

3 4 5 6 7 On Dynamic Fundamental Diagrams: Implications for 8 Automated Vehicles

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16 17 Abstract

18 The traffic fundamental diagram (FD) describes the relationships among fundamental
19 traffic variables of flow, density, and speed. FD represents fundamental properties of
20 traffic streams, giving insights into traffic performance. This paper presents a
21 theoretical investigation of dynamic FD properties, derived directly from vehicle car-
22 following (control) models to model traffic hysteresis. Analytical derivation of
23 dynamic FD is enabled by (i) frequency-domain representation of vehicle kinematics
24 (acceleration, speed, and position) to derive vehicle trajectories based on transfer
25 function and (ii) continuum approximation of density and flow, measured along the
26 derived trajectories using Edie's generalized definitions. The formulation is generic:
27 the derivation of dynamic FD is possible with any analytical car-following (control)
28 laws for human-driven vehicles or automated vehicles (AVs). Numerical experiments
29 shed light on the effects of the density-flow measurement region and car-following
30 parameters on the dynamic FD properties for an AV platoon.

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32 *Keywords:* Automated vehicle, Dynamic fundamental diagram, Traffic
33 hysteresis, Traffic oscillation.

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1 Introduction

The traffic fundamental diagram (FD) is a representation of the relationships among fundamental traffic variables of flow, density, and speed. FD describes the fundamental properties of traffic streams, giving insights into traffic performance. Since its inception by the seminal Greenshields model (Greenshields et al., 1935), numerous studies have ensued to confirm its existence (e.g., Ahn et al., 2004; Cassidy, 1998) and determine the shape, giving rise to various families of models (e.g., Newell, 1993; Smulders, 1986; Wu, 2002). FD has been widely used as an important basis for planning and operational analysis (e.g., level of service determination (HCM, 2016), and dynamic traffic management such as ramp metering (e.g., Papageorgiou & Kotsialos, 2002; Papamichail et al., 2010) and variable speed limit (VSL) control (Carlson et al., 2010; Chen et al., 2014a; Chen & Ahn, 2015; Hegyi et al., 2005; Knoop et al., 2010).

The FD, in the traditional sense, describes steady traffic properties and is often referred to as a ‘static’ traffic model. However, in their seminal empirical study, Treiterer & Myers, (1974) discovered ‘traffic hysteresis’ - an elliptical evolution of flow-density relationship as vehicles decelerate and accelerate during a major traffic disturbance. Since then, presence of traffic hysteresis has been confirmed and theorized by many studies (Chen et al., 2012; Zhang, 1999), though some studies suggest that the hysteresis magnitude, attributed to car-following, has been exaggerated in earlier studies (e.g., Ahn & Vadlamani, 2010; Coifman et al., 2018; Laval, 2011). Generally, there are two types of hysteresis discussed. The first type stems from traffic oscillations around an equilibrium traffic state in congested traffic. For this type, the initial equilibrium state is restored after the passage of disturbance (Treiterer & Myers, 1974). The other type involves a change in equilibrium state, particularly from uncongested to congested states. A three-phase car following theory by Zhang (1999) and Zhang & Kim (2005) describes this type of traffic hysteresis, where ‘capacity drop’ phenomenon (i.e., lower throughput after transitioning to a congested state) is emphasized. The present paper focuses on the first type.

In addition, automated vehicles (AVs), with their increasing adoption rate, will likely bring systematic changes to traffic properties, both static and dynamic. Notably, (T. Li et al., 2022) provided a thorough empirical analysis of FD using experimental data from 17 adaptive cruise control (ACC) vehicles using the measurement method developed by Shi & Li (2021). They found that the ACC vehicles exhibit linear FDs (in the congested branch), though the magnitudes of FD parameters can be significantly different from those for human-driven vehicles, depending on the input setting. Further, the experimental study using four different commercial AVs with ACC has verified the existence of traffic hysteresis in AV platoons (Makridis et al., 2021). While insightful, major shortcomings of these experimental studies are that (i) the investigations are limited to static properties, or (ii) the ACC algorithms are proprietary and unknown to the public, thus the underlying mechanisms remain unknown. Notably, some theoretical investigations of FD with AVs exist in the literature (Shi & Li, 2021; Yao et al., 2022; J. Zhou & Zhu, 2020); however, the scope remains largely limited to static properties. Thus, we have a limited understanding of how the control formulation and parameter setting impact dynamic FD features.

Dynamic properties of FD, particularly traffic hysteresis, have important implications for dynamic traffic control. For example, some well-known VSL control methods are based on the first-order kinematic wave (KW) theory (Chen et al., 2014b; Han et al., 2017; Hegyi et al., 2005), in which traffic evolution is described by solving a system of static FD equation and a first-order partial differential equation for flow conservation. These methods determine appropriate speed limits in a dynamic fashion by predicting the traffic states the imposed speed limits will induce. However, the first order KW models assume infinite acceleration/deceleration, thereby failing to

81 capture complex features such as capacity drop, stop-and-go oscillations, and hysteresis observed
82 in real traffic(Logghe & Immers, 2008; Nagel & Nelson, 2005). Higher order KW models
83 introduce an additional partial differential equation to capture finite vehicle
84 acceleration/deceleration (while still working with a static FD)(Aw & Rasche, 2000; Lebacque et
85 al., 2007; H. Payne, 1971; H. J. Payne, 1971; Van Wageningen-Kessels et al., 2015; Zhang, 1999).
86 Owing to this treatment, some of these higher-order KW models can successfully reproduce the
87 above traffic features, including traffic hysteresis. While these models provide a more accurate
88 description of dynamic traffic, their applications for dynamic traffic control have been limited (
89 exception: Carlson et al., 2010) due to their complexity in interpretation, calibration and validation,
90 numerical approximation, and computation.

91 In this paper, we conduct a theoretical investigation of dynamic FD, a kind that can depict
92 key higher-order, nonlinear traffic features such as traffic hysteresis. It is derived directly from
93 car-following control models for AV to provide direct insight into how vehicle behavior scales up
94 to dynamic traffic behavior, which is currently missing in the literature. We focus on dynamic FD
95 (rather than the KW model), in light of contemporary traffic control strategies, such as
96 reinforcement learning(Han et al., 2022; Li et al., 2017) and control theory(Zhou et al., 2020) based
97 control, that do not necessarily adhere to the KW theory. For these strategies, incorporation of
98 dynamic FD presents a classic and elegant approach to maintain physical validity. We also
99 emphasize the direct connection between vehicle and traffic behavior because connected AVs
100 (CAVs) will likely serve as control actuators in future traffic control. Thus, the linkage provides
101 direct insight into how CAVs should be controlled to achieve a specific traffic performance (e.g.,
102 beyond string stability). Further, it can provide a simpler platform for control: desirable vehicle
103 control can be achieved through adjusting control parameter settings, rather than prescribing a
104 precise form of vehicle trajectory. Our framework directly maps the car following law to the
105 dynamic fundamental diagram to provide insight into the mechanisms, which can be harnessed for
106 dynamic traffic management (e.g., ramp metering, and variable speed limit) together with vehicle
107 control.

108 In this study, analytical derivation of dynamic FD is enabled by (i) frequency-domain
109 representation of vehicle kinematics (acceleration, speed, and position) and (ii) continuum
110 approximation of flow and density measured along the derived trajectory using Edie’s generalized
111 definitions(Edie, 1963). The formulation is generic in the sense that derivation of dynamic FD is
112 possible with any analytical car-following (control) laws for human-driven vehicles or AVs. To
113 verify our derivations and identify potential factors affecting dynamic FDs, a series of numerical
114 experiments were conducted. The results show the presence of hysteresis within the AV traffic
115 flow when facing oscillations. The shape and orientation on hysteresis in the dynamic FD are
116 influenced by the frequency characteristics of oscillations (single-frequency or multi-frequency),
117 flow-density measurement region, and the car-following control parameter setting. Particularly,
118 we show that the control gains, total delay in sensing and control actuation, desired time gap, and
119 equilibrium speed all have unique effects on the properties of dynamic FD.

120 The remainder of this paper is organized as follows. Section 2 presents the trajectory
121 expressions for a CAV platoon in frequency domain. Section 3 then analytically derives
122 fundamental variables in dynamic FD. Simulation experiments and their results to illustrate the
123 efficacy of our dynamic FD are provided in Section 4. Finally, Section 5 contains our conclusions
124 and limitations.

126 2 Frequency-domain CAV Trajectory

127 In this section, we mathematically derive the trajectory of CAV based on a transfer function in the
 128 frequency domain. The frequency-domain representation can better describe the evolution of
 129 oscillations in a platoon, compared to traditional time-domain expressions(Zhou et al., 2023).
 130 Further, using a transfer function can unveil the input-output relationship for a time-invariant
 131 dynamic system such as the longitudinal control of CAVs.

132 First, we consider a platoon of N homogenous CAVs, indexed by $l \in \{0, 1, \dots, N - 1\}$, where
 133 0 indicates the leading vehicle. All CAVs are assumed to follow the same deterministic car-
 134 following law, f ,

$$a_l(t) = f(\Delta x_l(t - \theta), v_l(t - \theta), v_{l-1}(t - \theta)), \quad (1)$$

135 where $a_l(\cdot)$, $v_l(\cdot)$, and $\Delta x_l(\cdot)$ respectively denote vehicle l 's acceleration, speed, and spacing with
 136 its predecessor over time. In addition, θ denotes the total time delay caused by CAV's sensing and
 137 control actuation (i.e., system's latency), we treat it as constant for simplicity. By the physical
 138 kinematics law, $\Delta x_l(t) = \int_0^t [v_l(\varphi) - v_{l-1}(\varphi)]d\varphi + \Delta x_l(0)$. To describe car-following under
 139 traffic oscillations, we decompose vehicle trajectories into the nominal and the oscillatory
 140 components. The former represents an equilibrium state, described by a unique relationship
 141 between the equilibrium speed, v_e , and spacing, Δx_e , where v_e varies from 0 to the free flow speed.
 142 Both Δx_e and v_e are constants. In this state, $a_l(t) = f(\Delta x_e, v_e, v_e) = 0$. In the following context,
 143 we assume the equilibrium state will not change throughout the oscillations. Traffic hysteresis
 144 resulting from an equilibrium state change falls beyond the scope of this discussion. The oscillatory
 145 component describes the deviation from the equilibrium state, characterized by $\Delta \hat{x}_l(t) :=$
 146 $\Delta x_l(t) - \Delta x_e$ and $\hat{v}_l(t) := v_l(t) - v_e$. For convenience, here we linearize the system over the
 147 equilibrium point $(\Delta x_e, v_e, v_e)$ to analyze the first-order residual impacts of
 148 $(\Delta \hat{x}_l(t), \hat{v}_l(t), \hat{v}_{l-1}(t))$ on acceleration $a_l(t)$ by letting $\hat{f}(\Delta \hat{x}_l(t), \hat{v}_l(t), \hat{v}_{l-1}(t)) =$
 149 $f(\Delta x_l(t), v_l(t), v_{l-1}(t))$. \hat{f} is a shifted function of f via shifting $\Delta \hat{x}_l(t)$, $\hat{v}_l(t)$, $\hat{v}_{l-1}(t)$ by Δx_e ,
 150 v_e , v_e , respectively. Then the equilibrium state is $\hat{f}(0, 0, 0) = 0$.

151 We also assume the initial conditions of all following CAVs are at equilibrium, i.e., $\Delta x_l(0) =$
 152 Δx_e , $\forall l = 1, 2, \dots, N - 1$. Without loss of generality, let $x_0(0) = 0$. Then, by the kinematics law,
 153 the oscillatory position can be written as:

$$\Delta \hat{x}_l(t) = \int_0^t \hat{v}_l(\varphi)d\varphi - \int_0^t \hat{v}_{l-1}(\varphi)d\varphi. \quad (2)$$

154 Further, for derivation convenience, we conduct linearization on f . Through a Taylor series
 155 expansion near the origin $(0, 0, 0)$ and ignoring higher order terms as they are very close to zero, f
 156 can be linearized as:

$$\hat{f} = a_l(t) = \hat{f}_1' \Delta \hat{x}_l(t - \theta) + \hat{f}_2' \hat{v}_l(t - \theta) + \hat{f}_3' \hat{v}_{l-1}(t - \theta), \quad (3)$$

157 where \hat{f}_1' , \hat{f}_2' , and \hat{f}_3' are gradients, obtained via the first-order partial derivative corresponding to
 158 each term; i. e., $\hat{f}_1' = \frac{\partial f}{\partial \Delta \hat{x}_l}$, $\hat{f}_2' = \frac{\partial f}{\partial \hat{v}_l}$, and $\hat{f}_3' = \frac{\partial f}{\partial \hat{v}_{l-1}}$. Note that \hat{f} is the linear approximation of
 159 general nonlinear CF law f .

160 Combining Eqs. (2) and (3), we have

$$\hat{v}_l(t) = \hat{f}_1' \left(\int_0^{t-\theta} \hat{v}_l(\varphi)d\varphi - \int_0^{t-\theta} \hat{v}_{l-1}(\varphi)d\varphi \right) + \hat{f}_2' \hat{v}_l(t - \theta) + \hat{f}_3' \hat{v}_{l-1}(t - \theta). \quad (4)$$

161 For simplification, we further assume $\hat{v}_l(0) = 0$. As Eq. (4) is still highly non-linear, we take
 162 the Laplace transform for further derivation of CAV trajectory in the frequency domain. The
 163 transformation provides a convenient platform to study a linear dynamic system by substituting
 164 integration with division and differentiation with multiplication. Then, Eq. (4) is converted as
 165 follows in the frequency domain:

$$s\hat{V}_l(s) - \hat{f}_1' \frac{e^{-\theta s}}{s} \hat{V}_l(s) - \hat{f}_2' \hat{V}_l(s) e^{-\theta s} = -\hat{f}_1' \frac{e^{-\theta s}}{s} \hat{V}_{l-1}(s) + \hat{f}_3' \hat{V}_{l-1}(s) e^{-\theta s}, \quad (5)$$

166 where $s = j\omega$, j denotes the imaginary unit, and ω is the frequency. $\hat{V}_l(s)$ and $\hat{V}_{l-1}(s)$ are
 167 respectively the oscillatory speeds of vehicles l and $l - 1$ in the frequency domain. Then the
 168 xtransfer function, $G(s)$, can be defined using the oscillatory components of speed:

$$G(s) = \frac{\hat{x}_l(s)}{\hat{x}_{l-1}(s)} = \frac{s\hat{V}_l(s)}{s\hat{V}_{l-1}(s)} = \frac{\hat{V}_l(s)}{\hat{V}_{l-1}(s)} = \frac{-\hat{f}_1' e^{-\theta s} + \hat{f}_3' s e^{-\theta s}}{s^2 - \hat{f}_1' e^{-\theta s} - \hat{f}_2' s e^{-\theta s}}. \quad (6)$$

169 We write $G(j\omega) = |G(j\omega)|e^{j\angle G(j\omega)}$ with its norm $|G(j\omega)|$ and angle $\angle G(j\omega)$. The transfer
 170 function implies how output (i.e. $\hat{V}_l(s)$) responds to the input ((i.e. $\hat{V}_{l-1}(s)$)). It can be used to
 171 describe oscillation propagation along a platoon, which is a crucial element to describe dynamic
 172 traffic.

173 **Remark 1** The transfer function described above is intricately linked to the norm $|G(j\omega)|$ and the
 174 phase shift $\angle G(j\omega)$. The transfer function defined above is intricately linked to the norm $|G(j\omega)|$
 175 and the phase shift $\angle G(j\omega)$ (Zhou et al., 2023). It is noteworthy that the framework established
 176 herein remains applicable even in the presence of alternative nonlinear or unidentified car-
 177 following laws, where we can replace the transfer function with a describing function (Li et al.,
 178 2012, 2014; Li & Ouyang, 2011; Wang et al., 2020) or a data-driven transfer function (Y. Zhou et
 179 al., 2023) to approximate the behavior.

180 Based on the above work, we are ready to derive the position for vehicle l in CAV platoon.
 181 Without loss of generality, we assume the position of leading vehicle 0 as follows:

$$x_0(t) = \bar{x}_0(t) + \hat{x}_0(t), \quad (7)$$

Where

$$\bar{x}_0(t) = v_e t, \quad (8)$$

$$\hat{x}_0(t) = \sum_{m=1}^{\infty} A_0^{(m)} \sin(\omega_m t + \phi_m), \quad (9)$$

182 where $\bar{x}_0(t)$ is the nominal component and $\hat{x}_0(t)$ the oscillation component. Note that we
 183 decompose the oscillation into the sum of sinusoidal waves through its Fourier transform; see Eq.
 184 (6b). m is the index of oscillatory waves, and $A_0^{(m)}$ is the amplitude of oscillatory wave with
 185 frequency ω_m and phase shift ϕ_m . Sinusoidal functions are selected for their advantageous
 186 attributes of boundedness and periodicity. We employ the mathematical framework of sinusoidal
 187 functions, $\hat{x}_0(t) = \sum_{m=1}^{\infty} A_0^{(m)} \sin(\omega_m t + \phi_m)$, to represent the oscillatory components. We
 188 employ multiple sinusoidal waves to model oscillations with compound frequencies to be more
 189 realistic. This approach is widely adopted in analytical modeling of waves, as exemplified by Li
 190 et al. (2014).

191

192 We can also derive the speed and acceleration of the leading vehicle composed of the nominal
 193 and oscillatory components:

$$v_0(t) = \bar{v}_0(t) + \hat{v}_0(t) = v_e + A_0^{(m)} \omega_m \sum_{m=1}^{\infty} \cos(\omega_m t + \phi_m), \quad (4)$$

$$a_0(t) = \bar{a}_0(t) + \hat{a}_0(t) = -A_0^{(m)} \omega_m^2 \sum_{m=1}^{\infty} \sin(\omega_m t + \phi_m), \quad (5)$$

194 By Eq. (6), we can further find $\hat{X}_l(s) = \hat{X}_{l-1}(s)G(s)$. Therefore, given the platoon satisfying the
 195 aforementioned initial condition, vehicle l 's position has the following nominal and oscillatory
 196 components:

$$\bar{x}_l(t) = v_e t - l\Delta x_e, \quad (12)$$

$$\begin{aligned} \hat{x}_l(t) &= \mathcal{L}^{-1}(\hat{X}_l(s)) = \mathcal{L}^{-1}(\hat{X}_0(s)G^l(s)) = \mathcal{L}^{-1}(\hat{X}_0(s)|G(s)|^l e^{jl\angle G(s)}) \\ &= \lim_{M \rightarrow \infty} \sum_{m=1}^M A_0^{(m)} |G(j\omega_m)|^l \sin(\omega_m t + \phi_m + l\angle G(j\omega_m)). \end{aligned} \quad (13)$$

197 Correspondingly, for vehicle l , the speed and acceleration are:

$$v_l(t) = v_e + A_0^{(m)} |G(j\omega_m)|^l \omega_m \sum_{m=1}^{\infty} \cos(\omega_m t + \phi_m + l\angle G(j\omega_m)) \quad (14)$$

$$a_l(t) = -A_0^{(m)} |G(j\omega_m)|^l \omega_m^2 \sum_{m=1}^{\infty} \sin(\omega_m t + \phi_m + l\angle G(j\omega_m)) \quad (15)$$

198 It is noteworthy that the negative sign within the oscillatory component of acceleration (Eq.
 199 15) bears limited consequences, owing to the periodic nature of the sinusoidal function. Then for
 200 a platoon consisting of N homogenous CAVs, the length of the platoon can be derived as (after
 201 some simplification):

$$x_0(t) - x_N(t) = N\Delta x_e + \sum_{m=1}^M R A_0^{(m)} [\sin(\omega_m t + \phi_m - \phi_c)], \quad (6)$$

202 where $R = \sqrt{1 - 2|G(j\omega_m)|^N \cos(N\angle G(j\omega_m)) + |G(j\omega_m)|^{2N}}$ and $\phi_c =$
 203 $\arctan \frac{|G(j\omega_m)|^N \sin(N\angle G(j\omega_m))}{1 - |G(j\omega_m)|^N \cos(N\angle G(j\omega_m))}$ are two constant values. For a comprehensive derivation, please
 204 refer to [Appendix A](#).

205 3 Analytical Model of Dynamic FD

206 Here we derive the fundamental traffic variables using the position derived in Section 2 and then
 207 FD. This approach allows us scale up to dynamic FD while retaining dynamic vehicle
 208 characteristics. The discussion of FD pertains to traffic hysteresis within the congested regime. To
 209 this end, we first apply Edie's generalized definitions for traffic density and flow (Edie, 1963)
 210 since they are flexible for different measurement methods. (In Edie's definition, density is denoted
 211 as the total time spent by all N vehicles divided by the area of a measurement time-space region.

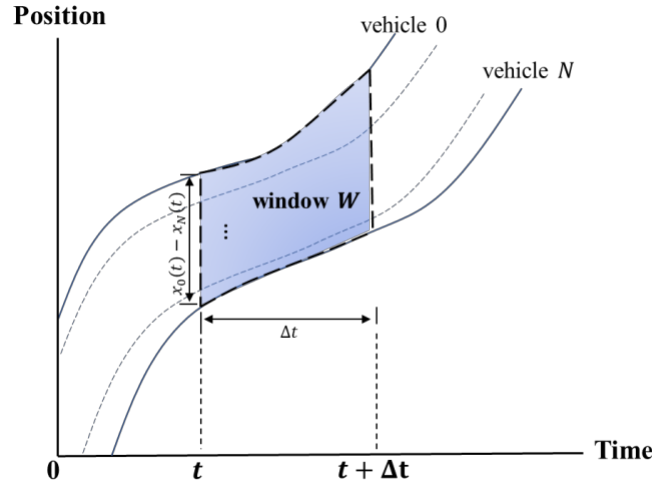
212 Flow is denoted as the total distance travelled by all N vehicles divided by the area of time-space
 213 region.) Specifically, the generalized definitions give density and flow as follows:

$$k(t, \Delta t, N) = \frac{N\Delta t}{W(t, \Delta t, N)}, \quad (7)$$

$$Q(t, \Delta t, N) = \frac{\sum_{l=1}^N (x_l(t + \Delta t) - x_l(t))}{W(t, \Delta t, N)}, \quad (8)$$

214 where k and Q are density and flow, respectively. W is the area of a predefined time-space region.
 215 Instead of a traditional rectangular window, we define a customized time-space window W at time
 216 t along the trajectories covering all N CAVs with a width Δt . This window, depicted in Fig. 1, can
 217 be referred to as vertical window¹. Hence, we have the area as:

$$\begin{aligned} W(t, \Delta t, N) &= \int_t^{t+\Delta t} [x_0(t) - x_N(t)] dt \\ &= N\Delta x_e \Delta t + \sum_{m=1}^M \frac{RA_0^{(m)}}{\omega_m} [\cos(\omega_m t + \phi_m - \phi_c) - \cos(\omega_m(t + \Delta t) + \phi_m - \phi_c)]. \end{aligned} \quad (99)$$



218

219

Figure 1: Illustration of vertical time-space window.

220 This customized window, referred to as the "vertical window," is recognized for its
 221 straightforward interpretability and mathematical convenience. Then, combined with Eq. (99), Eqs.
 222 (7) and (8) can be rewritten as:

$$k(t, \Delta t, N) = \frac{N\Delta t}{N\Delta x_e \Delta t + \sum_{m=1}^M \frac{RA_0^{(m)}}{\omega_m} [\cos(\omega_m t + \phi_m - \phi_c) - \cos(\omega_m(t + \Delta t) + \phi_m - \phi_c)]}, \quad (20)$$

¹ In Laval (2011), the window is slanted according to the maximum congestion wave speed to maximize the chance of having a homogenous traffic state. This paper uses vertical windows instead for mathematical elegance and due to the fact that the wave speed can be nonlinear along the vehicle platoon (considering the disturbance dampening) and potentially time-varying (Shi et al. 2023). This property renders analytical derivation based on slanted windows mathematically prohibitive.

$$Q(t, \Delta t, N) = \frac{Nv_e \Delta t + \sum_{l=1}^N \sum_{m=1}^M A_0^{(m)} |G(j\omega_m)|^l [\sin(\omega_m(t+\Delta t) + \phi_m + l\angle G(j\omega_m)) - \sin(\omega_m t + \phi_m + l\angle G(j\omega_m))]}{N\Delta x_e \Delta t + \sum_{m=1}^M \frac{RA_0^{(m)}}{\omega_m} [\cos(\omega_m t + \phi_m - \phi_c) - \cos(\omega_m(t+\Delta t) + \phi_m - \phi_c)]} \quad (21)$$

223 3.1 Continuum approximation

224 To make sure that the derived density and flow are physically meaningful and continuous in the
225 time domain, we take the limit, $\Delta t \rightarrow \mathbf{0}$ of Eqs. (20) and (21):

$$k(t, N) = \frac{N}{N\Delta x_e + \sum_{m=1}^M RA_0^{(m)} \sin(\omega_m t + \phi_m - \phi_c)}, \quad (22)$$

$$Q(t, N) = \frac{Nv_e + \sum_{l=1}^N \sum_{m=1}^M A_0^{(m)} |G(j\omega_m)|^l \omega_m \cos(\omega_m t + \phi_m + l\angle G(j\omega_m))}{N\Delta x_e + \sum_{m=1}^M RA_0^{(m)} \sin(\omega_m t + \phi_m - \phi_c)}. \quad (23)$$

226 The result shows that k and Q are cyclic functions, whose periods are the least common
227 multiple of $\frac{2\pi}{\omega_m}$, for $1 \leq m \leq M$.

228 **Remark 2** From Eqs. (22) and (23), Q and k are both functions of v_e (typically Δx_e is also a
229 function of v_e), oscillation components $A_0^{(m)}$, ω_m and ϕ_m , as well as AV CF control G . Thus, Q
230 and k together can describe the dynamics of FD directly.

231 Next, we will further discuss two special scenarios based on the continuum approximation
232 (CA). The first scenario pertains to deriving k and Q for a short oscillating platoon, where an
233 oscillation comprises a single dominant sinusoidal wave. In practice scenarios, even when
234 oscillations involve a combination of waves with varying frequencies, there tends to be one
235 ‘dominant wave’ where its frequency component has the highest magnitude or power. Identifying
236 the dominant wave can be done through techniques such as Fourier analysis, where an oscillation
237 wave is decomposed into its frequency components using the Fourier transform. The second
238 scenario explores k and Q when the single-wave oscillation evolves along a long string stable
239 platoon. Note that in a string stable platoon system, the system’s CF behavior remains controllable
240 even when faced with various disturbances in the environment. String stability is an important and
241 desired property from a safety perspective. Hereafter, we guarantee string stability via constraining
242 the norm of the transfer function.

243 **Scenario I (Short Oscillating Platoon).** In the case of a single dominant wave (i.e., $M = 1$) with
244 p as the principal frequency component, we have

$$k(t, N) = \frac{N}{N\Delta x_e + RA_0^{(p)} |G(j\omega_p)|^N \sin(\omega_p t + \phi_p - \phi_c)}, \quad (24)$$

$$Q(t, N) = \frac{Nv_e + \sum_{l=1}^N A_0^{(p)} |G(j\omega_p)|^l \omega_p \cos(\omega_p t + \phi_p + l\angle G(j\omega_p))}{N\Delta x_e + RA_0^{(p)} |G(j\omega_p)|^N \sin(\omega_p t + \phi_p - \phi_c)}. \quad (25)$$

245 In this case, k and Q are both cyclic functions over t with the same oscillation period of
246 $2\pi/\omega_p$. Within one deceleration-acceleration cycle, we can further compute the expectation of k
247 and Q , respectively. For simplicity, let $\phi_p, \phi_c = 0$ as we can adjust the upper bound and lower

248 bound of integral during integration. We further define, $\bar{A} = \frac{RA_0^{(p)}|G(j\omega_p)|^N}{N}$, where \bar{A} stands for the
 249 average magnitude of oscillation through the platoon.

250 **Proposition 1 (density expectation):** The expectation of k , $E(k)$, follows $E(k) = \frac{1}{\sqrt{(\Delta x_e)^2 - (\bar{A})^2}}$.

251 *Proof:*

$$E(k) = \frac{\omega_p}{2\pi} \int_0^{\frac{2\pi}{\omega_p}} k(t, N) dt = \frac{\omega_p}{2\pi} \int_0^{\frac{2\pi}{\omega_p}} \frac{N}{N\Delta x_e + RA_0^{(p)}|G(j\omega_p)|^N \sin(\omega_p t)} dt =$$

$$\frac{\omega_p}{2\pi} \int_0^{\frac{2\pi}{\omega_p}} \frac{1}{\Delta x_e + \left(\frac{RA_0^{(p)}|G(j\omega_p)|^N}{N} \right) \sin(\omega_p t)} dt,$$

252 Letting $\frac{RA_0^{(p)}|G(j\omega_p)|^N}{N} = \bar{A}$, we have $E(k) = \frac{\omega_p}{2\pi} \int_0^{\frac{2\pi}{\omega_p}} \frac{1}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt = \frac{\omega_p}{2\pi} \frac{\sqrt{(\Delta x_e)^2 - (\bar{A})^2}}{\omega_p} = \frac{1}{\sqrt{(\Delta x_e)^2 - (\bar{A})^2}}$.

253 Therefore, we obtain Proposition 1.

254 *Q. E. D.*

255 Note: when $|G(j\omega_p)| < 1$ and N is large (e.g., $N = 5$), $\bar{A} \ll \Delta x_e$, therefore, $E(k) \approx \frac{1}{\Delta x_e}$.

256 **Proposition 2 (flow expectation):** The expectation of Q , $E(Q) = \frac{v_e}{\sqrt{(\Delta x_e)^2 - (\bar{A})^2}}$.

257 *Proof:*

$$E(Q) = \frac{\omega_p}{2\pi} \int_0^{\frac{2\pi}{\omega_p}} \frac{Nv_e + \sum_{l=1}^N A_0^{(p)}|G(j\omega_p)|^l \omega_p \cos(\omega_p t + l\angle G(j\omega_p))}{N\Delta x_e + RA_0^{(1)}|G(j\omega_p)|^N \sin(\omega_p t)} dt$$

$$= \frac{\omega_p}{2\pi} v_e \int_0^{\frac{2\pi}{\omega_p}} \frac{1}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt + \frac{\omega_p}{2\pi} \frac{A_0^{(p)} \omega_p}{N} \int_0^{\frac{2\pi}{\omega_p}} \frac{\sum_{l=1}^N |G(j\omega_p)|^l \cos(\omega_p t + l\angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt$$

$$= \frac{\omega_p}{2\pi} v_e \int_0^{\frac{2\pi}{\omega_p}} \frac{1}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt + \frac{\omega_p}{2\pi} \frac{A_0^{(p)} \omega_p}{N} \sum_{l=1}^N |G(j\omega_p)|^l \int_0^{\frac{2\pi}{\omega_p}} \frac{\cos(\omega_p t + l\angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt$$

$$= \frac{\omega_p}{2\pi} v_e \frac{2\pi}{\omega_p \sqrt{(\Delta x_e)^2 - (\bar{A})^2}} + \frac{\omega_p}{2\pi} \frac{A_0^{(p)} |G(j\omega_p)|^l \omega_p}{N} \cdot F$$

258 where $F = \int_0^{\frac{2\pi}{\omega_p}} \frac{\cos(\omega_p t + l\angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt$. As $\int_0^{\frac{2\pi}{\omega_p}} \frac{\cos(\omega_p t + l\angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p t)} dt = 0$ ($l \in [1, N]$, $l \in Z$), $F = 0$.

259 Hence, $E(Q) = \frac{v_e}{\sqrt{(\Delta x_e)^2 - (\bar{A})^2}}$, we have Proposition 3.

260 *Q. E. D.*

261 Note: Similarly, when $|G(j\omega_p)| < 1$ and N is large (e.g., $N = 5$), $\bar{A} \ll \Delta x_e$. Therefore,
 262 $E(Q) \approx \frac{v_e}{\Delta x_e}$, which is consistent with the fundamental definition, $E(Q) = v_e E(k)$.

263 **Scenario II (Long String Stable Platoon).** Considering a long platoon (i.e., $N \rightarrow \infty$) that is string
 264 stable (i.e., $|G(j\omega_p)| < 1$, we have

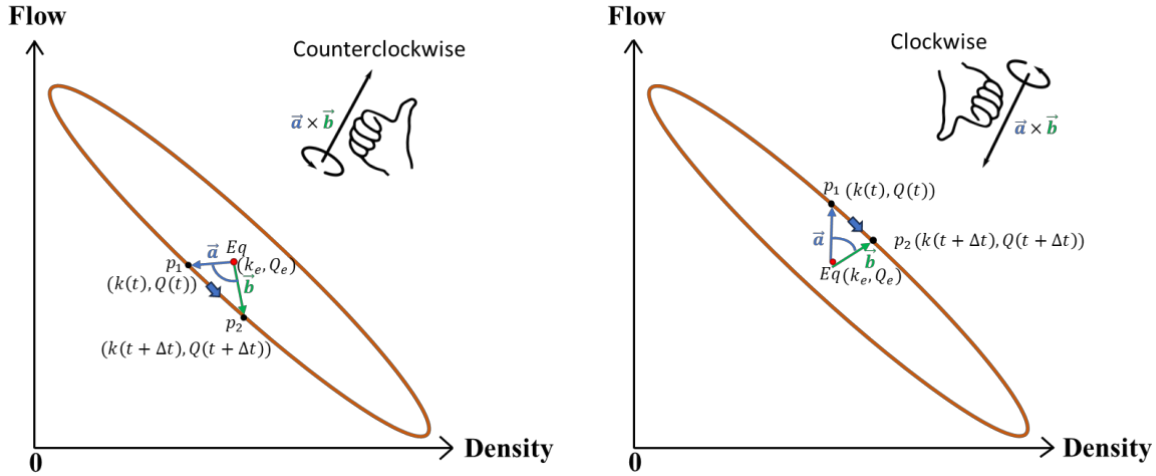
$$\lim_{N \rightarrow \infty} E(k(N)) = \frac{1}{\Delta x_e},$$

$$\lim_{N \rightarrow \infty} E(Q(N)) = \frac{v_e}{\Delta x_e},$$

265 which suggests a traditional FD for steady traffic flow. This is intuitive because if the platoon is
 266 long enough and string stable, oscillations would be dampened eventually to the equilibrium point.

267 3.2 Hysteresis orientation

268 Here we analytically determine the orientation of hysteresis that can arise during an oscillation.
 269 Specifically, a random point on the hysteresis loop, referred as p_1 , is selected, as depicted in Fig.2.
 270 Accordingly, its density and flow at time point t are denoted as $k(t)$ and $Q(t)$. Then, after a small
 271 time interval Δt , the corresponding point on the loop is denoted as p_2 , characterized by density
 272 $k(t + \Delta t)$ and flow $Q(t + \Delta t)$. The equilibrium point is expressed as $Eq(k_e, Q_e)$, where k_e and
 273 Q_e are respectively the equilibrium density and flow. Then, we can define two vectors \vec{a} and \vec{b} to
 274 describe the directional segments towards p_1 and p_2 , respectively, from the equilibrium point:
 275 $\vec{a}: (k(t) - k_e, Q(t) - Q_e)$, $\vec{b}: (k(t + \Delta t) - k_e, Q(t + \Delta t) - Q_e)$.
 276



277 (a)

278 (b)

279 Figure 2: Example of traffic hysteresis orientation (a) counter-clockwise and (b) clockwise

281 Then to find the hysteresis direction, we take the cross product of \vec{a} and \vec{b} and then apply the
 282 right-hand rule. $\vec{a} \times \vec{b}$ can be expressed as determinant:

$$\vec{a} \times \vec{b} = \begin{vmatrix} k(t) - k_e & Q(t) - Q_e \\ k(t + \Delta t) - k_e & Q(t + \Delta t) - Q_e \end{vmatrix}. \quad (26)$$

283 Eq. (20) can be further written as $\vec{a} \times \vec{b} = (Q(t + \Delta t) - Q_e)(k(t) - k_e) - (Q(t) -$
 284 $Q_e)(k(t + \Delta t) - k_e)$. After reorganizing, we obtain,

$$\vec{a} \times \vec{b} = Q(t + \Delta t)(k(t) - k_e) - k(t + \Delta t)(Q(t) - Q_e) + Q(t)k_e - k(t)Q_e. \quad (27)$$

285 From **Scenario II**, $k_e = \frac{1}{\Delta x_e}$ and $Q_e = \frac{v_e}{\Delta x_e}$. And by combining Eq. (26) and Eq. (27) from
 286 **Scenario I**, we obtain (after some simplification)

$$\vec{a} \times \vec{b} = \frac{A_0^{(p)} \omega_p}{N \Delta x_e} \left[\frac{\sum_{l=1}^N |G(j\omega_p)|^l \cos(\omega_p t + \phi_p + l \angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p t + \phi_p - \phi_c)} - \frac{\sum_{l=1}^N |G(j\omega_p)|^l \cos(\omega_p(t + \Delta t) + \phi_p + l \angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p(t + \Delta t) + \phi_p - \phi_c)} \right]. \quad (28)$$

287 **Proposition 3 (hysteresis orientation identification).** The hysteresis loop evolves clockwise on
 288 a dynamic FD if $\sum_{l=1}^N |G(j\omega_p)|^l \cos(l \angle G(j\omega_p) + \phi_c) < 0$ and counter-clockwise if
 289 $\sum_{l=1}^N |G(j\omega_p)|^l \cos(l \angle G(j\omega_p) + \phi_c) > 0$. Obviously, there is no hysteresis if
 290 $\sum_{l=1}^N |G(j\omega_p)|^l \cos(l \angle G(j\omega_p) + \phi_c) = 0$.

291 *Proof:*

292 Consider $f(t) = \frac{\sum_{l=1}^N |G(j\omega_p)|^l \cos(\omega_p t + \phi_p + l \angle G(j\omega_p))}{\Delta x_e + \bar{A} \sin(\omega_p t + \phi_p - \phi_c)}$, then $\vec{a} \times \vec{b} = \frac{A_0^{(p)} \omega_p}{N \Delta x_e} [f(t) - f(t + \Delta t)]$.

293 To investigate the orientation, it is necessary to examine the positivity of $\vec{a} \times \vec{b}$. Consider in one
 294 cycle, *i. e.*, $t \in \left[0, \frac{2\pi}{\omega_m}\right]$, since $\frac{A_0^{(p)} \omega_p}{N \Delta x_e}$ is always positive definite, and $f(t + \Delta t) = f(t) + f'(t) * \Delta t$
 295 when $\Delta t \rightarrow 0$ (first order approximation), we can rewrite $\vec{a} \times \vec{b} = \frac{A_0^{(p)} \omega_p}{N \Delta x_e} [-f'(t) * \Delta t]$, where
 296 $f'(t)$ represents the first order derivative of $f(t)$. Thus, $\vec{a} \times \vec{b} > 0$ when $f'(t) < 0$, otherwise,
 297 $\vec{a} \times \vec{b} < 0$ when $f'(t) > 0$. According to the quotient rule,

$$f'(t) = \frac{-\sum_{l=1}^N |G(j\omega_p)|^l \omega_p \sin(\omega_p t + \phi_p + l \angle G(j\omega_p)) * (\Delta x_e + \bar{A} \sin(\omega_p t + \phi_p - \phi_c)) - \sum_{l=1}^N |G(j\omega_p)|^l \cos(\omega_p t + \phi_p + l \angle G(j\omega_p)) * (\bar{A} \omega_p) \cos(\omega_p t + \phi_p - \phi_c)}{(\Delta x_e + \bar{A} \sin(\omega_p t + \phi_p - \phi_c))^2}. \quad (29)$$

298 In Eq. (23), since the denominator of $f'(t) > 0$ always holds, we will only focus on the
 299 numerator. Let $\alpha = \omega_p t + \phi_p + l \angle G(j\omega_p)$ and $\beta = \omega_p t + \phi_p - \phi_c$. Then the numerator of Eq.
 300 (23) can be written as:

$$\begin{aligned} & -\sum_{l=1}^N |G(j\omega_p)|^l \omega_p \sin \alpha * (\Delta x_e + \bar{A} \sin \beta) - \sum_{l=1}^N |G(j\omega_p)|^l \cos \alpha * (\bar{A} \omega_p) \cos \beta \\ & = -\omega_p \Delta x_e \sum_{l=1}^N |G(j\omega_p)|^l \sin(\omega_p t + \phi_p + l \angle G(j\omega_p)) - \omega_p \bar{A} \sum_{l=1}^N |G(j\omega_p)|^l \cos(l \angle G(j\omega_p) + \phi_c) \end{aligned} \quad (30)$$

301 Regarding the first term in Eq. (23), it should be noted that for each vehicle l within one cycle,
 302 $E\left(|G(j\omega_p)|^l \sin(\omega_p t + \phi_p + l \angle G(j\omega_p))\right) = 0$. This result arises from the cyclic property of a
 303 sinusoidal function. Therefore $E\left(\sum_{l=1}^N |G(j\omega_p)|^l \sin(\omega_p t + \phi_p + l \angle G(j\omega_p))\right) = \mathbf{0}$. As a result,
 304 the sign of Eq. (24) will only be determined by the second term
 305 $-\omega_p \bar{A} \sum_{l=1}^N |G(j\omega_p)|^l \cos(l \angle G(j\omega_p) + \phi_c)$. According to the right-hand rule, $\vec{a} \times \vec{b} > 0$
 306 corresponds to CCW (Fig. 2(a)), while $\vec{a} \times \vec{b} < 0$ corresponds to CW (Fig. 2(b)).

307 *Q. E. D.*

308 Note that as $A_0^{(p)}$, ω_p , and N are all positive, and the orientation of hysteresis loop is only
 309 determined by $|G(j\omega_p)|$ and $\angle G(j\omega_p)$.

310 4 Numerical Experiments

311 This section presents a series of numerical experiments to demonstrate how the derived analytical
 312 model of dynamic FD works. We start with the experiment setup and demonstrate how CA for FD
 313 can effectively represent traffic hysteresis. We further examine the individual and joint effects of
 314 key model parameters on the properties of dynamic FD. Through these experiments, we gain a
 315 deeper understanding of how the CAV behavior during an oscillation manifests itself in the
 316 hysteresis pattern. In addition, to show the generality of our framework, we put the derivation
 317 process of an HDV model and provide a comparison with AVs in Appendix B.

318 4.1 Numerical experiment setup

319 For the experiments, we adopt the second-order linear feedback controller by Van Arem et al.
 320 (2006) for the AV car-following law, as an example. This controller marks the pioneering car-
 321 following logic specifically designed for CAV. It is simple yet effective, finding wide adoption in
 322 the literature.

323 The control system state at time t is defined as $[\Delta x_e(t) - \Delta x_l(t), v_{l-1}(t) - v_l(t)]^T$, where
 324 the first term is the deviation from equilibrium (desired) spacing and the second term is speed
 325 difference, T denotes transpose. Here, we incorporate two gains: spacing feedback gain k_s and
 326 speed deviation gain k_v . These gains are time-invariant and utilized to regulate the deviation from
 327 equilibrium spacing and the speed difference, respectively. Thus, the acceleration is given by:

$$a_l(t) = k_s \cdot (\Delta x_e(t) - \Delta x_l(t)) - k_v \cdot v_l(t) + k_v \cdot v_{l-1}(t). \quad (31)$$

328 Note that the acceleration gain is not considered here as AV lacks access to feedforward
 329 information from the preceding vehicle. Nevertheless, the proposed analytical model is general
 330 and can be extended to the application of CAV. [The equilibrium spacing uses the widely adopted](#)
 331 [constant time gap policy: \$\Delta x_e\(t\) = v_e\(t\) \times \tau + s_0\$, where \$\tau\$ and \$s_0\$ represent constant desired](#)
 332 [time gap and standstill spacing, respectively.](#)

333 Based on Eq. (3) and Eq. (31), we can further derive $\hat{f}'_1 = \frac{\partial a_l(t)}{\partial \Delta \hat{x}_l(t)} = \frac{\partial k_s(-\Delta \hat{x}_l(t))}{\partial \Delta \hat{x}_l(t)} = -k_s$, $\hat{f}'_2 =$
 334 $\frac{\partial a_l(t)}{\partial \hat{v}_l(t)} = \frac{\partial(-k_s(v_l(t) \times \tau + s_0) - k_v v_l(t))}{\partial \hat{v}_l(t)} = -k_v - k_s \tau$, $\hat{f}'_3 = \frac{\partial a_l(t)}{\partial \hat{v}_{l-1}(t)} = \frac{\partial(k_v v_{l-1}(t))}{\partial \hat{v}_{l-1}(t)} = k_v$. Note that the
 335 partial derivatives are only related to the oscillatory parts ($\Delta \hat{x}_l(t)$, $\hat{v}_l(t)$, $\hat{v}_{l-1}(t)$) instead of the
 336 nominal parts ($\Delta \bar{x}_l(t)$, $\bar{v}_l(t)$, $\bar{v}_{l-1}(t)$).

337 Then, the corresponding transfer function is given as:

$$G(j\omega) = \frac{\hat{V}_l(j\omega)}{\hat{V}_{l-1}(j\omega)} = \frac{k_s e^{-j\theta\omega} + jk_v \omega e^{-j\theta\omega}}{-\omega^2 + k_s e^{-j\theta\omega} + j(k_v + k_s \tau) \omega e^{-\theta j\omega}}. \quad (32)$$

338 As an example, we set $k_s = 1 (s^{-2})$, $k_v = 1 (s^{-1})$, $\tau = 0.8 s$, $v_e = 10 m/s$, $\theta = 0.5 sec$.
 339 [See Table 1 for details.](#) Section 4.3 provides an in-depth exploration of the implications of
 340 parameter settings on dynamic FDs, where the parameters are set based on previous empirical
 341 studies(Gunter et al., 2020, 2021). The total study period is 40 sec and the platoon size, N , is 20.
 342 [Note that this setting serves as the default configuration for all subsequent experiments unless](#)
 343 [otherwise specified \(i.e., Single oscillation case in Table 1\).](#)

344 Table 1: Default settings

Parameters		Values	
Spacing deviation gain k_s		$1(s^{-2})$	
Speed difference gain k_v		$1(s^{-1})$	
Desired time gap τ		$0.8(s)$	
Equilibrium speed v_e		$10(m/s)$	
Equilibrium spacing x_e		$13(m)$	
Standstill spacing s_0		$5(m)$	
Vehicle number in the platoon N		20	
Total study period T		$40(s)$	
Single oscillation case		Compound oscillation case	
Total time delay θ_1	$0.5(s)$	θ_1, θ_2	$0.5(s), 0.3(s)$
Frequency ω_1	$0.1\pi(Hz)$	ω_1, ω_2	$0.1\pi(Hz),$
Phase shift ϕ_1	$\pi/2(^{\circ})$	ϕ_1, ϕ_2	$\pi/2(^{\circ}), 0(^{\circ})$
Amplitude $A_0^{(1)}$	$10m$	$A_0^{(1)}, A_0^{(2)}$	$10m, 3m$

345

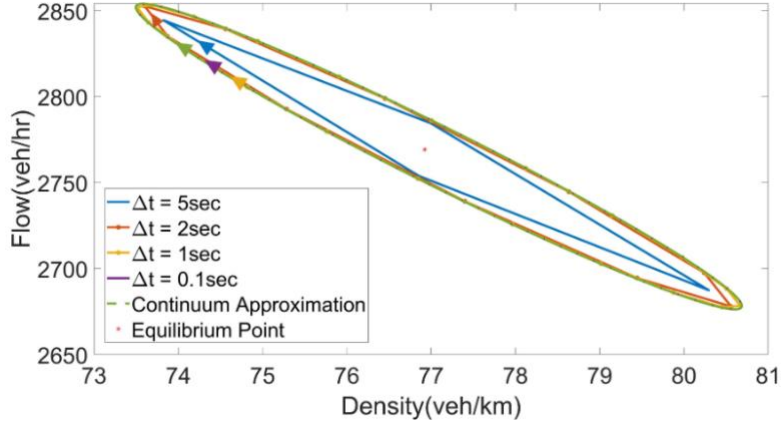
346 **4.2 Effects of measurement region**

347 Here we investigate the effects of various parameters on the features of dynamic FD. We first
348 examine the effects of the measurement region, in terms of the width of time window and the
349 platoon size, on the traffic hysteresis under single-frequency and compound (i.e., multi-frequency)
350 oscillations.

351 **Width of time window**

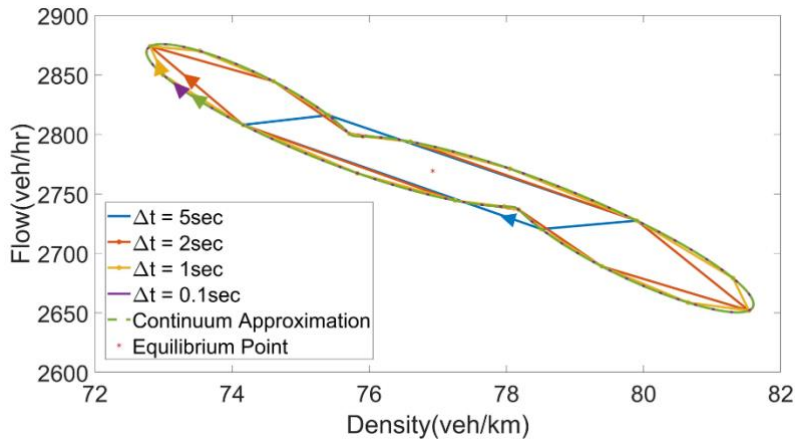
352 We first analyze a case with a single-frequency oscillation. As an example, we set $\omega_1 =$
353 $0.1\pi Hz$, $\phi_1 = \frac{\pi}{2}$, and oscillation magnitude $A_0^{(1)} = 10m$. Then, we have $|G(j\omega_p)| = 0.9917$ and
354 $\angle G(j\omega_p) = -0.2429 sec$. To examine the effectiveness of CA, we vary the width of time
355 window, $\Delta t = \{5, 2, 1, 0.1, 0\} sec$, where $\Delta t = 0 sec$ representing CA. The results are shown
356 in Fig. 3. Notably, all FDs evolve clockwise over time, displaying evident hysteresis around the
357 equilibrium point. Further, we observe a noticeable transformation in the shape of the
358 hysteresis as Δt decreases, transitioning from a polygon to an ellipse. This indicates that
359 reducing the window width allows for a more precise measurement of the hysteresis and
360 that the proposed CA method for measurement is highly desired. A large Δt (e.g., $\Delta t = 5s$)
361 evidently underestimates the hysteresis.

362



363
364 Figure 3: Dynamic FD of different Δt and continuum approximation: Single-frequency oscillation

365 We further extend our investigation to compound oscillations. This is accomplished by adding
366 another oscillation component with $\omega_2 = 0.3\pi \text{ Hz}$, $A_0^{(2)} = 3m$, and $\phi_2 = 0$. Moreover, we set
367 $\theta = 0.3 \text{ sec}$. The window width is also varied at $\Delta t = \{5, 2, 1, 0.1, 0\} \text{ sec}$. The results are given
368 in Fig. 4. Compared with the single-frequency oscillation case, the shapes of hysteresis loops with
369 compound oscillations are less regular, albeit still cyclic. Furthermore, under the compound
370 oscillations, the underestimation of hysteresis is more pronounced with greater Δt (e.g., 60.45%
371 underestimation of the loop area for $\Delta t = 5 \text{ sec}$, as opposed to 48.47% in the single-frequency
372 case). The CA-based measurement remains effective in capturing the comprehensive hysteresis
373 phenomenon. As per the finding, subsequent experiments are conducted based on the CA method.

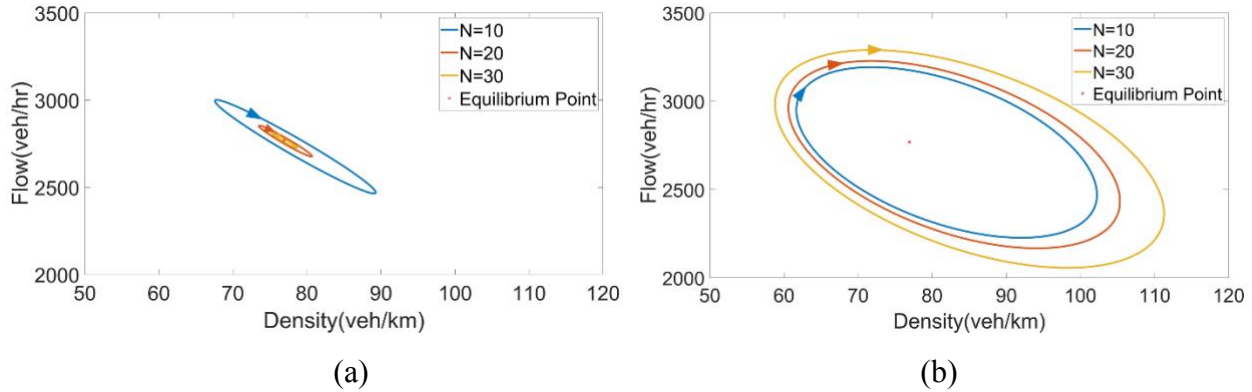


374
375 Figure 4: Dynamic FD of different Δt and continuum approximation: Compound oscillations

376 Platoon size

377 Here we investigate the effect of the platoon size (N) on the dynamic FD (based on CA). Fig.
378 5(a) and 5(b) illustrate the effect for string stable and string unstable platoons, respectively. For
379 the former, we use the default setting. For the string unstable platoons, however, we modify both
380 k_s and k_v to 0.5, which gives $|G(j\omega_p)| = 1.0847 > 1$ and $\angle G(j\omega_p) = -0.2818 \text{ sec}$. In Fig. 5(a)
381 and 5(b), we see that each FD evolves clockwise as an ellipse over time. The regularity in shape
382 (perfect ellipse) is attributed to the single-frequency oscillation and the assumption of time-

383 invariant and deterministic CF behavior. It is worth noting that the loop areas for the string unstable
 384 platoons are generally much larger than the string stable ones. Further, from Fig. 5(a), the
 385 movement area, in terms of both ellipsoidal length and width, decreases as N increases because
 386 $|G(j\omega_p)| < 1$, causing the oscillation to be dampened over space. Conversely, in Fig. 5(b), the
 387 hysteresis loop area expands as N increases, suggesting the oscillation being amplified through the
 388 platoon.



389
 390 (a) (b)
 391 Figure 5: CA Dynamic FD with different N : (a) string stable ($|G(j\omega_p)| = 0.9917$) and (b) string
 392 unstable ($|G(j\omega_p)| = 1.0847$)

393 4.3 Effects of car-following (control) parameters

394 This subsection aims to unveil the impact of the car-following control parameter setting on the
 395 dynamic FD. Specifically, the control gains, total time delay in sensing and control actuation,
 396 desired time gap, and equilibrium speed are explored.

397 Control gains

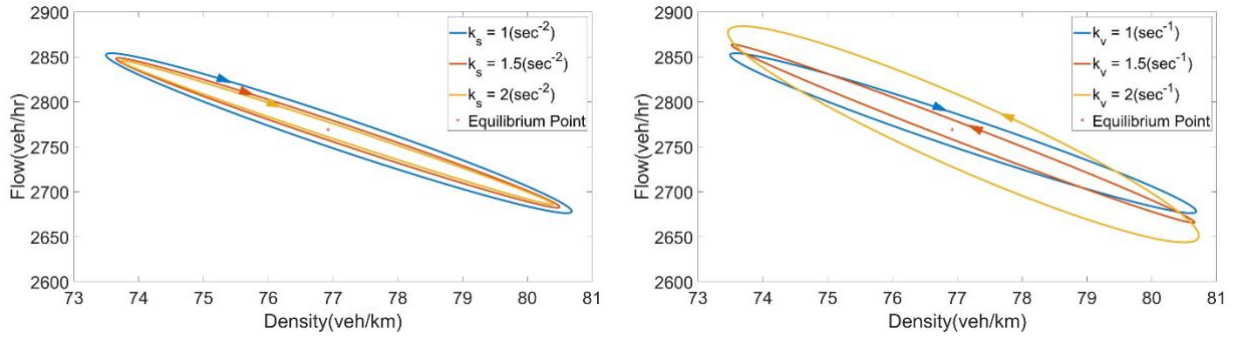
398 To analyze the impact of the control gain setting on the dynamic FD, we conduct a twofold
 399 investigation: (i) the effect of each control gain and (ii) the joint effect of two control gains. we
 400 vary the values of k_s and k_v based on the feasible regions identified in Kontar et al. (2021).
 401 Specifically, we vary them from 1 to 2 with an increment of 0.5. (The units for k_s and k_v are
 402 respectively sec^{-2} and sec^{-1} .) The result for k_s is shown in Fig. 6(a). Notably, an increase in
 403 spacing feedback gain leads to a reduction of the hysteresis loop. This can be attributed to the fact
 404 that a larger k_s results in a stronger response to a deviation from the equilibrium spacing, leading
 405 to less fluctuations and a smaller hysteresis loop.

406 The impact of k_v is shown in Fig. 6(b). Interestingly, unlike k_s , k_v mainly affects the slope
 407 (congested wave speed), with a higher value of k_v indicating a higher wave speed. Since k_v is
 408 responsible for regulating the speed difference, a higher sensitivity to the speed difference (i.e., a
 409 higher k_v) leads to a quicker response, leading to a higher observed wave speed. This is consistent
 410 with the finding in Kontar et al. (2021).

411 To quantify the features of hysteresis loops in Fig. 6, other than the ranges of density and flow
 412 (i.e., $k_{range} = k_{max} - k_{min}$, $Q_{range} = Q_{max} - Q_{min}$), we further define the width, length, and the
 413 area of each loop as illustrated in Fig. 7. Besides, the average wave speed in the congested branch
 414 (i.e., $w = \frac{Q_{min} - Q_{max}}{k_{max} - k_{min}}$) is also reported. The results presented in Table 2 show that changing k_s
 415 affects both width and length of the ellipsoidal loop, while changing k_v mainly affects the length

416 of hysteresis. Note that the area initially decreases and then increases as k_v increases. This
 417 phenomenon is primarily attributed to a change in the orientation of hysteresis loop from clockwise
 418 to counter-clockwise. Theoretically, the area would become zero (reducing to a traditional linear
 419 relationship) where this change in orientation occurs. We will put more emphasis on the orientation
 420 later in the joint influence.

421 The above findings underscore the difference between the traditional zero-order (static) FD
 422 and our dynamic FD, which possesses the capability to capture higher-order traffic flow features.



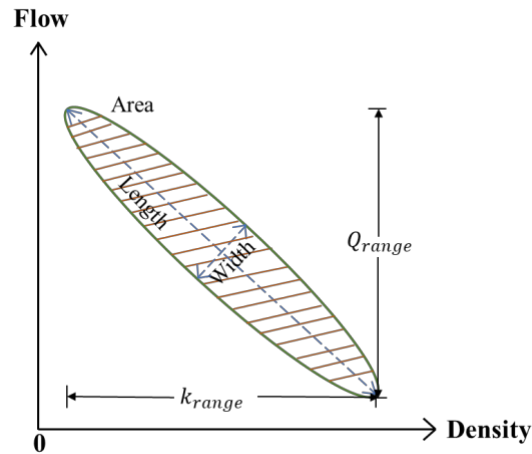
423
424

425

(a)

(b)

426 Figure 6: Dynamic FD with different control parameters (a) k_s and (b) k_v ($\tau = 0.8s, v_e =$
 427 $10m/s, \theta = 5sec, k_s = 1s^{-1}$)



428

429 Figure 7: Illustration of hysteresis loop length, width, area, and range

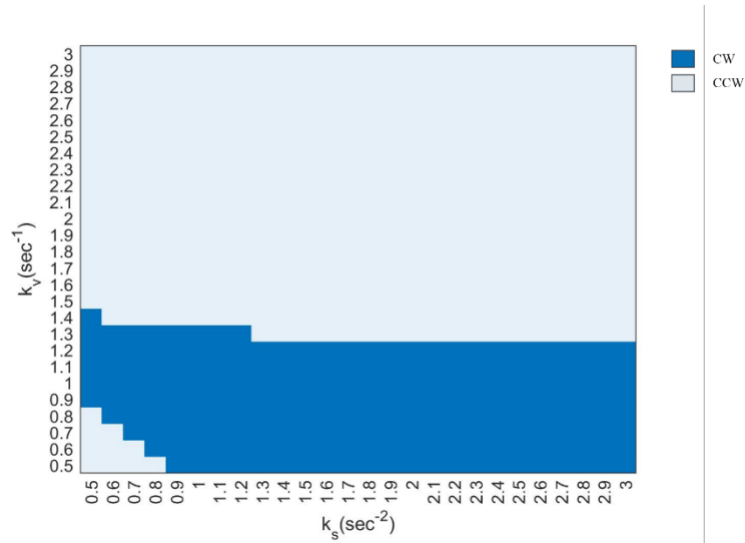
430

431 Table 2: Traffic hysteresis measures with respect to (k_s, k_v)

	$k_s(\text{sec}^{-2})$			$k_v(\text{sec}^{-1})$		
	1	1.5	2	1	1.5	2
	$k_e = 76.92(\text{veh}/\text{km}), Q_e = 2769.23(\text{veh}/\text{hr})$					
$k_{range}(\text{veh}/\text{km})$	7.12	6.84	6.67	7.12	6.96	7.10
$Q_{range}(\text{veh}/\text{hr})$	170.79	163.44	158.75	170.79	190.05	238.00
<i>Maximum flow</i> $(Q_{max}, \text{veh}/\text{hr})$	2854.03	2848.79	2845.69	2854.03	2863.84	2884.13
<i>Width</i> (veh/km)	14.28	13.64	13.30	14.28	14.25	14.29
<i>Length</i> (veh/hr)	177.27	166.02	159.36	177.27	197.96	240.11
<i>Area</i> $(\text{veh}^2/(\text{km} \cdot \text{hr}))$	213.91	125.77	82.61	213.91	117.33	473.86
$\bar{w}(\text{km}/\text{hr})$	-23.97	-23.91	-23.83	-23.97	-27.32	-33.51

432 We are further interested in the joint influence of control gains over the dynamic FD as they
433 are designed to work together. To explore this systematically, we expand the range of both k_s and
434 k_v to $[0.5, 3.0]$ with a step size of 0.1. Fig. 8 provides a heatmap of the hysteresis orientation with
435 respect to k_s and k_v . From the figure, two distinct boundaries are notable where the orientation
436 switches from clockwise (noted as CW in the figure) to counter-clockwise (CCW) and vice versa.
437 A significant majority (approximately 70%) of these loops exhibit a counter-clockwise pattern.
438 This finding diverges from the conclusion drawn in the empirical study by Ahn et al. (2013) for
439 human-driven vehicles, where clockwise loops are predominantly observed.

440 Fig. 9 presents the results of a sensitivity analysis, illustrating the performances of the CAV
441 system for different (k_s, k_v) pairs within a physically reasonable range of $[1.0, 3.0]$. Notably, we
442 have adjusted the lower bound for k_s and k_v to 1.0, as small values of k_s and k_v indicate an
443 unresponsive controller, which is undesired. From Fig. 9(a), we observe that $|G|$ demonstrates a
444 monotonically decreasing trend as k_v increases. Conversely, for k_s , $|G|$ initially experiences a
445 decline until reaching a minimum value at $k_s = 1.1$, subsequently ascending as k_s further
446 increases. In Fig. 9(b), the z-axis is reversed to provide a visual representation of the response time
447 (i.e., the absolute value of ΔG). Notably, the system exhibits the shortest response time when k_s is
448 relatively small and k_v is large. This observation suggests that the system's damping
449 characteristics are more pronounced when employing lower values of k_s in conjunction with
450 higher values of k_v . Besides, the range of density in hysteresis is more affected by the setting of
451 k_s than k_v (Fig. 9(c)). The behavior of density is non-monotonic, with the combination of $k_s = 3$
452 and $k_v = 1.4$ yielding the minimum density range. The same trend holds for the flow range, where
453 the optimal combination is $k_s = 3$ and $k_v = 1.1$ (Fig. 9(d)).



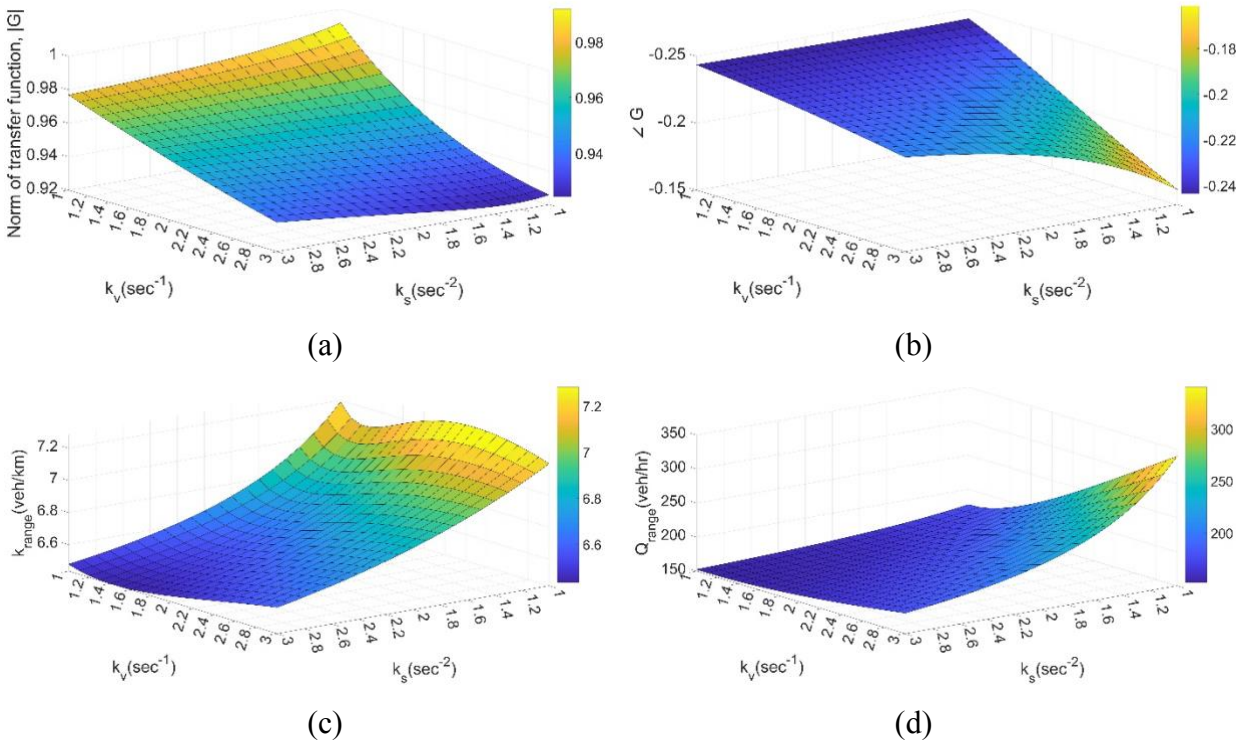
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Figure 8: Heatmap for hysteresis orientation with varying (k_s, k_v) ($\tau = 0.8s, v_e = 10m/s, \theta = 5sec$)



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Figure 9: Performances with varying (k_s, k_v) in terms of (a) norm of transfer function, (b) angle of transfer function, (c) range of density, and (d) range of flow

464 **Other parameters**

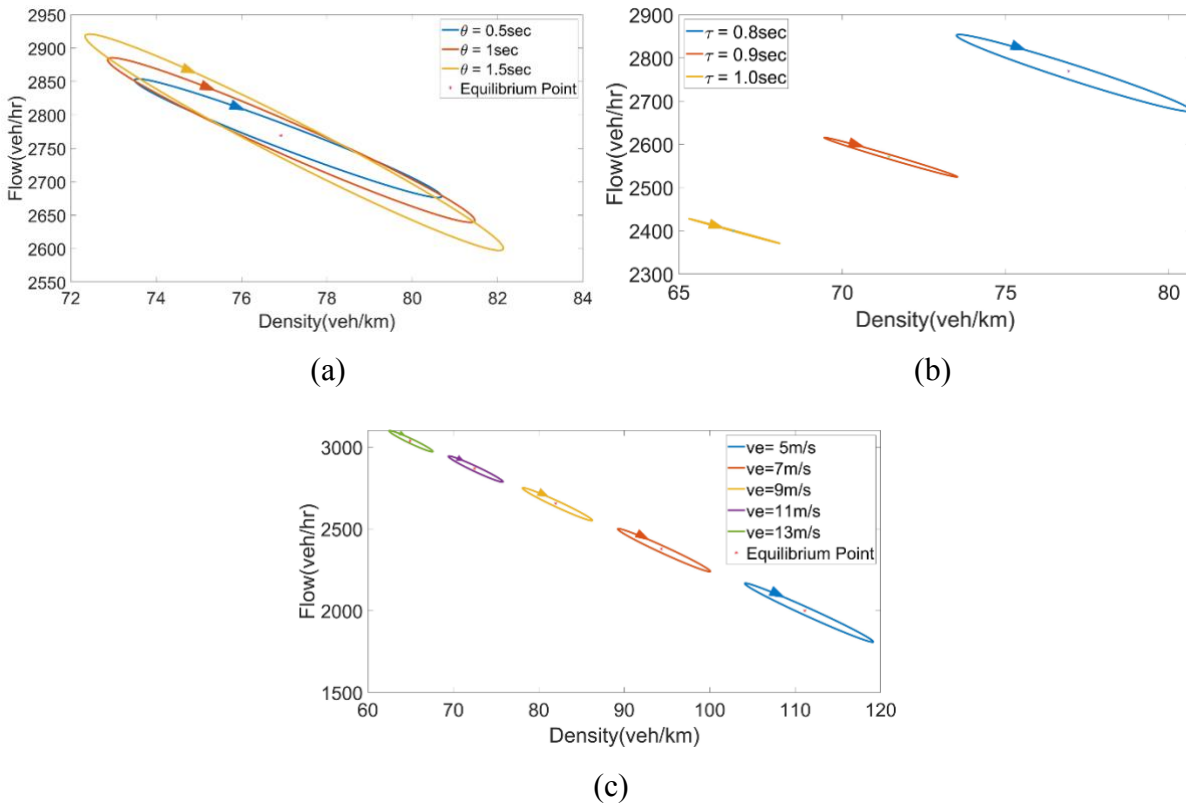
465 Next, the individual impacts of the total time delay (θ), desired time gap (τ), and equilibrium
 466 speed (v_e) on the dynamic FD are investigated. Fig. 10(a) illustrates the effect of the time delay
 467 (0.5sec, 1sec, 1.5sec), revealing that an increase in delay leads to a deterioration in the platoon's
 468 performance, accompanied by a larger hysteresis loop and higher wave speed. This is intuitive

469 since a larger delay in sensing and control actuation is known to make the system unstable (M.
 470 Wang, 2018).

471 For the desired time gap (set to 0.8sec , 0.9sec , and 1sec), the result in Fig. 10(b) shows that
 472 a tighter time gap (e.g., $\tau = 0.8\text{sec}$) yields higher flow and density, albeit at the expense of a larger
 473 hysteresis loop. This observation is also intuitive since maintaining a tighter spacing gives a
 474 smaller room for error, leading to system instability. This also suggests that a dynamic FD is
 475 particularly useful under more aggressive settings, in which traffic hysteresis can arise.

476 **Table 3** exhibits the quantitative measures of performance obtained through varying θ and τ .
 477 Notice that increasing θ also increases the wave speed. And when τ is set to 1sec , the hysteresis
 478 loop area is very small and approaches a linear pattern. This indicates that employing a more
 479 conservative (larger) desired time gap setting will lead to a more static traditional linear FD
 480 relationship.

481 Fig. 10(c) shows the dynamic FD with the equilibrium speed varying from 5m/s to 13m/s
 482 with an increase of 2m/s . Notably, the slope from the origin to the equilibrium point of each
 483 hysteresis loop represents the value of v_e . Since v_e does not affect the hysteresis orientation (see
 484 Eq. (21)), all FDs in Fig. 10(c) are clockwise. **Table 4** further provides quantitative measures of
 485 performances of different v_e . It is noteworthy that the setting of v_e can have a significant impact
 486 on dynamic FD, as reflected by the area change. Further, the average wave speed tends to increase
 487 with v_e , albeit at a smaller magnitude.
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493 Figure 10: Dynamic FD with different (a) time delay (θ)($\tau = 0.8\text{s}$, $v_e = 10\text{m/s}$), (b) desired
 494 time gap (τ)($v_e = 10\text{m/s}$, $\theta = 5\text{sec}$), and (c) equilibrium speed (v_e) ($\tau = 0.8\text{s}$, $\theta = 5\text{sec}$)

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Table 3: Traffic hysteresis measures with respect to θ and τ

	$\theta(\text{sec})$			$\tau(\text{sec})$		
	0.5	1	1.5	0.8	0.9	1
$k_e(\text{veh/km})$	76.92	76.92	76.92	79.62	71.43	66.67
$Q_e(\text{veh/hr})$	2769.23	2769.23	2769.23	2769.23	2571.43	2400.00
$k_{\text{range}}(\text{veh/km})$	7.12	8.60	9.78	7.12	4.07	2.64
$Q_{\text{range}}(\text{veh/hr})$	170.79	236.88	320.49	170.79	90.31	54.78
Maximum flow ($Q_{\text{max}}, \text{veh/hr}$)	2854.03	2885.64	2920.93	2854.03	2615.85	2427.96
Width(veh/km)	14.28	17.10	19.48	14.28	8.13	5.54
Length(veh/hr)	177.27	245.76	323.06	177.27	91.16	56.90
Area($\text{veh}^2/(\text{km} \cdot \text{hr})$)	213.91	337.49	457.70	213.91	34.46	3.66
$\bar{w}(\text{km/hr})$	-23.97	-27.57	-32.79	-23.97	-22.17	-20.77

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Table 4: Traffic hysteresis measures with respect to v_e

	$v_e(\text{m/s})$				
	5	7	9	11	13
$k_e(\text{veh/km})$	111.11	94.34	81.97	72.46	64.94
$Q_e(\text{veh/hr})$	2000.00	2377.36	2655.74	2869.57	3038.96
$k_{\text{range}}(\text{veh/km})$	15.04	10.83	8.17	6.38	5.12
$Q_{\text{range}}(\text{veh/hr})$	359.66	262.28	200.47	158.75	129.22
Maximum flow ($Q_{\text{max}}, \text{veh/hr}$)	2167.85	2501.11	2751.10	2945.54	3101.10
Width(veh/km)	29.96	21.55	16.24	12.67	10.15
Length(veh/hr)	358.73	261.62	199.98	158.37	128.92
Area($\text{veh}^2/(\text{km} \cdot \text{hr})$)	581.91	500.56	391.37	295.13	243.26
$\bar{w}(\text{km/hr})$	-23.91	-24.22	-24.54	-24.88	-25.24

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501 5 Conclusions

502 FDs have been extensively analyzed for decades, mainly for human-driven vehicles. Recently, the
503 advent of AVs has challenged the applicability of classic static FDs as they do not fully capture
504 the higher-order characteristics, such as traffic hysteresis, of AV car-following control. To fill this
505 gap, this paper analytically formulated dynamic FD, based on an analytical car-following control
506 law. The frequency domain representation of the car-following law as a transfer function simplified
507 mathematical derivations. We further applied CA for the derived dynamic FD to provide more
508 accurate, higher-resolution evolution of flow-density relationship over time and incorporate
509 higher-order features. The derived dynamic FD enabled a systematic investigation into the
510 presence of traffic hysteresis in AV platoons and potential factors that impact AV FDs. To
511 systematically determine the orientation of traffic hysteresis, a right-hand rule based criterion was
512 presented.

513 We conducted a series of numerical experiments to verify the analytical results and examine
514 the properties of the derived dynamic FD with respect to various parameters. First, we investigated
515 the effects of the flow-density measurement region, expressed by the width of time window and

516 the platoon size. The results suggest that a large time window can lead to underestimation of
 517 hysteresis, underscoring the importance of using CA. The underestimation was more pronounced
 518 with compound oscillations, consisting of multiple frequency components. We also provided some
 519 quantitative insights into the impact of car-following control parameter setting on the properties of
 520 dynamic FD. Specifically, the control gains, particularly the gain that regulates speed difference,
 521 significantly impact the hysteresis magnitude, orientation, and the average wave speed. Further,
 522 the delay in sensing and control actuation, desired time gap, and equilibrium speed were all found
 523 to have unique impacts on the hysteresis properties. Understanding these effects would be crucial
 524 for optimizing the platoon behavior in various traffic scenarios and effectively managing traffic
 525 flow.

526 Some future studies are nonetheless desirable. First, in this study, we focus on the congested
 527 regime of pure AV traffic for analytical tractability. Traffic hysteresis associated with transition
 528 between uncongested and congested states or between different equilibrium states is deferred to a
 529 future study. Second, we adopted the vertical measurement window for simplicity due to the lack
 530 of knowledge on the wave propagation in AV traffic. A better understanding of this feature may
 531 lead to a better selection of the measurement window. Third, for the purpose of analytical
 532 derivation, we made several simplifying assumptions such as homogeneous AV traffic and time-
 533 invariant and deterministic car-following features (i.e., constant parameters). However, the
 534 intrinsic stochastic nature of traffic flow can significantly impact traffic properties such as stability
 535 and capacity. Therefore, the deterministic assumption should be relaxed in the future. This can be
 536 achieved, for example, by embracing stochastic and time-varying CF models (e.g., Jiang et al.,
 537 2023). Further, linearization was undertaken to establish an analytical linkage between a CF law
 538 and dynamic FD. For enhanced precision, future research may address the nonlinear characteristics
 539 in CF laws by replacing the transfer function with data-derive functions, as demonstrated by prior
 540 works (e.g., Li & Ouyang, 2011; Zhou et al., 2023). Lastly, this study can also be extended to
 541 mixed traffic consisting of both AVs and human-driven vehicles to unveil how dynamic traffic
 542 features, as represented by dynamic FD, would evolve as the AV market penetration increases in
 543 the future.
 544

545 Appendix A. Derivation of spacing

546 From Eqs. (7) and (8), we obtain:

$$x_0(t) = v_e t + \lim_{M \rightarrow \infty} \sum_{m=1}^M A_0^{(m)} \sin(\omega_m t + \phi_m), \quad (\text{A.1})$$

$$x_N(t) = v_e t - N\Delta x_e + \lim_{M \rightarrow \infty} \sum_{m=1}^M A_0^{(m)} |G(j\omega_m)|^N \sin(\omega_m t + \phi_m + N\angle G(j\omega_m)). \quad (\text{A.2})$$

547 Therefore,

$$\begin{aligned} x_0(t) - x_N(t) &= N\Delta x_e + \sum_{m=1}^M A_0^{(m)} [\sin(\omega_m t + \phi_m) - |G(j\omega_m)|^N \sin(\omega_m t + \\ &\quad \phi_m + N\angle G(j\omega_m))] \\ &= N\Delta x_e + \sum_{m=1}^M A_0^{(m)} \{[(1 - |G(j\omega_m)|^N \cos(N\angle G(j\omega_m))) \sin(\omega_m t + \\ &\quad \phi_m)] - |G(j\omega_m)|^N \sin(N\angle G(j\omega_m)) \cos(\omega_m t + \phi_m)\} \end{aligned} \quad (\text{A.3})$$

548 For further simplification, assume $f(t) = \sum_{m=1}^M A_0^{(m)} \{[(1 -$
549 $|G(j\omega_m)|^N \cos(N\angle G(j\omega_m))] \sin(\omega_m t + \phi_m) - |G(j\omega_m)|^N \sin(N\angle G(j\omega_m)) \cos(\omega_m t +$
550 $\phi_m)\} = C_1 \sin(\omega_m t + \phi_m) - C_2 \cos(\omega_m t + \phi_m)$.

551 According to trigonometry,

$$f(t) = C_1 \sin(\omega_m t + \phi_m) - C_2 \cos(\omega_m t + \phi_m) = R \sin(\omega_m t + \phi_m - \phi_c), \quad (\text{A.4})$$

552 where the amplitude

$$R = \sqrt{(C_1)^2 + (C_2)^2} = \sqrt{(1 - |G(j\omega_m)|^N \cos(N\angle G(j\omega_m)))^2 + (|G(j\omega_m)|^N \sin(N\angle G(j\omega_m)))^2} = \sqrt{1 - 2|G(j\omega_m)|^N \cos(N\angle G(j\omega_m)) + |G(j\omega_m)|^{2N}}, \quad (\text{A.5})$$

$$\phi_c = \arctan \frac{C_2}{C_1} = \arctan \frac{|G(j\omega_m)|^N \sin(N\angle G(j\omega_m))}{1 - |G(j\omega_m)|^N \cos(N\angle G(j\omega_m))}. \quad (\text{A.6})$$

553 Therefore, we can have Equation (9).

554 **Appendix B. Human driven vehicle vs. commercial AV**

555 To apply our framework on human driven vehicles, we select the generalized linear optimal
556 velocity model (GL-OVM). This model, extensively employed by researchers for its capability to
557 emulate traffic flow, adopts a car-following structure represented as follows:

$$a_l(t) = \dot{v}_l(t) = \kappa[V(\Delta x_l(t)) - v_l(t)] \quad (\text{A.7})$$

558 where $V(\Delta x_l(t))$ is the optimal speed function related to the spacing $\Delta x_l(t)$, κ is the sensitivity
559 parameter. Then, the exponential optimal speed function established by Newell (1961) is denoted
560 as:

$$V(\Delta x_l(t)) = v_0 \left[1 - \exp\left(-\frac{\alpha}{v_0}(\Delta x_l(t) - s_0)\right) \right] \quad (\text{A.8})$$

561 where v_0 is the free-flow speed, s_0 is the jam distance, and α is the parameter related to the
562 congested wave speed. Define $\hat{v}_l(t) := v_l(t) - v_e$ and $\Delta \hat{x}_l(t) := \Delta x_l(t) - \Delta x_e$, where $\hat{v}_l(t)$ is
563 speed deviation from equilibrium speed v_e , $\Delta \hat{x}_l(t)$ is the spacing deviation from equilibrium
564 spacing Δx_e .

565 Then, we linearize the CF law for human driven vehicles in Eq. (A.7) via Taylor expansion
566 around the equilibrium state, in which $a_l(t) = 0$,

$$V(\Delta x_l(t)) = V(\Delta x_e) + \frac{V'(\Delta x_e)}{1!} (\Delta x_l(t) - \Delta x_e) \quad (\text{A.9})$$

567 As $V(\Delta x_e)$ is the equilibrium speed, i.e., $V(\Delta x_e) = v_e = v_0 \left[1 - \exp\left(-\frac{\alpha}{v_0}(\Delta x_l(t) - s_0)\right) \right]$
568 according to Eq. (A.7). Eq. (A.9) can be further written as:

$$V(\Delta x_l(t)) = V'(\Delta x_e)(\Delta x_l(t) - \Delta x_e) + v_e \quad (\text{A.10})$$

569 Substitute Eq. (A.10) into Eq. (A.7),

$$a_l(t) = \dot{v}_l(t) = \kappa[V'(\Delta x_e)(\Delta x_l(t) - \Delta x_e) + v_e - v_l(t)] \quad (\text{A.11})$$

570 where $V'(\Delta x_e)$ is the first order derivative of optimal speed function in Eq. (A.7) with respect to
571 the equilibrium spacing. It can be computed as:

$$V'(\Delta x_e) = \alpha \exp\left[-\frac{\alpha}{v_0}(\Delta x_e(v_e) - s_0)\right] \quad (\text{A.12})$$

572 where $\Delta x_e(v_e)$ is the equilibrium spacing as a function of equilibrium speed v_e . Combining with
573 Eq. (A.8) and solving for Δx_e when $v_l(t) = v_e$, we obtain

$$\Delta x_e(v_e) = s_0 - \frac{v_0}{\alpha} \ln\left(1 - \frac{v_e}{v_0}\right) \quad (\text{A.13})$$

574 Plugging $\Delta \hat{x}_l$ and $\hat{v}_l(t)$ into Eq. (A.11), we have

$$\dot{\hat{v}}_l(t) = \kappa V'(\Delta x_e) \Delta \hat{x}_l(t) - \kappa \hat{v}_l(t) \quad (\text{A.14})$$

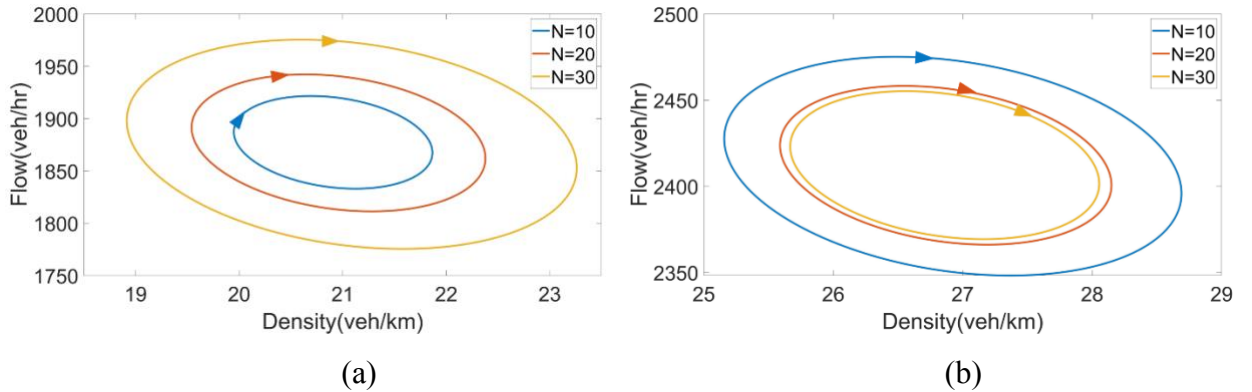
575 Then we conduct Laplace transform to convert \hat{v}_l and $\Delta \hat{x}_l$ into the frequency domain:

$$\Delta \hat{x}_l(s) = \frac{\hat{v}_{l-1}(s) - \hat{v}_l(s)}{s} \quad (\text{A.15})$$

576 Finally, the transfer function of linearized OVM based on first order Taylor derivative $G_l(s)$
577 can be expressed as:

$$G_l(s) = \frac{\kappa V'(\Delta x_e)}{s^2 + \kappa s + \kappa V'(\Delta x_e)} \quad (\text{A.16})$$

578 We further conduct a numerical experiment to compare the features of traffic hysteresis
579 between human driven vehicles and AVs. For the parameter setting, we adopt the calibration
580 results of OVM by Wang et al. (2012), where $v_0 = 33.3m/s$, $\kappa = 0.700s^{-1}$, $\alpha = 0.999s^{-1}$, and
581 $s_0 = 1.62m$. For the linear controller, we use the calibration results for a commercial AV by Jiang
582 et al. (2023), where $k_s = 0.3790 (s^{-2})$, $k_v = 0.3937(s^{-1})$, $\tau = 0.8 s$, $s_0 = 7.3625m$, and $\theta =$
583 $0sec$. Note that here $v_e = 25m/s$ and $A_0 = 10m$. Figure A1 shows the dynamic FD of GL-OVM
584 and commercial AV respectively.



585
586
587 Figure A1: CA Dynamic FD with different N : (a) GL-OVM and (b) commercial AV

588 From the figure, we can notice that though all hysteresis loops are clockwise for both models.
589 However, the AV exhibits higher throughput and is string stable under the given oscillation setting,
590 whereas a platoon based on GL-OVM is string unstable.

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593

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