i}^{b}, s_{i}^{e}}\right) / v_{f}$ be the hovering point of $f_{k}$, and set $H_{k}=H_{k} \cup\left\{H P_{s_{i}}\right\}, T_{k}=$ $T_{k} \cup\left\{L\left(P_{s_{i}^{b}, s_{i}^{e}}\right) / v_{f}\right\}$ and $U_{k}^{h}=U_{k}^{h} \cup P_{s_{i}^{1}, s_{i}^{2}} \cup P_{s_{i}^{2}, s_{i}^{1}}$, where $U_{k}^{h}$ represents the vertical flight path of $f_{k}$. If $\Omega\left(s_{i}\right)$ is the last data collection area visited by $P_{k}$, then the paths $P_{s_{i}^{b}, s_{i}^{e}}$ and $P_{s_{i}^{e}, c_{k}}$ are deleted from $P_{k}$, the path $P_{s_{i}^{1}, s_{i}^{2}}$ is added into $P_{k}^{i}$, and the end point $c_{k}^{e}$ of $P_{k}$ is changed to $s_{i}^{1}$. Otherwise, if $\Omega\left(s_{i}\right)$ is the first data collection area visited by $P_{k}$, then the paths $P_{s_{i}, s_{i}^{e}}$ and $P_{c_{k-1}, s_{i}^{b}}$ are deleted from $P_{k}$, the path $P_{s_{i}^{2}, s_{i}^{1}}$ is added into $P_{k}^{i}$, and the start point $c_{k}^{b}$ of $P_{k}$ is updated to $s_{i}^{1}$. In the latter case, we let $H P_{s_{i}}$ with the hovering time $L\left(P_{s_{i}^{b}, c_{k}}\right) / v_{f}$ be the hovering point of $f_{k}$, and set $H_{k}=H_{k} \cup\left\{H P_{s_{i}}\right\}, T_{k}=T_{k} \cup\left\{L\left(P_{s_{i}^{b}, c_{k}}\right) / v_{f}\right\}$.

```
Algorithm 3 FTPM
    Input: \(S=\left\{s_{1}, s_{2}, \cdots, s_{n}\right\}, V_{i}\) for each \(s_{i} \in S, r, R\),
            \(F=\left\{f_{1}, f_{2}, \cdots, f_{m}\right\}, h, v_{f}, v_{h}, h_{0}, W, \gamma_{0}\),
            \(\Theta=\left\{\Omega\left(s_{1}\right), \Omega\left(s_{2}\right), \cdots, \Omega\left(s_{n}\right)\right\}\),
            \(D=\left\{N\left(s_{1}^{\prime}\right), N\left(s_{2}^{\prime}\right), \cdots, N\left(s_{n}^{\prime}\right)\right\}\)
    Output: A flight plan \(\Phi(U, H, T)\) for \(m\) UAVs and \(E_{\Phi}\)
    1 Compute the flight plan \(\Phi\left(U_{t}, H_{t}, T_{t}\right)\) for the single UAV
    on \(\Theta\) by executing Algorithm FTPSA;
    for each \(s_{i} \in S\) do
        \(P_{s_{i}^{b}, s_{i}^{e}}=t_{s_{i}} \cdot v_{f}, P_{s_{i}^{1}, s_{i}^{2}}=\frac{d_{q_{i}, H P_{s_{i}}}}{v_{h}} \cdot v_{f} ;\)
        \(Q=Q \cup P_{s_{i}^{b}, s_{i}^{e}} \cup P_{s_{i}^{1}, s_{i}} \cup P_{s_{i}^{2}, s_{i}^{1}} ;\)
    end
    \({ }^{6} U_{f}=U_{t}^{f} \cup Q\), and divide \(U_{f}\) into \(m\) paths \(P_{1}, P_{2}, \cdots, P_{m}\)
    such that \(L\left(P_{k}\right)=\frac{L\left(U_{f}\right)}{m}\), and let \(c_{k}\) denote the connection
    point between \(P_{k}\) and \(P_{k+1}\);
    for \(k\) from 1 to \(m\) do
        for each \(s_{i} \in S\) do
            if \(P_{s_{i}^{b}, s_{i}^{e}} \subset P_{k}\) then
                    \(\left.H_{k}=H_{k} \cup\left\{H P_{s_{i}}\right\}, T_{k}=T_{k} \cup\left\{\frac{L\left(P_{s_{i}^{b}, s_{i}}\right)}{v_{f}}\right)\right\}\),
                    \(U_{k}^{h}=U_{k}^{h} \cup P_{s_{i}^{1}, s_{i}} \cup P_{s_{i}^{2}, s_{i}^{1}}\);
                    if \(\Omega\left(s_{i}\right)\) is the last area visited by \(P_{k}\) then
                    \(P_{k}=\left(P_{k}-P_{s_{i}^{b}, s_{i}^{e}}-P_{s_{i}^{e}, c_{k}}\right) \cup P_{s_{i}^{2}, s_{i}^{1}}, c_{k}^{e}=s_{i}^{1} ;\)
                end
                if \(\Omega\left(s_{i}\right)\) is the first area visited by \(P_{k}\) then
                    \(\begin{aligned} P_{k} & =\left(P_{k}-P_{s_{i}^{b}, s_{i}^{e}}-P_{c_{k-1}, s_{i}^{b}}\right) \cup P_{s_{i}^{2}, s_{i}^{1}} ; \\ c^{b} & =s_{1}^{1} .\end{aligned}\)
                    \(c_{k}^{b}=s_{i}^{1} ;\)
                end
            else
                if \(c_{k} \in P_{s_{i}^{b}, s_{i}^{e}}\) then
                    \(\left.H_{k}=H_{k} \cup\left\{H P_{s_{i}}\right\}, T_{k}=T_{k} \cup\left\{\frac{L\left(P_{s_{i}, c_{k}}\right)}{v_{f}}\right)\right\} ;\)
\(\left.P_{k}=\left(P_{k}-P_{s_{i}^{b}}\right) \cup c_{k}\right) \cup P_{s_{i}^{2}, s_{i}^{1}}, c_{k}^{e}=s_{i}^{1} ;\)
\(s_{i}^{b}=c_{k}, U_{k}^{h}=U_{k}^{h} \cup P_{s_{i}, s_{i}^{2}} \cup P_{s_{i}^{2}, s_{i}^{1}} ;\)
                end
            end
        end
        \(U_{k}=P_{k} \cup\left(s_{0}, c_{k}^{b}\right) \cup\left(c_{k}^{e}, s_{0}\right), U_{k}^{f}=U_{k}-U_{k}^{h} ;\)
        \(E_{\phi}^{k}=\frac{L\left(U_{k}^{f}\right)}{v_{f}}+\frac{L\left(U_{k}^{h}\right)}{v_{h}}+\sum_{t_{s_{i}}^{k} \in T_{k}} t_{s_{i}}^{k}, U=U \cup U_{k} ;\)
    8 end
    \(\Phi=\{U, H, T\}, E_{\Phi}=\max _{1 \leq k \leq m} E_{\phi}^{k} ;\)
```

Afterwards, we delete the path $P_{s_{i}^{b}, c_{k}}$ from $P_{k}$ and add the path $P_{s_{i}^{2}, s_{i}^{1}}$ into $P_{k}$. Then the ending point $c_{k}^{e}$ of $P_{k}$ is set to $s_{i}^{1}$ and the starting point of path $P_{s_{i}^{b}, s_{i}^{e}}$ is changed to $c_{k}$. This is because when the amount of sensory data carried by $s_{i}$ is very large, we may need several UAVs to collect its data simultaneously. Afterwards, we add the paths $P_{s_{i}^{1}, s_{i}^{2}}$ and $P_{s_{i}^{2}, s_{i}^{1}}$ into the path $U_{h}^{k}$. After completing this step, for any $f_{k} \in F$, the flight plan $\phi\left(P_{k}, H_{k}, T_{k}\right)$ is obtained.

Finally, for any $f_{k} \in F$, we construct the flight tour of $f_{k}$ by combining $P_{k},\left(s_{0}, c_{k}^{b}\right)$ and $\left(c_{k}^{e}, s_{0}\right)$, i.e., $U_{k}=P_{k} \cup$ $\left(s_{0}, c_{k}^{b}\right) \cup\left(c_{k}^{e}, s_{0}\right)$, and its horizontal flight time is $U_{k}^{f}=U_{k}-U_{k}^{h}$.

Therefore, we can calculate the time cost of $f_{k}$ as $E_{\phi}^{k}=$ $L\left(U_{k}^{f}\right) / v_{f}+L\left(U_{k}^{h}\right) / v_{h}+\sum_{t_{s_{i}} \in T_{k}} t_{s_{i}}^{k}$. Consequently, the time cost $E_{\Phi}=\max _{1 \leq k \leq m} E_{\phi}^{k}$ is obtained. The pseudo-code of the FTPM algorithm is given in Algorithm 3.

Next, we will analyze the performance of the FTPM algorithm. Suppose $\Phi\left(U^{*}, H^{*}, T^{*}\right)$ is an optimal flight plan of $m$ UAVs for the FTP problem, where $U^{*}=\left\{U_{1}^{*}, U_{2}^{*}, \cdots, U_{m}^{*}\right\}$, $H^{*}=\left\{H_{1}^{*}, H_{2}^{*}, \cdots, H_{m}^{*}\right\}, T^{*}=\left\{T_{1}^{*}, T_{2}^{*}, \cdots, T_{m}^{*}\right\}$ and let $E_{\Phi}^{*}$ denote the time cost of the flight plan $\Phi\left(U^{*}, H^{*}, T^{*}\right)$. For any $1 \leq k \leq m$, let $U_{k}^{*}=U_{k}^{f *} \cup U_{k}^{h *}$, where $U_{k}^{f *}$ and $U_{k}^{h *}$ denote the optimal horizontal flight path and vertical flight path of $f_{k}$, respectively. Let $U_{f}^{*}=\left\{U_{1}^{f *}, U_{2}^{f *}, \cdots, U_{m}^{f *}\right\}$.

Theorem 4: We have $E_{\Phi} \leq(3+\varepsilon) \cdot E_{\Phi}^{*}+\frac{3 r}{v_{f}}$, where $E_{\Phi}$ is obtained by the FTPM algorithm.

Proof: Suppose $U_{c}^{f *}=\bigcup_{k=1}^{m} U_{k}^{f *}$ is the union of all tours in $U_{f}^{*}$. Since all tours in $U_{f}^{*}$ can jointly visit all disks in $D$ and converge on $s_{0}$, we can find that $U_{c}^{f *}$ is a feasible solution for the TSPN problem. Thus, we have $L\left(U_{c}^{f *}\right) \geq L\left(U_{t p}^{*}\right)$. Let $L\left(U_{f}^{*}\right)=\max \left\{L\left(U_{k}^{f *}\right) \mid U_{k}^{f *} \in U_{f}^{*}\right\}$. Then, we can obtain

$$
\begin{equation*}
L\left(U_{f}^{*}\right) \geq \frac{1}{m} \cdot L\left(U_{c}^{f *}\right) \geq \frac{1}{m} \cdot L\left(U_{t p}^{*}\right) \tag{29}
\end{equation*}
$$

According to the definition of FTP problem, we can get

$$
\begin{equation*}
E_{\Phi}^{*} \geq \frac{L\left(U_{f}^{*}\right)}{v_{f}} \tag{30}
\end{equation*}
$$

and

$$
\begin{equation*}
E_{\Phi}^{*} \geq \frac{1}{m} \cdot \sum_{s_{i} \in S} E_{\varphi}^{i *} \tag{31}
\end{equation*}
$$

Based on the FTPM algorithm, we can obtain for each $1 \leq$ $k \leq m$, both the starting point $c_{k}^{b}$ and ending point $c_{k}^{e}$ of $P_{k}$ are located in the disks in $D$. Suppose $c_{k}^{b}$ is located in $N\left(s_{i}^{\prime}\right)$ and $c_{k}^{e}$ is in the disk $N\left(s_{l}^{\prime}\right)$. Then we can obtain

$$
\begin{equation*}
L\left(s_{0}, c_{k}^{b}\right) \leq L\left(s_{0}, s_{i}^{\prime}\right)+r \tag{32}
\end{equation*}
$$

and

$$
\begin{equation*}
L\left(c_{k}^{e}, s_{0}\right) \leq L\left(s_{0}, s_{l}^{\prime}\right)+r \tag{33}
\end{equation*}
$$

Since for any data collection area $\Omega\left(s_{i}\right) \in \Theta$, it should be visited by one of UAVs in $F$ and the UAV should arrive at $\Omega\left(s_{i}\right)$ from $s_{0}$ and go back to $s_{0}$. Therefore, we have

$$
\begin{equation*}
E_{\Phi}^{*} \geq 2 \max _{\Omega\left(s_{i}\right) \in \Theta} \frac{L\left(s_{0}, s_{s}^{\prime}\right)-r}{v_{f}} \tag{34}
\end{equation*}
$$

Based on Theorem 2 and the inequations (29)-(34), for any $f_{k} \in F$ and $m \geq 2$, we can obtain

$$
\begin{aligned}
E_{\phi}^{k} & =\frac{L\left(U_{f}^{k}\right)}{v_{f}}+\frac{L\left(U_{h}^{k}\right)}{v_{h}}+\sum_{t_{s_{i}}^{k} \in T_{k}} t_{s_{i}}^{k} \\
& =\frac{L\left(P_{k}\right)}{v_{f}}+\frac{L\left(s_{0}, c_{k}^{b}\right)}{v_{f}}+\frac{L\left(c_{k}^{e}, s_{0}\right)}{v_{f}}+\sum_{t_{s_{i}} \in T_{k}} t_{s_{i}}^{k} \\
& \leq \frac{1}{m}\left(\frac{L\left(U_{t}^{f}\right)}{v_{f}}+\sum_{s_{i} \in S} E_{\varphi}^{i}\right)+\frac{L\left(s_{0}, c_{k}^{b}\right)}{v_{f}}+\frac{L\left(c_{k}^{e}, s_{0}\right)}{v_{f}} \\
& \leq \frac{1}{m}\left((1+\varepsilon) \cdot \frac{L\left(U_{t p}^{*}\right)}{v_{f}}+\sum_{s_{i} \in S}\left(E_{\varphi}^{i *}+\frac{r}{n \cdot v_{f}}\right)\right)+E_{\Phi}^{*}+\frac{2 r}{v_{f}}
\end{aligned}
$$

$$
\begin{aligned}
& \leq \frac{1}{m}(1+\varepsilon) \frac{L\left(U_{f}^{*}\right)}{v_{f}}+\frac{1}{m} \cdot \sum_{s_{i} \in S} E_{\varphi}^{i *}+\frac{1}{m} \cdot \frac{r}{v_{f}}+E_{\Phi}^{*}+\frac{2 r}{v_{f}} \\
& \leq(3+\varepsilon) \cdot E_{\Phi}^{*}+\frac{3 r}{v_{f}}
\end{aligned}
$$

Consequently, we have

$$
E_{\Phi}=\max _{1 \leq k \leq m} E_{\phi}^{k} \leq(3+\varepsilon) \cdot E_{\Phi}^{*}+\frac{3 r}{v_{f}}
$$

Hence, the theorem has been proved.
According to Theorem 4, we can obtain that the FTPM have the constant approximation ratio for the FTP problem. However, in the case of steep terrain, such as in a mountainous environment, since sensors are deployed in the different altitudes, the data collection areas of sensors are different in size when the UAVs fly horizontally with fixed altitude. Based on the proposed algorithm for the FTP problem, we can firstly design traveling paths of UAVs for visiting all data collection areas in various sizes. If UAVs need to fly above and outside the transmission ranges of sensors due to the complex terrain, we can use the the projection points of the sensors on the flight plane of UAVs to be the starting points for visiting their data collection areas. Then for each sensor in the network, we find the optimal traveling paths of the visited UAVs by changing vertical cruising height to obtain the best tradeoff between traveling cost and hovering consumption of UAVs.

## VI. Simulation Results

In this section, we evaluate the average performance of the approximation algorithm FTPM through simulations on several key performance metrics under different settings. We implement the code using MATLAB 2013. In the simulations, sensors are deployed in a $2000 \mathrm{~m} \times 2000 \mathrm{~m}$ detection area, the reference SNR at transmission distance 1 m is set to $\gamma_{0}=80 \mathrm{~dB}$ and the path loss exponent is set to $\alpha=3$. For each parameter setting, we create 100 instances, execute the simulations, and obtain the average results.

Given an FTP instance, we compute the lower bound of the time cost of any feasible solution for the FTP problem as follows: (a) A minimum spanning tree $T_{r}$ of $D$ is computed, and we let $U_{D}^{*}$ denote an optimal tour to visit all disks in $D$, where disks in $D$ are referred as points. (b) The time cost $E_{\varphi}^{i}$ of UAV for each $\Omega\left(s_{i}\right) \in \Theta$ is computed by algorithm PPSA. (c) The lower bound of the solution for the FTP problem is equal to $\frac{1}{m} \cdot\left(\frac{L\left(T_{r}\right)}{v_{f}}+\sum_{\Omega\left(s_{i}\right) \in \Theta} E_{\varphi}^{i}-\frac{r}{v_{f}}\right)$, since $E_{\Phi}^{*} \geq \frac{1}{m}$. $\left(\frac{L\left(U_{D}^{*}\right)}{v_{f}}+\sum_{\Omega\left(s_{i}\right) \in \Theta} E_{\varphi}^{i *}\right) \geq \frac{1}{m} \cdot\left(\frac{L\left(T_{r}\right)}{v_{f}}+\sum_{\Omega\left(s_{i}\right) \in \Theta} E_{\varphi}^{i}-\frac{r}{v_{f}}\right)$ based on Theorem 2. In the following, we use $E_{\max }=\max \left\{E_{\phi}^{k} \mid f_{k} \in F\right\}$ and $E_{\text {min }}=\min \left\{E_{\phi}^{k} \mid f_{k} \in F\right\}$ to denote the maximum time cost and the minimum time cost of $m$ UAVs obtained by the FTPM algorithm, respectively. Then we evaluate how the network configurations, such as the number of sensors $n$, the number of UAVs $m$, the Bandwidth $W$, the amount of data $V_{i}$ carried by each sensor $s_{i} \in S$, the data transmission range $R$, the horizontal flight speed $v_{f}$, the vertical flight speed $v_{h}$ and horizontal flight altitude $h$, impact on the performance of FTPM algorithm by comparing $E_{\max }, E_{\min }$ with Lower Bound.

In Fig. 7, we give the performance of FTPM when we set $R=100 \mathrm{~m}, h=60 \mathrm{~m}, h_{0}=10 \mathrm{~m}, W=2 \mathrm{MB} / \mathrm{s}, v_{f}=10 \mathrm{~m} / \mathrm{s}$,


Fig. 7. Simulations by varing $n$ from 20 to 80 under different $m$.
$v_{h}=2 \mathrm{~m} / \mathrm{s}$ and use the interval $[1,3]$ MB to pick a uniformly distributed random data size $V_{i}$ for each sensor $s_{i} \in S$, and vary $m$ to $1,3,5,7,9$ and $n$ from 20 to 80 increased by 5 . In Fig. 7(a), we compare the performance of FTPM against the lower bound in terms of the ratio of $E_{\max }$ to the Lower Bound. It is observed that the ratio becomes higher as $m$ grows, and that the performance gap is getting smaller and stabilized with an increase in the number of sensors, since the total time consumption for each UAV in data collection areas of sensors increases as $n$ grows and the time cost of the lower bound in each data collection area got by PPSA is infinitely close to the optimal solution for the PPS problem. We also find that the ratio of $E_{\max }$ to the Lower Bound is always less than 3 , which verifies the effectiveness of the FTPM algorithm, and that FTPM performs reasonably well in a larger network. Fig. 7(b) is to illustrate the impact of the number of sensors on the time cost of UAV. We can find that $E_{\max }$ increases with the increasing of the number of sensors since both the hovering time and traveling time of UAVs are increased as the number of sensors grows. We can also observe that the performance gap is becoming smaller with increasing $m$. This is because the traveling time of UAVs becomes the main part of the time cost of UAVs while the traveling distance of each UAV does not increase very much as $m$ grows. In Fig. 7(c), we give the comparison of $E_{\max }, E_{\text {min }}$ and Lower Bound with $m=5$. It shows that all three increase as the number of sensors increases, which can guarantee the ratios in Fig. 7(a). We can also observe that the performance gap between $E_{\max }$ and $E_{\text {min }}$ got by the algorithm FTPM is very small and stabilized, which can prove the validity of the algorithm. Fig. 7(d) illustrates the ratios between any pairs of $E_{\max }, E_{\min }$ and Lower Bound when $m=5$. We can find the ratios $E_{\max } /$ LowerBound, $E_{\min } /$ LowerBound and $E_{\max } / E_{\min }$ derease with the increasing of $n$ since both the hovering time and traveling time of UAVs are increased as $n$ grows.


Fig. 8. Simulations through changing $v_{h}$ from $1 \mathrm{~m} / \mathrm{s}$ to $7 \mathrm{~m} / \mathrm{s}$ under different $V_{i}$.


Fig. 9. Simulations through changing $v_{f}$ from $8 \mathrm{~m} / \mathrm{s}$ to $20 \mathrm{~m} / \mathrm{s}$ under different $R$.

Fig. 8 illustrates the performance of FTPM when we set $n=60, R=100 \mathrm{~m}, h=60 \mathrm{~m}, h_{0}=10 \mathrm{~m}, v_{f}=15 \mathrm{~m} / \mathrm{s}, m=5$ and randomly pick $V_{i}$ from the intervals [1,2], [2,3], [3,4], [4,5], and [5,6] MB, respectively and change $v_{h}$ from $1 \mathrm{~m} / \mathrm{s}$ to $7 \mathrm{~m} / \mathrm{s}$. Fig. 8(a) gives the changing trend of the ratio of $E_{\max }$ to the lower bound as $v_{h}$ grows. It is observed that the ratio of $E_{\max }$ to the Lower Bound tends to balance with the increasing of $v_{h}$. This is because although the vertical flight time of UAV decreases as $v_{h}$ grows, the traveling time outside of data collection areas becomes the major part of $E_{\text {max }}$, which makes the ratio of $E_{\max }$ to lower bound remain unchanged. We can find the ratio of $E_{\max }$ to the Lower Bound is always less than 1.5 , which can prove the validity of the algorithm. We also find that the ratio decreases with increasing $V_{i}$ for each sensor since both the hovering times of $E_{\text {max }}$ and Lower Bound increase as the amount of data carried by sensors grows. Fig. 8(b) shows that $E_{\text {max }}$ decreases with the increasing of $v_{h}$ since UAVs need less vertical flight time to arrive the hovering point in each data collection area for gathering data from sensor.

In Fig. 9, we use the interval [1,3] MB to pick a uniformly distributed random $V_{i}$ for each sensor $s_{i} \in S$ and set $n=30$, $h=50 \mathrm{~m}, W=2 \mathrm{MB} / \mathrm{s}, m=3$ and $R=60,80,100,120$, 140 m , and change $v_{f}$ from 8 to $20 \mathrm{~m} / \mathrm{s}$. Fig. 9(a) shows that the ratio of $E_{\max }$ to the Lower Bound decreases with increasing $v_{f}$. This is because the hovering time part of both $E_{\max }$ and Lower Bound is unchanged while the traveling time of them decreases as $v_{f}$ grows. We also observe that the ratio of $E_{\max }$ to Lower Bound becomes larger as $R$ increases, since the traveling time of both $E_{\max }$ and the Lower Bound increases with increasing $R$ but the hovering time of them is unchanged and becomes the main time cost of UAVs. Fig. 9(b) shows that the time cost of UAVs decreases as $v_{f}$ grows since the total traveling time of UAV decreases with the increasing of $v_{f}$. We also find the time consumption of UAV decreases as $R$


Fig. 10. Simulations through varying $W$ from $1 \mathrm{Mb} / \mathrm{s}$ to $7 \mathrm{Mb} / \mathrm{s}$ under different $h$.
decreases. This is because that the data transmission rate from sensors to UAV raises with the decreasing of $R$, which leads to descent in the total time consumption of UAV.

Fig. 10 illustrates the performance of FTPM when we set $n=60, R=100 \mathrm{~m}, m=3, v_{f}=10 \mathrm{~m} / \mathrm{s}, v_{h}=3 \mathrm{~m} / \mathrm{s}$, $h_{0}=10 \mathrm{~m}$, randomly pick $V_{i}$ from the interval [5, 6] Mb and $h=50,60,70,80,90$, and change $W$ from $1 \mathrm{Mb} / \mathrm{s}$ to $7 \mathrm{Mb} / \mathrm{s}$. Fig. 10(a) gives the changing trend of the ratio of $E_{\max }$ to the Lower Bound. It is observed that the ratio decreases with the increasing of $W$. This is because as $W$ increases, the proportion of the horizontal flight time outside of data collection areas to $E_{\max }$ increases while the proportion of the horizontal flight time outside of data collection areas of Lower Bound, $\frac{L(T r)}{v_{f}}$ is unchanged. We also find the ratio of $E_{\max }$ to Lower Bound remains stable as $h$ increases, since both the horizontal flight time and vertical flight time of UAVs decreases, which can keep their values unchanged. Fig. 10(b) shows that $E_{\text {max }}$ decreases with the increasing of $W$ since UAVs need less data transmission time to collect data from sensors.

## VII. CONCLUSION

In this paper, we identify the Fine-grained Trajectory Plan for multi-UAVs (FTP) problem, which focuses on finding the fine-grained trajectory plans for $m$ UAVs. The objective of the problem is to minimize the maximum time cost of UAVs such that all sensory data carried by sensors in WSN is collected and transported to the base station. Then we prove that the FTP problem is NP-hard. Afterwards, we first investigate a special case of FTP problem where $m=1$, called FTPS. Then we propose an approximation algorithm FTPSA for solving the FTPS problem. Based on the FTPSA algorithm, we present an approximation algorithm FTPM to design a fine-grained flight plan for each of $m$ UAVs, which not only gives the flight paths of multiple UAVs but also provides the detailed hovering and traveling plans of UAVs. According to the theoretical analysis and simulations, we can verify that the proposed algorithms have great performance.

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